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- (614) 447-3698 Elsewhere

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We can also be contacted via email at HELP@CAS.ORG.

Connection closed by remote host

Connecting via Winsock to STN

Welcome to STN International! Enter x:X

LOGINID:ssptamlsl1742

PASSWORD:

TERMINAL (ENTER 1, 2, 3, OR ?):2

* * * * * Welcome to STN International * * * * *

NEWS 1 Web Page for STN Seminar Schedule - N. America

NEWS 2 APR 04 STN AnaVist, Version 1, to be discontinued
 NEWS 3 APR 15 WPIDS, WPINDEX, and WPIX enhanced with new
 predefined hit display formats
 NEWS 4 APR 28 EMBASE Controlled Term thesaurus enhanced
 NEWS 5 APR 28 IMSRESEARCH reloaded with enhancements
 NEWS 6 MAY 30 INPAFAMDB now available on STN for patent family
 searching
 NEWS 7 MAY 30 DGENE, PCTGEN, and USGENE enhanced with new homology
 sequence search option
 NEWS 8 JUN 06 EPFULL enhanced with 260,000 English abstracts
 NEWS 9 JUN 06 KOREAPAT updated with 41,000 documents
 NEWS 10 JUN 13 USPATFULL and USPAT2 updated with 11-character
 patent numbers for U.S. applications
 NEWS 11 JUN 19 CAS REGISTRY includes selected substances from
 web-based collections
 NEWS 12 JUN 25 CA/CAPplus and USPAT databases updated with IPC
 reclassification data
 NEWS 13 JUN 30 AEROSPACE enhanced with more than 1 million U.S.
 patent records
 NEWS 14 JUN 30 EMBASE, EMBAL, and LEMBASE updated with additional
 options to display authors and affiliated
 organizations
 NEWS 15 JUN 30 STN on the Web enhanced with new STN AnaVist
 Assistant and BLAST plug-in
 NEWS 16 JUN 30 STN AnaVist enhanced with database content from EPFULL
 NEWS 17 JUL 28 CA/CAPplus patent coverage enhanced
 NEWS 18 JUL 28 EPFULL enhanced with additional legal status
 information from the epoline Register
 NEWS 19 JUL 28 IFICDB, IFIPAT, and IFIUDB reloaded with enhancements
 NEWS 20 JUL 28 STN Viewer performance improved
 NEWS 21 AUG 01 INPADOCDB and INPAFAMDB coverage enhanced
 NEWS 22 AUG 13 CA/CAPplus enhanced with printed Chemical Abstracts
 page images from 1967-1998
 NEWS 23 AUG 15 CAOLD to be discontinued on December 31, 2008
 NEWS 24 AUG 15 CAPplus currency for Korean patents enhanced
 NEWS 25 AUG 25 CA/CAPplus, CASREACT, and IFI and USPAT databases
 enhanced for more flexible patent number searching
 NEWS 26 AUG 27 CAS definition of basic patents expanded to ensure
 comprehensive access to substance and sequence
 information

NEWS EXPRESS JUNE 27 08 CURRENT WINDOWS VERSION IS V8.3,
 AND CURRENT DISCOVER FILE IS DATED 23 JUNE 2008.

NEWS HOURS STN Operating Hours Plus Help Desk Availability
 NEWS LOGIN Welcome Banner and News Items
 NEWS IPC8 For general information regarding STN implementation of IPC 8

Enter NEWS followed by the item number or name to see news on that
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* * * * * STN Columbus * * * * *

FILE 'HOME' ENTERED AT 09:40:56 ON 08 SEP 2008

=>

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=> file registry
COST IN U.S. DOLLARS                SINCE FILE      TOTAL
                                     ENTRY      SESSION
FULL ESTIMATED COST                0.21        0.21
```

FILE 'REGISTRY' ENTERED AT 09:41:08 ON 08 SEP 2008
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```
STRUCTURE FILE UPDATES:    7 SEP 2008  HIGHEST RN 1047406-12-1
DICTIONARY FILE UPDATES:   7 SEP 2008  HIGHEST RN 1047406-12-1
```

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TSCA INFORMATION NOW CURRENT THROUGH July 5, 2008.

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Connecting via Winsock to STN

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LOGINID:ssptamls1742

PASSWORD:

```
* * * * * RECONNECTED TO STN INTERNATIONAL * * * * *
SESSION RESUMED IN FILE 'REGISTRY' AT 09:41:50 ON 08 SEP 2008
FILE 'REGISTRY' ENTERED AT 09:41:50 ON 08 SEP 2008
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```

```
COST IN U.S. DOLLARS                SINCE FILE      TOTAL
                                     ENTRY      SESSION
FULL ESTIMATED COST                0.46        0.67
```

```
=> file registry
COST IN U.S. DOLLARS                SINCE FILE      TOTAL
                                     ENTRY      SESSION
FULL ESTIMATED COST                0.46        0.67
```

FILE 'REGISTRY' ENTERED AT 09:42:04 ON 08 SEP 2008
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STRUCTURE FILE UPDATES: 7 SEP 2008 HIGHEST RN 1047406-12-1
DICTIONARY FILE UPDATES: 7 SEP 2008 HIGHEST RN 1047406-12-1

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on property searching in REGISTRY, refer to:

<http://www.cas.org/support/stngen/stndoc/properties.html>

=> s 3.5-4.5 Mg/mac and 0.8-1.5 Mn/mac and 0-0.5 Si/mac and 0-0.5 Fe and 80-100
Al/mac

181008 3.5-4.5/MAC
79605 MG/MAC
5521 3.5-4.5 MG/MAC
(3.5-4.5/MAC (P) MG/MAC)
419377 0.8-1.5/MAC
385693 MN/MAC
134596 0.8-1.5 MN/MAC
(0.8-1.5/MAC (P) MN/MAC)
529244 0-0.5/MAC
412336 SI/MAC
216154 0-0.5 SI/MAC
(0-0.5/MAC (P) SI/MAC)
911670 0
64739 0.5
655143 FE
407 FES
655531 FE
(FE OR FES)
32 0-0.5 FE
(0(W)0.5(W)FE)
514427 80-100/MAC
263320 AL/MAC
74008 80-100 AL/MAC
(80-100/MAC (P) AL/MAC)

L1 1 3.5-4.5 MG/MAC AND 0.8-1.5 MN/MAC AND 0-0.5 SI/MAC AND 0-0.5 FE
AND 80-100 AL/MAC

=> d l1

L1 ANSWER 1 OF 1 REGISTRY COPYRIGHT 2008 ACS on STN
RN 1001846-01-0 REGISTRY
ED Entered STN: 06 Feb 2008
CN Aluminum alloy, base, Al 82-99, Mn 1.1-7, Mg 0.1-6, Sc 0-1.5, Cr 0-0.5, Cu
0-0.5, Ni 0-0.5, Si 0-0.5, Co 0-0.1, Fe 0-0.1, La 0-0.1, Mo 0-0.1, Nb 0-0.1, Ti
0-0.1, V 0-0.1, W 0-0.1, Y 0-0.1, Zn 0-0.1, Zr 0-0.1 (CA INDEX NAME)
OTHER NAMES:
CN Mn 1.1-7, Mg 0.1-6, Sc 0-1.5, Si 0-0.5, Fe 0-0.1, Cu 0-0.5, Cr 0-0.5,
Ni 0-0.5, Ti 0-0.1, V 0-0.1, Co 0-0.1, Zn 0-0.1, Zr 0-0.1, Nb 0-0.1, Mo
0-0.1, Y 0-0.1, W 0-0.1, La 0-0.1, Al bal.
MF Al . Co . Cr . Cu . Fe . La . Mg . Mn . Mo . Nb . Ni . Sc . Si . Ti . V .
W . Y . Zn . Zr
CI AYS
SR CA

LC STN Files: CA, CAPLUS

Component	Component Percent	Component Registry Number
Al	82 - 99	7429-90-5
Mn	1.1 - 7	7439-96-5
Mg	0.1 - 6	7439-95-4
Sc	0 - 1.5	7440-20-2
Cr	0 - 0.5	7440-47-3
Cu	0 - 0.5	7440-50-8
Ni	0 - 0.5	7440-02-0
Si	0 - 0.5	7440-21-3
Co	0 - 0.1	7440-48-4
Fe	0 - 0.1	7439-89-6
La	0 - 0.1	7439-91-0
Mo	0 - 0.1	7439-98-7
Nb	0 - 0.1	7440-03-1
Ti	0 - 0.1	7440-32-6
V	0 - 0.1	7440-62-2
W	0 - 0.1	7440-33-7
Y	0 - 0.1	7440-65-5
Zn	0 - 0.1	7440-66-6
Zr	0 - 0.1	7440-67-7

PROPERTY DATA AVAILABLE IN THE 'PROP' FORMAT

2 REFERENCES IN FILE CA (1907 TO DATE)
2 REFERENCES IN FILE CAPLUS (1907 TO DATE)

=> FIL REGISTRY

COST IN U.S. DOLLARS	SINCE FILE ENTRY	TOTAL SESSION
FULL ESTIMATED COST	39.43	40.10

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DICTIONARY FILE UPDATES: 7 SEP 2008 HIGHEST RN 1047406-12-1

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=> SET TERMSET E#

SET COMMAND COMPLETED

=> DEL SEL Y

=> SEL L1 1 RN

E1 THROUGH E1 ASSIGNED

=> S E1/RN

L2 1 1001846-01-0/RN

=> SET TERMSET LOGIN

SET COMMAND COMPLETED

=> FIL CAPLUS

COST IN U.S. DOLLARS	SINCE FILE ENTRY	TOTAL SESSION
FULL ESTIMATED COST	0.55	40.65

FILE 'CAPLUS' ENTERED AT 09:43:41 ON 08 SEP 2008
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FILE COVERS 1907 - 8 Sep 2008 VOL 149 ISS 11
FILE LAST UPDATED: 7 Sep 2008 (20080907/ED)

Caplus now includes complete International Patent Classification (IPC) reclassification data for the second quarter of 2008.

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=> S L2

L3 2 L2

=> DIS L3 1 IBIB IABS

THE ESTIMATED COST FOR THIS REQUEST IS 2.91 U.S. DOLLARS
DO YOU WANT TO CONTINUE WITH THIS REQUEST? (Y)/N:Y

L3 ANSWER 1 OF 2 CAPLUS COPYRIGHT 2008 ACS on STN

ACCESSION NUMBER: 2008:91482 CAPLUS
 DOCUMENT NUMBER: 148:173393
 TITLE: Aluminum alloys containing nanocomposite phases
 INVENTOR(S): Hung, Wei-Peng; Chen, Chien-Tong
 PATENT ASSIGNEE(S): Advanced Material Specialty Inc., Taiwan; Nelson Precision Casting Co., Ltd.
 SOURCE: Jpn. Kokai Tokkyo Koho, 10pp.
 CODEN: JKXXAF
 DOCUMENT TYPE: Patent
 LANGUAGE: Japanese
 FAMILY ACC. NUM. COUNT: 1
 PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
-----	----	-----	-----	-----
JP 2008013826	A	20080124	JP 2006-187815	20060707
PRIORITY APPLN. INFO.:			JP 2006-187815	20060707

ABSTRACT:

The title alloy has a chemical composition contain Mn 1.1-7.0, Mg 0.1-6.0, and Sc 0.01-1.5 weight% and includes long cylindrical nanocomposite phases. Optionally, the alloys also contain Si 0.01-0.5, Fe 0.01-0.10, Cu 0.01-0.50, Cr 0.01-0.50, Ni 0.01-0.50, Ti 0.01-0.1, V 0.01-0.1, Co 0.01-0.1, Zn 0.01-0.1, Zr 0.01-0.1, Nb 0.01-0.1, Mo 0.01-0.1, Y 0.01-0.1, W 0.01-0.1, and/or La 0.01-0.1 weight%. The alloys are especially suitable for golf club heads and golf club shafts.

=> DIS L3 2 IBIB IABS
 THE ESTIMATED COST FOR THIS REQUEST IS 2.91 U.S. DOLLARS
 DO YOU WANT TO CONTINUE WITH THIS REQUEST? (Y)/N:Y

L3 ANSWER 2 OF 2 CAPLUS COPYRIGHT 2008 ACS on STN

ACCESSION NUMBER: 2008:51647 CAPLUS
 DOCUMENT NUMBER: 148:219699
 TITLE: Aluminum alloy having nanometer compound phase for golf clubs
 INVENTOR(S): Hong, Weipeng; Chen, Jiantong
 PATENT ASSIGNEE(S): Amspec Material Inc., Peop. Rep. China; Fu Sheng Group
 SOURCE: Faming Zhuanli Shenqing Gongkai Shuomingshu, 10pp.
 CODEN: CNXXEV
 DOCUMENT TYPE: Patent
 LANGUAGE: Chinese
 FAMILY ACC. NUM. COUNT: 1
 PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
-----	----	-----	-----	-----
CN 101100716	A	20080109	CN 2006-10090373	20060703
PRIORITY APPLN. INFO.:			CN 2006-10090373	20060703

ABSTRACT:

The alloy comprises Mn 1.1-7, Mg 0.1-6, Sc 0.01-1.5%, Al bal. The alloy may further contain Si 0.01-0.5, Fe 0.01-0.1, Cu 0.01-0.5, Cr 0.01-0.5, Ni 0.01-0.5, Ti 0.01-0.1, V 0.01-0.1, Co 0.01-0.1, Zn 0.01-0.1, Zr 0.01-0.1, Nb 0.01-0.1, Mo 0.01-0.1, Y 0.01-0.1, W 0.01-0.1, and/or La 0.01-0.1.

=> FIL REGISTRY
 COST IN U.S. DOLLARS
 FULL ESTIMATED COST

SINCE FILE	TOTAL
ENTRY	SESSION
7.26	47.91

DISCOUNT AMOUNTS (FOR QUALIFYING ACCOUNTS)	SINCE FILE ENTRY	TOTAL SESSION
CA SUBSCRIBER PRICE	-1.60	-1.60

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 DICTIONARY FILE UPDATES: 7 SEP 2008 HIGHEST RN 1047406-12-1

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 predicted properties as well as tags indicating availability of
 experimental property data in the original document. For information
 on property searching in REGISTRY, refer to:

<http://www.cas.org/support/stngen/stndoc/properties.html>

=> s 3.5-4.5 Mg/mac and 0.8-1.5 Mn/mac and 0-0.5 Si/mac and 0-0.5 Fe and 0.1-2.0
 Ti/mac and 0.1-2.0 Zr/mac

181008 3.5-4.5/MAC
 79605 MG/MAC
 5521 3.5-4.5 MG/MAC
 (3.5-4.5/MAC (P) MG/MAC)
 419377 0.8-1.5/MAC
 385693 MN/MAC
 134596 0.8-1.5 MN/MAC
 (0.8-1.5/MAC (P) MN/MAC)
 529244 0-0.5/MAC
 412336 SI/MAC
 216154 0-0.5 SI/MAC
 (0-0.5/MAC (P) SI/MAC)
 911670 0
 64739 0.5
 655143 FE
 407 FES
 655531 FE
 (FE OR FES)
 32 0-0.5 FE
 (0(W)0.5(W)FE)
 674473 0.1-2.0/MAC
 142635 TI/MAC
 78487 0.1-2.0 TI/MAC
 (0.1-2.0/MAC (P) TI/MAC)
 674473 0.1-2.0/MAC
 71325 ZR/MAC
 37671 0.1-2.0 ZR/MAC
 (0.1-2.0/MAC (P) ZR/MAC)

L4 1 3.5-4.5 MG/MAC AND 0.8-1.5 MN/MAC AND 0-0.5 SI/MAC AND 0-0.5 FE
 AND 0.1-2.0 TI/MAC AND 0.1-2.0 ZR/MAC

=> d 14

L4 ANSWER 1 OF 1 REGISTRY COPYRIGHT 2008 ACS on STN
RN 1001846-01-0 REGISTRY
ED Entered STN: 06 Feb 2008
CN Aluminum alloy, base, Al 82-99,Mn 1.1-7,Mg 0.1-6,Sc 0-1.5,Cr 0-0.5,Cu
0-0.5,Ni 0-0.5,Si 0-0.5,Co 0-0.1,Fe 0-0.1,La 0-0.1,Mo 0-0.1,Nb 0-0.1,Ti
0-0.1,V 0-0.1,W 0-0.1,Y 0-0.1,Zn 0-0.1,Zr 0-0.1 (CA INDEX NAME)
OTHER NAMES:
CN Mn 1.1-7, Mg 0.1-6, Sc 0-1.5, Si 0-0.5, Fe 0-0.1, Cu 0-0.5, Cr 0-0.5,
Ni 0-0.5, Ti 0-0.1, V 0-0.1, Co 0-0.1, Zn 0-0.1, Zr 0-0.1, Nb 0-0.1, Mo
0-0.1, Y 0-0.1, W 0-0.1, La 0-0.1, Al bal.
MF Al . Co . Cr . Cu . Fe . La . Mg . Mn . Mo . Nb . Ni . Sc . Si . Ti . V .
W . Y . Zn . Zr
CI AYS
SR CA
LC STN Files: CA, CAPLUS

Component	Component Percent	Component Registry Number
Al	82 - 99	7429-90-5
Mn	1.1 - 7	7439-96-5
Mg	0.1 - 6	7439-95-4
Sc	0 - 1.5	7440-20-2
Cr	0 - 0.5	7440-47-3
Cu	0 - 0.5	7440-50-8
Ni	0 - 0.5	7440-02-0
Si	0 - 0.5	7440-21-3
Co	0 - 0.1	7440-48-4
Fe	0 - 0.1	7439-89-6
La	0 - 0.1	7439-91-0
Mo	0 - 0.1	7439-98-7
Nb	0 - 0.1	7440-03-1
Ti	0 - 0.1	7440-32-6
V	0 - 0.1	7440-62-2
W	0 - 0.1	7440-33-7
Y	0 - 0.1	7440-65-5
Zn	0 - 0.1	7440-66-6
Zr	0 - 0.1	7440-67-7

PROPERTY DATA AVAILABLE IN THE 'PROP' FORMAT

2 REFERENCES IN FILE CA (1907 TO DATE)
2 REFERENCES IN FILE CAPLUS (1907 TO DATE)

=> FIL REGISTRY

COST IN U.S. DOLLARS	SINCE FILE	TOTAL
	ENTRY	SESSION
FULL ESTIMATED COST	43.66	91.57
DISCOUNT AMOUNTS (FOR QUALIFYING ACCOUNTS)	SINCE FILE	TOTAL
	ENTRY	SESSION
CA SUBSCRIBER PRICE	0.00	-1.60

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DICTIONARY FILE UPDATES: 7 SEP 2008 HIGHEST RN 1047406-12-1

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=> SET TERMSET E#

SET COMMAND COMPLETED

=> DEL SEL Y

=> SEL L4 1 RN

E1 THROUGH E1 ASSIGNED

=> S E1/RN

L5 1 1001846-01-0/RN

=> SET TERMSET LOGIN

SET COMMAND COMPLETED

=> FIL CAPLUS

COST IN U.S. DOLLARS	SINCE FILE	TOTAL
	ENTRY	SESSION
FULL ESTIMATED COST	0.55	92.12
DISCOUNT AMOUNTS (FOR QUALIFYING ACCOUNTS)	SINCE FILE	TOTAL
	ENTRY	SESSION
CA SUBSCRIBER PRICE	0.00	-1.60

FILE 'CAPLUS' ENTERED AT 09:46:29 ON 08 SEP 2008
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FILE COVERS 1907 - 8 Sep 2008 VOL 149 ISS 11
FILE LAST UPDATED: 7 Sep 2008 (20080907/ED)

Caplus now includes complete International Patent Classification (IPC) reclassification data for the second quarter of 2008.

Effective October 17, 2005, revised CAS Information Use Policies apply. They are available for your review at:

<http://www.cas.org/legal/infopolicy.html>

=> S L5

L6 2 L5

=> DIS L6 1 TI

L6 ANSWER 1 OF 2 CAPLUS COPYRIGHT 2008 ACS on STN
TI Aluminum alloys containing nanocomposite phases

=> DIS L6 2 TI

L6 ANSWER 2 OF 2 CAPLUS COPYRIGHT 2008 ACS on STN
TI Aluminum alloy having nanometer compound phase for golf clubs

=> s 3.5-4.5 Mg/mac and 0.8-1.5 Mn/mac and 0-3.0 Si/mac and 0-3.0 Fe and 0.1-2.0 Ti/mac and 0.1-2.0 Zr/mac

'MAC' IS NOT A VALID FIELD CODE

0 3.5-4.5 MG/MAC

0 0.8-1.5 MN/MAC

0 0-3.0 SI/MAC

5989705 0

7366573 3

5989705 0

876873 FE

11373 FES

883113 FE

(FE OR FES)

28 0-3.0 FE

(0(W)3(W)0(W)FE)

0 0.1-2.0 TI/MAC

0 0.1-2.0 ZR/MAC

L7 0 3.5-4.5 MG/MAC AND 0.8-1.5 MN/MAC AND 0-3.0 SI/MAC AND 0-3.0 FE
AND 0.1-2.0 TI/MAC AND 0.1-2.0 ZR/MAC

=> s 3.5-4.5 Mg/mac and 0.8-1.5 Mn/mac and 0-3.0 Si/mac and 0-3.0 Fe/mac and 0.1-2.0 Ti/mac and 0.1-2.0 Zr/mac

'MAC' IS NOT A VALID FIELD CODE

0 3.5-4.5 MG/MAC

0 0.8-1.5 MN/MAC

0 0-3.0 SI/MAC

0 0-3.0 FE/MAC

0 0.1-2.0 TI/MAC

0 0.1-2.0 ZR/MAC

L8 0 3.5-4.5 MG/MAC AND 0.8-1.5 MN/MAC AND 0-3.0 SI/MAC AND 0-3.0
FE/MAC AND 0.1-2.0 TI/MAC AND 0.1-2.0 ZR/MAC

=> s 3.5-4.5 MG/mac and 0.8-1.5 MN/mac and 0.0-3.0 SI/mac and 0.0-3.0 Fe/mac and 0.1-2.0 TI/mac and 0.1-2.0 ZR/mac

'MAC' IS NOT A VALID FIELD CODE

0 3.5-4.5 MG/MAC

0 0.8-1.5 MN/MAC

0 0.0-3.0 SI/MAC

0 0.0-3.0 FE/MAC

0 0.1-2.0 TI/MAC

0 0.1-2.0 ZR/MAC

L9 0 3.5-4.5 MG/MAC AND 0.8-1.5 MN/MAC AND 0.0-3.0 SI/MAC AND 0.0-3.0 FE/MAC AND 0.1-2.0 TI/MAC AND 0.1-2.0 ZR/MAC

=> s 3.5-4.5 Mg/mac and 0.8-1.5 Mn/mac and 0.0-1.0 SI/mac and 0.0-1.0 FE/mac

'MAC' IS NOT A VALID FIELD CODE

0 3.5-4.5 MG/MAC

0 0.8-1.5 MN/MAC

0 0.0-1.0 SI/MAC

0 0.0-1.0 FE/MAC

L10 0 3.5-4.5 MG/MAC AND 0.8-1.5 MN/MAC AND 0.0-1.0 SI/MAC AND 0.0-1.0 FE/MAC

=> file registry

COST IN U.S. DOLLARS

SINCE FILE

TOTAL

ENTRY

SESSION

FULL ESTIMATED COST

11.60

103.72

DISCOUNT AMOUNTS (FOR QUALIFYING ACCOUNTS)

SINCE FILE

TOTAL

ENTRY

SESSION

CA SUBSCRIBER PRICE

0.00

-1.60

FILE 'REGISTRY' ENTERED AT 09:49:41 ON 08 SEP 2008

USE IS SUBJECT TO THE TERMS OF YOUR STN CUSTOMER AGREEMENT.

PLEASE SEE "HELP USAGETERMS" FOR DETAILS.

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Property values tagged with IC are from the ZIC/VINITI data file provided by InfoChem.

STRUCTURE FILE UPDATES: 7 SEP 2008 HIGHEST RN 1047406-12-1

DICTIONARY FILE UPDATES: 7 SEP 2008 HIGHEST RN 1047406-12-1

New CAS Information Use Policies, enter HELP USAGETERMS for details.

TSCA INFORMATION NOW CURRENT THROUGH July 5, 2008.

Please note that search-term pricing does apply when conducting SmartSELECT searches.

REGISTRY includes numerically searchable data for experimental and predicted properties as well as tags indicating availability of experimental property data in the original document. For information on property searching in REGISTRY, refer to:

<http://www.cas.org/support/stngen/stndoc/properties.html>

=> s 3.5-4.5 Mg/mac and 0.8-1.5 Mn/mac and 0.0-1.0 SI/mac and 0.0-1.0 FE/mac

181008 3.5-4.5/MAC

79605 MG/MAC

5521 3.5-4.5 MG/MAC

(3.5-4.5/MAC (P) MG/MAC)

419377 0.8-1.5/MAC

```

385693 MN/MAC
134596 0.8-1.5 MN/MAC
      (0.8-1.5/MAC (P) MN/MAC)
606186 0.0-1.0/MAC
412336 SI/MAC
275399 0.0-1.0 SI/MAC
      (0.0-1.0/MAC (P) SI/MAC)
606186 0.0-1.0/MAC
548616 FE/MAC
56550 0.0-1.0 FE/MAC
      (0.0-1.0/MAC (P) FE/MAC)
L11      300 3.5-4.5 MG/MAC AND 0.8-1.5 MN/MAC AND 0.0-1.0 SI/MAC AND 0.0-1.0
      FE/MAC

```

=> s 3.5-4.5 Mg/mac and 0.8-1.5 Mn/mac and 0.0-1.0 SI/mac and 0.0-1.0 FE/mac and 80-100 AL/mac

```

181008 3.5-4.5/MAC
79605 MG/MAC
5521 3.5-4.5 MG/MAC
      (3.5-4.5/MAC (P) MG/MAC)
419377 0.8-1.5/MAC
385693 MN/MAC
134596 0.8-1.5 MN/MAC
      (0.8-1.5/MAC (P) MN/MAC)
606186 0.0-1.0/MAC
412336 SI/MAC
275399 0.0-1.0 SI/MAC
      (0.0-1.0/MAC (P) SI/MAC)
606186 0.0-1.0/MAC
548616 FE/MAC
56550 0.0-1.0 FE/MAC
      (0.0-1.0/MAC (P) FE/MAC)
514427 80-100/MAC
263320 AL/MAC
74008 80-100 AL/MAC
      (80-100/MAC (P) AL/MAC)
L12      295 3.5-4.5 MG/MAC AND 0.8-1.5 MN/MAC AND 0.0-1.0 SI/MAC AND 0.0-1.0
      FE/MAC AND 80-100 AL/MAC

```

=> s 3.5-4.5 Mg/mac and 0.8-1.5 Mn/mac and 0.0-1.0 SI/mac and 0.0-1.0 FE/mac and 80-100 AL/mac and 0.1-3.0 TI/mac and 0.1-3.0 Zr/mac

```

181008 3.5-4.5/MAC
79605 MG/MAC
5521 3.5-4.5 MG/MAC
      (3.5-4.5/MAC (P) MG/MAC)
419377 0.8-1.5/MAC
385693 MN/MAC
134596 0.8-1.5 MN/MAC
      (0.8-1.5/MAC (P) MN/MAC)
606186 0.0-1.0/MAC
412336 SI/MAC
275399 0.0-1.0 SI/MAC
      (0.0-1.0/MAC (P) SI/MAC)
606186 0.0-1.0/MAC
548616 FE/MAC
56550 0.0-1.0 FE/MAC
      (0.0-1.0/MAC (P) FE/MAC)
514427 80-100/MAC
263320 AL/MAC
74008 80-100 AL/MAC
      (80-100/MAC (P) AL/MAC)
714863 0.1-3.0/MAC

```

142635 TI/MAC
 84790 0.1-3.0 TI/MAC
 (0.1-3.0/MAC (P) TI/MAC)
 714863 0.1-3.0/MAC
 71325 ZR/MAC
 40560 0.1-3.0 ZR/MAC
 (0.1-3.0/MAC (P) ZR/MAC)
 L13 60 3.5-4.5 MG/MAC AND 0.8-1.5 MN/MAC AND 0.0-1.0 SI/MAC AND 0.0-1.0
 FE/MAC AND 80-100 AL/MAC AND 0.1-3.0 TI/MAC AND 0.1-3.0 ZR/MAC

=> d 113

L13 ANSWER 1 OF 60 REGISTRY COPYRIGHT 2008 ACS on STN
 RN 1045685-50-4 REGISTRY
 ED Entered STN: 02 Sep 2008
 CN INDEX NAME NOT YET ASSIGNED
 MF Al . Cr . Cu . Fe . Mg . Mn . Si . Ti . Zn . Zr
 CI AYS
 SR CA
 LC STN Files: CA, CAPLUS

Component	Component Percent	Component Registry Number
Al	91 - 97	7429-90-5
Mg	3 - 5	7439-95-4
Mn	0 - 1	7439-96-5
Cu	0 - 0.6	7440-50-8
Fe	0 - 0.5	7439-89-6
Si	0 - 0.5	7440-21-3
Zn	0 - 0.5	7440-66-6
Cr	0 - 0.4	7440-47-3
Zr	0 - 0.3	7440-67-7
Ti	0 - 0.2	7440-32-6

1 REFERENCES IN FILE CA (1907 TO DATE)
 1 REFERENCES IN FILE CAPLUS (1907 TO DATE)

=> d 113 2

L13 ANSWER 2 OF 60 REGISTRY COPYRIGHT 2008 ACS on STN
 RN 1043448-39-0 REGISTRY
 ED Entered STN: 25 Aug 2008
 CN Aluminum alloy, base, Al 86-99, Mg 0.6-4.5, Si 0.3-2.5, Cu 0-1.5, Fe 0-1.5, Zn
 0-1.5, Mn 0-1, Cr 0-0.5, Zr 0-0.5, V 0-0.3, Ti 0-0.2 (CA INDEX NAME)
 MF Al . Cr . Cu . Fe . Mg . Mn . Si . Ti . V . Zn . Zr
 CI AYS
 SR CA
 LC STN Files: CA, CAPLUS

Component	Component Percent	Component Registry Number
Al	86 - 99	7429-90-5
Mg	0.6 - 4.5	7439-95-4
Si	0.3 - 2.5	7440-21-3
Cu	0 - 1.5	7440-50-8
Fe	0 - 1.5	7439-89-6
Zn	0 - 1.5	7440-66-6
Mn	0 - 1	7439-96-5
Cr	0 - 0.5	7440-47-3

Zr	0	-	0.5	7440-67-7
V	0	-	0.3	7440-62-2
Ti	0	-	0.2	7440-32-6

1 REFERENCES IN FILE CA (1907 TO DATE)
1 REFERENCES IN FILE CAPLUS (1907 TO DATE)

=> d 113 3

L13 ANSWER 3 OF 60 REGISTRY COPYRIGHT 2008 ACS on STN
RN 1015163-38-8 REGISTRY
ED Entered STN: 17 Apr 2008
CN Aluminum alloy, base, Al 74-100, Zn 0-10, Mg 0.3-5, Si 0.2-2, Cr 0-2, Cu 0-2, Fe 0-1, Mn 0-1, Nb 0-1, V 0-1, Zr 0-1, Ti 0-0.5 (CA INDEX NAME)
MF Al . Cr . Cu . Fe . Mg . Mn . Nb . Si . Ti . V . Zn . Zr
CI AYS
SR CA
LC STN Files: CA, CAPLUS

Component	Component Percent	Component Registry Number
Al	74 - 100	7429-90-5
Zn	0 - 10	7440-66-6
Mg	0.3 - 5	7439-95-4
Si	0.2 - 2	7440-21-3
Cr	0 - 2	7440-47-3
Cu	0 - 2	7440-50-8
Fe	0 - 1	7439-89-6
Mn	0 - 1	7439-96-5
Nb	0 - 1	7440-03-1
V	0 - 1	7440-62-2
Zr	0 - 1	7440-67-7
Ti	0 - 0.5	7440-32-6

1 REFERENCES IN FILE CA (1907 TO DATE)
1 REFERENCES IN FILE CAPLUS (1907 TO DATE)

=> d 113 4

L13 ANSWER 4 OF 60 REGISTRY COPYRIGHT 2008 ACS on STN
RN 1001846-01-0 REGISTRY
ED Entered STN: 06 Feb 2008
CN Aluminum alloy, base, Al 82-99, Mn 1.1-7, Mg 0.1-6, Sc 0-1.5, Cr 0-0.5, Cu 0-0.5, Ni 0-0.5, Si 0-0.5, Co 0-0.1, Fe 0-0.1, La 0-0.1, Mo 0-0.1, Nb 0-0.1, Ti 0-0.1, V 0-0.1, W 0-0.1, Y 0-0.1, Zn 0-0.1, Zr 0-0.1 (CA INDEX NAME)
OTHER NAMES:
CN Mn 1.1-7, Mg 0.1-6, Sc 0-1.5, Si 0-0.5, Fe 0-0.1, Cu 0-0.5, Cr 0-0.5, Ni 0-0.5, Ti 0-0.1, V 0-0.1, Co 0-0.1, Zn 0-0.1, Zr 0-0.1, Nb 0-0.1, Mo 0-0.1, Y 0-0.1, W 0-0.1, La 0-0.1, Al bal.
MF Al . Co . Cr . Cu . Fe . La . Mg . Mn . Mo . Nb . Ni . Sc . Si . Ti . V . W . Y . Zn . Zr
CI AYS
SR CA
LC STN Files: CA, CAPLUS

Component	Component Percent	Component Registry Number
Al	82 - 99	7429-90-5

Mn	1.1	-	7	7439-96-5
Mg	0.1	-	6	7439-95-4
Sc	0	-	1.5	7440-20-2
Cr	0	-	0.5	7440-47-3
Cu	0	-	0.5	7440-50-8
Ni	0	-	0.5	7440-02-0
Si	0	-	0.5	7440-21-3
Co	0	-	0.1	7440-48-4
Fe	0	-	0.1	7439-89-6
La	0	-	0.1	7439-91-0
Mo	0	-	0.1	7439-98-7
Nb	0	-	0.1	7440-03-1
Ti	0	-	0.1	7440-32-6
V	0	-	0.1	7440-62-2
W	0	-	0.1	7440-33-7
Y	0	-	0.1	7440-65-5
Zn	0	-	0.1	7440-66-6
Zr	0	-	0.1	7440-67-7

PROPERTY DATA AVAILABLE IN THE 'PROP' FORMAT

2 REFERENCES IN FILE CA (1907 TO DATE)
2 REFERENCES IN FILE CAPLUS (1907 TO DATE)

=> d 113 5

L13 ANSWER 5 OF 60 REGISTRY COPYRIGHT 2008 ACS on STN
RN 954098-28-3 REGISTRY
ED Entered STN: 16 Nov 2007
CN Aluminum alloy, base, Al 76-99, Si 0-19, Cu 0-10, Mg 0-10, Zn 0-7.7, Sn
0-6.2, Ni 0-2.5, Fe 0.1-2, Mn 0-1.2, Zr 0-1, Cr 0-0.5, Ti 0-0.4, B 0-0.1, V 0-0.1
(CA INDEX NAME)
OTHER NAMES:
CN Aluminum 76-99, boron 0-0.1, chromium 0-0.5, copper 0-10, iron 0.1-2,
magnesium 0-10, manganese 0-1.2, nickel 0-2.5, silicon 0-19, titanium
0-0.4, vanadium 0-0.1, zinc 0-7.7, zirconium 0-1, tin 0-6.2
MF Al . B . Cr . Cu . Fe . Mg . Mn . Ni . Si . Sn . Ti . V . Zn . Zr
CI AYS
SR CA
LC STN Files: CA, CAPLUS

Component	Component Percent	Component Registry Number
Al	76 - 99	7429-90-5
Si	0 - 19	7440-21-3
Cu	0 - 10	7440-50-8
Mg	0 - 10	7439-95-4
Zn	0 - 7.7	7440-66-6
Sn	0 - 6.2	7440-31-5
Ni	0 - 2.5	7440-02-0
Fe	0.1 - 2	7439-89-6
Mn	0 - 1.2	7439-96-5
Zr	0 - 1	7440-67-7
Cr	0 - 0.5	7440-47-3
Ti	0 - 0.4	7440-32-6
B	0 - 0.1	7440-42-8
V	0 - 0.1	7440-62-2

1 REFERENCES IN FILE CA (1907 TO DATE)

1 REFERENCES IN FILE CAPLUS (1907 TO DATE)

=> d 113 6

L13 ANSWER 6 OF 60 REGISTRY COPYRIGHT 2008 ACS on STN
 RN 952105-95-2 REGISTRY
 ED Entered STN: 31 Oct 2007
 CN Aluminum alloy, base, Al 92-95, Mg 4-5.2, Mn 0.7-1, Zn 0-0.4, Cr 0-0.2, Cu 0-0.2, Fe 0-0.2, Si 0-0.2, Ti 0-0.2, Zr 0-0.2 (CA INDEX NAME)
 MF Al . Cr . Cu . Fe . Mg . Mn . Si . Ti . Zn . Zr
 CI AYS
 SR CA
 LC STN Files: CA, CAPLUS

Component	Component Percent	Component Registry Number
Al	92 - 95	7429-90-5
Mg	4 - 5.2	7439-95-4
Mn	0.7 - 1	7439-96-5
Zn	0 - 0.4	7440-66-6
Cr	0 - 0.2	7440-47-3
Cu	0 - 0.2	7440-50-8
Fe	0 - 0.2	7439-89-6
Si	0 - 0.2	7440-21-3
Ti	0 - 0.2	7440-32-6
Zr	0 - 0.2	7440-67-7

PROPERTY DATA AVAILABLE IN THE 'PROP' FORMAT

1 REFERENCES IN FILE CA (1907 TO DATE)
 1 REFERENCES IN FILE CAPLUS (1907 TO DATE)

=> d 113 7

L13 ANSWER 7 OF 60 REGISTRY COPYRIGHT 2008 ACS on STN
 RN 949114-99-2 REGISTRY
 ED Entered STN: 03 Oct 2007
 CN Aluminum alloy, base, Al 81-97, Mg 3-14, Fe 0-1, Mn 0-1, Cr 0-0.5, Cu 0-0.5, Si 0-0.5, Ti 0-0.5, Zn 0-0.5, V 0-0.3, Zr 0-0.3 (CA INDEX NAME)
 OTHER NAMES:
 CN Aluminum 81-97, chromium 0-0.5, copper 0-0.5, iron 0-1, magnesium 3-14, manganese 0-1, silicon 0-0.5, titanium 0-0.5, vanadium 0-0.3, zinc 0-0.5, zirconium 0-0.3
 MF Al . Cr . Cu . Fe . Mg . Mn . Si . Ti . V . Zn . Zr
 CI AYS
 SR CA
 LC STN Files: CA, CAPLUS

Component	Component Percent	Component Registry Number
Al	81 - 97	7429-90-5
Mg	3 - 14	7439-95-4
Fe	0 - 1	7439-89-6
Mn	0 - 1	7439-96-5
Cr	0 - 0.5	7440-47-3
Cu	0 - 0.5	7440-50-8
Si	0 - 0.5	7440-21-3

Ti	0	-	0.5	7440-32-6
Zn	0	-	0.5	7440-66-6
V	0	-	0.3	7440-62-2
Zr	0	-	0.3	7440-67-7

1 REFERENCES IN FILE CA (1907 TO DATE)
1 REFERENCES IN FILE CAPLUS (1907 TO DATE)

=> s 3.5-4.5 Mg/mac and 0.8-1.5 Mn/mac and 0.0-1.0 SI/mac and 0.0-1.0 FE/mac and 80-100 AL/mac and 0.2-3.0 TI/mac and 0.3-3.0 Zr/mac

181008 3.5-4.5/MAC
79605 MG/MAC
5521 3.5-4.5 MG/MAC
(3.5-4.5/MAC (P) MG/MAC)
419377 0.8-1.5/MAC
385693 MN/MAC
134596 0.8-1.5 MN/MAC
(0.8-1.5/MAC (P) MN/MAC)
606186 0.0-1.0/MAC
412336 SI/MAC
275399 0.0-1.0 SI/MAC
(0.0-1.0/MAC (P) SI/MAC)
606186 0.0-1.0/MAC
548616 FE/MAC
56550 0.0-1.0 FE/MAC
(0.0-1.0/MAC (P) FE/MAC)
514427 80-100/MAC
263320 AL/MAC
74008 80-100 AL/MAC
(80-100/MAC (P) AL/MAC)
699880 0.2-3.0/MAC
142635 TI/MAC
62383 0.2-3.0 TI/MAC
(0.2-3.0/MAC (P) TI/MAC)
681560 0.3-3.0/MAC
71325 ZR/MAC
19205 0.3-3.0 ZR/MAC
(0.3-3.0/MAC (P) ZR/MAC)

L14 32 3.5-4.5 MG/MAC AND 0.8-1.5 MN/MAC AND 0.0-1.0 SI/MAC AND 0.0-1.0 FE/MAC AND 80-100 AL/MAC AND 0.2-3.0 TI/MAC AND 0.3-3.0 ZR/MAC

=> d l14

L14 ANSWER 1 OF 32 REGISTRY COPYRIGHT 2008 ACS on STN
RN 1045685-50-4 REGISTRY
ED Entered STN: 02 Sep 2008
CN INDEX NAME NOT YET ASSIGNED
MF Al . Cr . Cu . Fe . Mg . Mn . Si . Ti . Zn . Zr
CI AYS
SR CA
LC STN Files: CA, CAPLUS

Component	Component Percent	Component Registry Number
Al	91 - 97	7429-90-5
Mg	3 - 5	7439-95-4
Mn	0 - 1	7439-96-5
Cu	0 - 0.6	7440-50-8
Fe	0 - 0.5	7439-89-6
Si	0 - 0.5	7440-21-3

Zn	0	-	0.5	7440-66-6
Cr	0	-	0.4	7440-47-3
Zr	0	-	0.3	7440-67-7
Ti	0	-	0.2	7440-32-6

1 REFERENCES IN FILE CA (1907 TO DATE)
1 REFERENCES IN FILE CAPLUS (1907 TO DATE)

=> d 114 2

L14 ANSWER 2 OF 32 REGISTRY COPYRIGHT 2008 ACS on STN
RN 1043448-39-0 REGISTRY
ED Entered STN: 25 Aug 2008
CN Aluminum alloy, base, Al 86-99,Mg 0.6-4.5,Si 0.3-2.5,Cu 0-1.5,Fe 0-1.5,Zn 0-1.5,Mn 0-1,Cr 0-0.5,Zr 0-0.5,V 0-0.3,Ti 0-0.2 (CA INDEX NAME)
MF Al . Cr . Cu . Fe . Mg . Mn . Si . Ti . V . Zn . Zr
CI AYS
SR CA
LC STN Files: CA, CAPLUS

Component	Component Percent	Component Registry Number
Al	86 - 99	7429-90-5
Mg	0.6 - 4.5	7439-95-4
Si	0.3 - 2.5	7440-21-3
Cu	0 - 1.5	7440-50-8
Fe	0 - 1.5	7439-89-6
Zn	0 - 1.5	7440-66-6
Mn	0 - 1	7439-96-5
Cr	0 - 0.5	7440-47-3
Zr	0 - 0.5	7440-67-7
V	0 - 0.3	7440-62-2
Ti	0 - 0.2	7440-32-6

1 REFERENCES IN FILE CA (1907 TO DATE)
1 REFERENCES IN FILE CAPLUS (1907 TO DATE)

=> d 114 3

L14 ANSWER 3 OF 32 REGISTRY COPYRIGHT 2008 ACS on STN
RN 1015163-38-8 REGISTRY
ED Entered STN: 17 Apr 2008
CN Aluminum alloy, base, Al 74-100,Zn 0-10,Mg 0.3-5,Si 0.2-2,Cr 0-2,Cu 0-2,Fe 0-1,Mn 0-1,Nb 0-1,V 0-1,Zr 0-1,Ti 0-0.5 (CA INDEX NAME)
MF Al . Cr . Cu . Fe . Mg . Mn . Nb . Si . Ti . V . Zn . Zr
CI AYS
SR CA
LC STN Files: CA, CAPLUS

Component	Component Percent	Component Registry Number
Al	74 - 100	7429-90-5
Zn	0 - 10	7440-66-6
Mg	0.3 - 5	7439-95-4
Si	0.2 - 2	7440-21-3
Cr	0 - 2	7440-47-3
Cu	0 - 2	7440-50-8
Fe	0 - 1	7439-89-6

Mn	0	-	1	7439-96-5
Nb	0	-	1	7440-03-1
V	0	-	1	7440-62-2
Zr	0	-	1	7440-67-7
Ti	0	-	0.5	7440-32-6

1 REFERENCES IN FILE CA (1907 TO DATE)
1 REFERENCES IN FILE CAPLUS (1907 TO DATE)

=> d 114 4

L14 ANSWER 4 OF 32 REGISTRY COPYRIGHT 2008 ACS on STN
RN 954098-28-3 REGISTRY
ED Entered STN: 16 Nov 2007
CN Aluminum alloy, base, Al 76-99, Si 0-19, Cu 0-10, Mg 0-10, Zn 0-7.7, Sn 0-6.2, Ni 0-2.5, Fe 0.1-2, Mn 0-1.2, Zr 0-1, Cr 0-0.5, Ti 0-0.4, B 0-0.1, V 0-0.1 (CA INDEX NAME)
OTHER NAMES:
CN Aluminum 76-99, boron 0-0.1, chromium 0-0.5, copper 0-10, iron 0.1-2, magnesium 0-10, manganese 0-1.2, nickel 0-2.5, silicon 0-19, titanium 0-0.4, vanadium 0-0.1, zinc 0-7.7, zirconium 0-1, tin 0-6.2
MF Al . B . Cr . Cu . Fe . Mg . Mn . Ni . Si . Sn . Ti . V . Zn . Zr
CI AYS
SR CA
LC STN Files: CA, CAPLUS

Component	Component Percent	Component Registry Number
Al	76 - 99	7429-90-5
Si	0 - 19	7440-21-3
Cu	0 - 10	7440-50-8
Mg	0 - 10	7439-95-4
Zn	0 - 7.7	7440-66-6
Sn	0 - 6.2	7440-31-5
Ni	0 - 2.5	7440-02-0
Fe	0.1 - 2	7439-89-6
Mn	0 - 1.2	7439-96-5
Zr	0 - 1	7440-67-7
Cr	0 - 0.5	7440-47-3
Ti	0 - 0.4	7440-32-6
B	0 - 0.1	7440-42-8
V	0 - 0.1	7440-62-2

1 REFERENCES IN FILE CA (1907 TO DATE)
1 REFERENCES IN FILE CAPLUS (1907 TO DATE)

=> d 114 5

L14 ANSWER 5 OF 32 REGISTRY COPYRIGHT 2008 ACS on STN
RN 949114-99-2 REGISTRY
ED Entered STN: 03 Oct 2007
CN Aluminum alloy, base, Al 81-97, Mg 3-14, Fe 0-1, Mn 0-1, Cr 0-0.5, Cu 0-0.5, Si 0-0.5, Ti 0-0.5, Zn 0-0.5, V 0-0.3, Zr 0-0.3 (CA INDEX NAME)
OTHER NAMES:
CN Aluminum 81-97, chromium 0-0.5, copper 0-0.5, iron 0-1, magnesium 3-14, manganese 0-1, silicon 0-0.5, titanium 0-0.5, vanadium 0-0.3, zinc 0-0.5, zirconium 0-0.3
MF Al . Cr . Cu . Fe . Mg . Mn . Si . Ti . V . Zn . Zr
CI AYS

SR CA
LC STN Files: CA, CAPLUS

Component	Component Percent	Component Registry Number
Al	81 - 97	7429-90-5
Mg	3 - 14	7439-95-4
Fe	0 - 1	7439-89-6
Mn	0 - 1	7439-96-5
Cr	0 - 0.5	7440-47-3
Cu	0 - 0.5	7440-50-8
Si	0 - 0.5	7440-21-3
Ti	0 - 0.5	7440-32-6
Zn	0 - 0.5	7440-66-6
V	0 - 0.3	7440-62-2
Zr	0 - 0.3	7440-67-7

1 REFERENCES IN FILE CA (1907 TO DATE)
1 REFERENCES IN FILE CAPLUS (1907 TO DATE)

=> d 114 6

L14 ANSWER 6 OF 32 REGISTRY COPYRIGHT 2008 ACS on STN
RN 936561-06-7 REGISTRY
ED Entered STN: 05 Jun 2007
CN Aluminum alloy, base, Al 86-94, Cu 4.5-7, Mg 1.8-4.5, Mn 0.2-0.8, Co 0-0.4, Fe 0-0.4, Ti 0-0.4, Zr 0-0.4, Si 0-0.2, Be 0-0.1, Ca 0-0.1 (CA INDEX NAME)
MF Al . Be . Ca . Co . Cu . Fe . Mg . Mn . Si . Ti . Zr
CI AYS
SR CA
LC STN Files: CA, CAPLUS

Component	Component Percent	Component Registry Number
Al	86 - 94	7429-90-5
Cu	4.5 - 7	7440-50-8
Mg	1.8 - 4.5	7439-95-4
Mn	0.2 - 0.8	7439-96-5
Co	0 - 0.4	7440-48-4
Fe	0 - 0.4	7439-89-6
Ti	0 - 0.4	7440-32-6
Zr	0 - 0.4	7440-67-7
Si	0 - 0.2	7440-21-3
Be	0 - 0.1	7440-41-7
Ca	0 - 0.1	7440-70-2

1 REFERENCES IN FILE CA (1907 TO DATE)
1 REFERENCES IN FILE CAPLUS (1907 TO DATE)

=> d 114 7

L14 ANSWER 7 OF 32 REGISTRY COPYRIGHT 2008 ACS on STN
RN 926624-86-4 REGISTRY
ED Entered STN: 16 Mar 2007
CN Aluminum alloy, base, Al 88-96, Mg 3.5-6, Zn 0-1.7, Mn 0.4-1.2, Fe 0-0.5, Li 0-0.5, Sc 0-0.5, Si 0-0.5, Zr 0-0.5, Ag 0-0.4, Cr 0-0.3, Cu 0-0.2, Ti 0-0.2 (CA INDEX NAME)
OTHER NAMES:

CN Aluminum 88-96, chromium 0-0.3, copper 0-0.2, iron 0-0.5, lithium 0-0.5, magnesium 3.5-6, manganese 0.4-1.2, scandium 0-0.5, silicon 0-0.5, silver 0-0.4, titanium 0-0.2, zinc 0-1.7, zirconium 0-0.5
 MF Ag . Al . Cr . Cu . Fe . Li . Mg . Mn . Sc . Si . Ti . Zn . Zr
 CI AYS
 SR CA
 LC STN Files: CA, CAPLUS

Component	Component Percent	Component Registry Number
Al	88 - 96	7429-90-5
Mg	3.5 - 6	7439-95-4
Zn	0 - 1.7	7440-66-6
Mn	0.4 - 1.2	7439-96-5
Fe	0 - 0.5	7439-89-6
Li	0 - 0.5	7439-93-2
Sc	0 - 0.5	7440-20-2
Si	0 - 0.5	7440-21-3
Zr	0 - 0.5	7440-67-7
Ag	0 - 0.4	7440-22-4
Cr	0 - 0.3	7440-47-3
Cu	0 - 0.2	7440-50-8
Ti	0 - 0.2	7440-32-6

1 REFERENCES IN FILE CA (1907 TO DATE)
 1 REFERENCES IN FILE CAPLUS (1907 TO DATE)

=> d 114 8

L14 ANSWER 8 OF 32 REGISTRY COPYRIGHT 2008 ACS on STN
 RN 918789-28-3 REGISTRY
 ED Entered STN: 30 Jan 2007
 CN Aluminum alloy, base, Al 81-98, Zn 1.4-8.4, Mg 0.3-4, Cu 0.1-3, Mn 0.1-0.9, Si 0.1-0.8, Fe 0.1-0.7, Cr 0.1-0.4, Zr 0-0.3, Ni 0-0.2, Ti 0-0.2 (CA INDEX NAME)
 OTHER NAMES:
 CN Aluminum 81-98, chromium 0.1-0.4, copper 0.1-3, iron 0.1-0.7, magnesium 0.3-4, manganese 0.1-0.9, nickel 0-0.2, silicon 0.1-0.8, titanium 0-0.2, zinc 1.4-8.4, zirconium 0-0.3
 MF Al . Cr . Cu . Fe . Mg . Mn . Ni . Si . Ti . Zn . Zr
 CI AYS
 SR CA
 LC STN Files: CA, CAPLUS

Component	Component Percent	Component Registry Number
Al	81 - 98	7429-90-5
Zn	1.4 - 8.4	7440-66-6
Mg	0.3 - 4	7439-95-4
Cu	0.1 - 3	7440-50-8
Mn	0.1 - 0.9	7439-96-5
Si	0.1 - 0.8	7440-21-3
Fe	0.1 - 0.7	7439-89-6
Cr	0.1 - 0.4	7440-47-3
Zr	0 - 0.3	7440-67-7
Ni	0 - 0.2	7440-02-0
Ti	0 - 0.2	7440-32-6

1 REFERENCES IN FILE CA (1907 TO DATE)
 1 REFERENCES IN FILE CAPLUS (1907 TO DATE)

=> d 114 9

L14 ANSWER 9 OF 32 REGISTRY COPYRIGHT 2008 ACS on STN
RN 902164-07-2 REGISTRY
ED Entered STN: 17 Aug 2006
CN Aluminum alloy, base, Al 90-95, Mg 4.2-6.5, Mn 0.5-1.2, Fe 0-0.3, Sc 0-0.3, Zr 0-0.3, Cr 0-0.2, Si 0-0.2, Ti 0-0.2, Zn 0-0.2, Ce 0-0.1, Cu 0-0.1, Nd 0-0.1, Y 0-0.1 (9CI) (CA INDEX NAME)
MF Al . Ce . Cr . Cu . Fe . Mg . Mn . Nd . Sc . Si . Ti . Y . Zn . Zr
CI AYS
SR CA
LC STN Files: CA, CAPLUS

Component	Component Percent	Component Registry Number
Al	90 - 95	7429-90-5
Mg	4.2 - 6.5	7439-95-4
Mn	0.5 - 1.2	7439-96-5
Fe	0 - 0.3	7439-89-6
Sc	0 - 0.3	7440-20-2
Zr	0 - 0.3	7440-67-7
Cr	0 - 0.2	7440-47-3
Si	0 - 0.2	7440-21-3
Ti	0 - 0.2	7440-32-6
Zn	0 - 0.2	7440-66-6
Ce	0 - 0.1	7440-45-1
Cu	0 - 0.1	7440-50-8
Nd	0 - 0.1	7440-00-8
Y	0 - 0.1	7440-65-5

1 REFERENCES IN FILE CA (1907 TO DATE)
1 REFERENCES IN FILE CAPLUS (1907 TO DATE)

=> d 114 10

L14 ANSWER 10 OF 32 REGISTRY COPYRIGHT 2008 ACS on STN
RN 886758-22-1 REGISTRY
ED Entered STN: 05 Jun 2006
CN Aluminum alloy, base, Al 92-96, Mg 3.5-5, Mn 0-0.8, Zn 0-0.6, Fe 0-0.4, Cr 0-0.3, Zr 0-0.3, Cu 0-0.2, Si 0-0.2, Ti 0-0.2 (9CI) (CA INDEX NAME)
MF Al . Cr . Cu . Fe . Mg . Mn . Si . Ti . Zn . Zr
CI AYS
SR CA
LC STN Files: CA, CAPLUS

Component	Component Percent	Component Registry Number
Al	92 - 96	7429-90-5
Mg	3.5 - 5	7439-95-4
Mn	0 - 0.8	7439-96-5
Zn	0 - 0.6	7440-66-6
Fe	0 - 0.4	7439-89-6
Cr	0 - 0.3	7440-47-3
Zr	0 - 0.3	7440-67-7
Cu	0 - 0.2	7440-50-8
Si	0 - 0.2	7440-21-3
Ti	0 - 0.2	7440-32-6

1 REFERENCES IN FILE CA (1907 TO DATE)
1 REFERENCES IN FILE CAPLUS (1907 TO DATE)

=> d 114 11

L14 ANSWER 11 OF 32 REGISTRY COPYRIGHT 2008 ACS on STN
RN 851905-53-8 REGISTRY
ED Entered STN: 08 Jun 2005
CN Aluminum alloy, base, Al 80-98, Mg 1-8, Si 1-4, Cu 0-1, Mn 0-0.8, Fe 0-0.6, Er
0-0.5, Gd 0-0.5, Hf 0-0.5, Mo 0-0.5, Nb 0-0.5, Sc 0-0.5, Tb 0-0.5, V 0-0.5, Zr
0-0.5, Cr 0-0.3, Ti 0-0.2, Zn 0-0.1 (9CI) (CA INDEX NAME)
MF Al . Cr . Cu . Er . Fe . Gd . Hf . Mg . Mn . Mo . Nb . Sc . Si . Tb . Ti .
V . Zn . Zr
CI AYS
SR CA
LC STN Files: CA, CAPLUS, USPATFULL

Component	Component Percent	Component Registry Number
=====+	=====+	=====
Al	80 - 98	7429-90-5
Mg	1 - 8	7439-95-4
Si	1 - 4	7440-21-3
Cu	0 - 1	7440-50-8
Mn	0 - 0.8	7439-96-5
Fe	0 - 0.6	7439-89-6
Er	0 - 0.5	7440-52-0
Gd	0 - 0.5	7440-54-2
Hf	0 - 0.5	7440-58-6
Mo	0 - 0.5	7439-98-7
Nb	0 - 0.5	7440-03-1
Sc	0 - 0.5	7440-20-2
Tb	0 - 0.5	7440-27-9
V	0 - 0.5	7440-62-2
Zr	0 - 0.5	7440-67-7
Cr	0 - 0.3	7440-47-3
Ti	0 - 0.2	7440-32-6
Zn	0 - 0.1	7440-66-6

1 REFERENCES IN FILE CA (1907 TO DATE)
1 REFERENCES IN FILE CAPLUS (1907 TO DATE)

=> d 114 12

L14 ANSWER 12 OF 32 REGISTRY COPYRIGHT 2008 ACS on STN
RN 833466-05-0 REGISTRY
ED Entered STN: 18 Feb 2005
CN Aluminum alloy, base, Al 94-96, Mg 3-3.5, Mn 0.5-1, Ti 0-0.5, Zr 0.1-0.4, Fe
0-0.2, Si 0-0.2, Cu 0-0.1 (9CI) (CA INDEX NAME)
MF Al . Cu . Fe . Mg . Mn . Si . Ti . Zr
CI AYS
SR CA
LC STN Files: CA, CAPLUS

Component	Component Percent	Component Registry Number
=====+	=====+	=====
Al	94 - 96	7429-90-5
Mg	3 - 3.5	7439-95-4

Mn	0.5	-	1	7439-96-5
Ti	0	-	0.5	7440-32-6
Zr	0.1	-	0.4	7440-67-7
Fe	0	-	0.2	7439-89-6
Si	0	-	0.2	7440-21-3
Cu	0	-	0.1	7440-50-8

1 REFERENCES IN FILE CA (1907 TO DATE)
1 REFERENCES IN FILE CAPLUS (1907 TO DATE)

=> d 114 13

L14 ANSWER 13 OF 32 REGISTRY COPYRIGHT 2008 ACS on STN
RN 749250-15-5 REGISTRY
ED Entered STN: 22 Sep 2004
CN Aluminum alloy, base, Al 90-98, Mg 2-6, Mn 0-1, Fe 0-0.7, Cu 0-0.6, Si 0-0.5, Zn 0-0.5, Cr 0-0.4, Zr 0-0.3, Ti 0-0.2 (9CI) (CA INDEX NAME)
MF Al . Cr . Cu . Fe . Mg . Mn . Si . Ti . Zn . Zr
CI AYS
SR CA
LC STN Files: CA, CAPLUS

Component	Component Percent	Component Registry Number
Al	90 - 98	7429-90-5
Mg	2 - 6	7439-95-4
Mn	0 - 1	7439-96-5
Fe	0 - 0.7	7439-89-6
Cu	0 - 0.6	7440-50-8
Si	0 - 0.5	7440-21-3
Zn	0 - 0.5	7440-66-6
Cr	0 - 0.4	7440-47-3
Zr	0 - 0.3	7440-67-7
Ti	0 - 0.2	7440-32-6

1 REFERENCES IN FILE CA (1907 TO DATE)
1 REFERENCES IN FILE CAPLUS (1907 TO DATE)

=> d 114 14

L14 ANSWER 14 OF 32 REGISTRY COPYRIGHT 2008 ACS on STN
RN 627892-61-9 REGISTRY
ED Entered STN: 19 Dec 2003
CN Aluminum alloy, base, Al 94, Mg 4, Mn 1, Zr 0.3, Fe 0.2, Si 0.2, Ti 0.2 (9CI) (CA INDEX NAME)
MF Al . Fe . Mg . Mn . Si . Ti . Zr
CI AYS
SR CA
LC STN Files: CA, CAPLUS, USPATFULL

Component	Component Percent	Component Registry Number
Al	94	7429-90-5
Mg	4	7439-95-4
Mn	1	7439-96-5
Zr	0.3	7440-67-7
Fe	0.2	7439-89-6
Si	0.2	7440-21-3

Ti 0.2 7440-32-6

PROPERTY DATA AVAILABLE IN THE 'PROP' FORMAT

1 REFERENCES IN FILE CA (1907 TO DATE)
1 REFERENCES IN FILE CAPLUS (1907 TO DATE)

=> d 114 15

L14 ANSWER 15 OF 32 REGISTRY COPYRIGHT 2008 ACS on STN
RN 528578-87-2 REGISTRY
ED Entered STN: 10 Jun 2003
CN Aluminum alloy, base, Al 89-98,Mg 2-6,Mn 0-1.5,B 0-0.5,Fe 0-0.5,Si
0-0.5,Ti 0-0.5,Cr 0-0.4,Zn 0-0.4,Zr 0-0.4,Cu 0-0.2 (9CI) (CA INDEX NAME)
MF Al . B . Cr . Cu . Fe . Mg . Mn . Si . Ti . Zn . Zr
CI AYS
SR CA
LC STN Files: CA, CAPLUS

Component	Component Percent	Component Registry Number
Al	89 - 98	7429-90-5
Mg	2 - 6	7439-95-4
Mn	0 - 1.5	7439-96-5
B	0 - 0.5	7440-42-8
Fe	0 - 0.5	7439-89-6
Si	0 - 0.5	7440-21-3
Ti	0 - 0.5	7440-32-6
Cr	0 - 0.4	7440-47-3
Zn	0 - 0.4	7440-66-6
Zr	0 - 0.4	7440-67-7
Cu	0 - 0.2	7440-50-8

1 REFERENCES IN FILE CA (1907 TO DATE)
1 REFERENCES IN FILE CAPLUS (1907 TO DATE)

=> d 114 16

L14 ANSWER 16 OF 32 REGISTRY COPYRIGHT 2008 ACS on STN
RN 527685-44-5 REGISTRY
ED Entered STN: 09 Jun 2003
CN Aluminum alloy, base, Al 80-96,Zn 3.5-7.5,Fe 0-5.5,Mg 0.5-4,Mn 0-0.8,B
0-0.5,Si 0-0.5,Ti 0-0.5,Cr 0-0.4,Zr 0-0.4,V 0-0.2 (9CI) (CA INDEX NAME)
MF Al . B . Cr . Fe . Mg . Mn . Si . Ti . V . Zn . Zr
CI AYS
SR CA
LC STN Files: CA, CAPLUS

Component	Component Percent	Component Registry Number
Al	80 - 96	7429-90-5
Zn	3.5 - 7.5	7440-66-6
Fe	0 - 5.5	7439-89-6
Mg	0.5 - 4	7439-95-4
Mn	0 - 0.8	7439-96-5
B	0 - 0.5	7440-42-8
Si	0 - 0.5	7440-21-3

Ti	0	-	0.5	7440-32-6
Cr	0	-	0.4	7440-47-3
Zr	0	-	0.4	7440-67-7
V	0	-	0.2	7440-62-2

1 REFERENCES IN FILE CA (1907 TO DATE)
1 REFERENCES IN FILE CAPLUS (1907 TO DATE)

=> d 114 17

L14 ANSWER 17 OF 32 REGISTRY COPYRIGHT 2008 ACS on STN
RN 494837-73-9 REGISTRY
ED Entered STN: 26 Feb 2003
CN Aluminum alloy, base, Al 76-99, Si 0.3-10, Mg 0.3-6, Cr 0-1, Cu 0-1, Fe 0-1, Mn 0-1, Ti 0-1, Zn 0-1, Zr 0-1, Ca 0-0.5, Na 0-0.5, Sr 0-0.3 (9CI) (CA INDEX NAME)
MF Al . Ca . Cr . Cu . Fe . Mg . Mn . Na . Si . Sr . Ti . Zn . Zr
CI AYS
SR CA
LC STN Files: CA, CAPLUS, USPATFULL

Component	Component Percent	Component Registry Number
Al	76 - 99	7429-90-5
Si	0.3 - 10	7440-21-3
Mg	0.3 - 6	7439-95-4
Cr	0 - 1	7440-47-3
Cu	0 - 1	7440-50-8
Fe	0 - 1	7439-89-6
Mn	0 - 1	7439-96-5
Ti	0 - 1	7440-32-6
Zn	0 - 1	7440-66-6
Zr	0 - 1	7440-67-7
Ca	0 - 0.5	7440-70-2
Na	0 - 0.5	7440-23-5
Sr	0 - 0.3	7440-24-6

PROPERTY DATA AVAILABLE IN THE 'PROP' FORMAT

1 REFERENCES IN FILE CA (1907 TO DATE)
1 REFERENCES IN FILE CAPLUS (1907 TO DATE)

=> d 114 18

L14 ANSWER 18 OF 32 REGISTRY COPYRIGHT 2008 ACS on STN
RN 440626-49-3 REGISTRY
ED Entered STN: 29 Jul 2002
CN Aluminum alloy, base, Al 90-96, Mg 2.5-4.5, Si 1-3.5, Mn 0.3-1.5, Zr 0-0.3, Fe 0-0.2, Ti 0-0.2 (9CI) (CA INDEX NAME)
MF Al . Fe . Mg . Mn . Si . Ti . Zr
CI AYS
SR CA
LC STN Files: CA, CAPLUS, USPATFULL

Component	Component Percent	Component Registry Number
Al	90 - 96	7429-90-5
Mg	2.5 - 4.5	7439-95-4

Si	1	-	3.5	7440-21-3
Mn	0.3	-	1.5	7439-96-5
Zr	0	-	0.3	7440-67-7
Fe	0	-	0.2	7439-89-6
Ti	0	-	0.2	7440-32-6

1 REFERENCES IN FILE CA (1907 TO DATE)
1 REFERENCES IN FILE CAPLUS (1907 TO DATE)

=> d 114 19

L14 ANSWER 19 OF 32 REGISTRY COPYRIGHT 2008 ACS on STN
RN 389626-12-4 REGISTRY
ED Entered STN: 05 Feb 2002
CN Aluminum alloy, base, Al 85-100,Mg 0-5,Zn 0-5,Fe 0-1,Mn 0-1,Ni 0-1,Si 0-1,Ti 0-0.3,Zr 0-0.3 (9CI) (CA INDEX NAME)
MF Al . Fe . Mg . Mn . Ni . Si . Ti . Zn . Zr
CI AYS
SR CA
LC STN Files: CA, CAPLUS

Component	Component Percent	Component Registry Number
Al	85 - 100	7429-90-5
Mg	0 - 5	7439-95-4
Zn	0 - 5	7440-66-6
Fe	0 - 1	7439-89-6
Mn	0 - 1	7439-96-5
Ni	0 - 1	7440-02-0
Si	0 - 1	7440-21-3
Ti	0 - 0.3	7440-32-6
Zr	0 - 0.3	7440-67-7

1 REFERENCES IN FILE CA (1907 TO DATE)
1 REFERENCES IN FILE CAPLUS (1907 TO DATE)

=> d 114 20

L14 ANSWER 20 OF 32 REGISTRY COPYRIGHT 2008 ACS on STN
RN 371165-10-5 REGISTRY
ED Entered STN: 20 Nov 2001
CN Aluminum alloy, base, Al 65-95,Zn 2.5-10,Zr 0.2-6.5,Mg 1-6.4,Cr 0.1-2.5,Mn 0.1-2.5,Cu 1-1.7,Si 0-1.5,Fe 0.1-1.2,Ni 0.2-1,B 0.1-1,Ti 0-0.5 (9CI) (CA INDEX NAME)
MF Al . B . Cr . Cu . Fe . Mg . Mn . Ni . Si . Ti . Zn . Zr
CI AYS
SR CA
LC STN Files: CA, CAPLUS, USPATFULL

Component	Component Percent	Component Registry Number
Al	65 - 95	7429-90-5
Zn	2.5 - 10	7440-66-6
Zr	0.2 - 6.5	7440-67-7
Mg	1 - 6.4	7439-95-4
Cr	0.1 - 2.5	7440-47-3
Mn	0.1 - 2.5	7439-96-5
Cu	1 - 1.7	7440-50-8

Si	0	-	1.5	7440-21-3
Fe	0.1	-	1.2	7439-89-6
Ni	0.2	-	1	7440-02-0
B	0.1	-	1	7440-42-8
Ti	0	-	0.5	7440-32-6

1 REFERENCES IN FILE CA (1907 TO DATE)
1 REFERENCES IN FILE CAPLUS (1907 TO DATE)

=> d 114 21

L14 ANSWER 21 OF 32 REGISTRY COPYRIGHT 2008 ACS on STN
RN 361484-70-0 REGISTRY
ED Entered STN: 11 Oct 2001
CN Aluminum alloy, base, Al 88-97, Mg 2.7-6, Zn 0.1-1.5, Mn 0.4-1.4, Si 0-1.4, Fe 0-1, Sc 0-0.3, V 0-0.3, Zr 0-0.3, Ti 0-0.2 (9CI) (CA INDEX NAME)
MF Al . Fe . Mg . Mn . Sc . Si . Ti . V . Zn . Zr
CI AYS
SR CA
LC STN Files: CA, CAPLUS, USPAT2, USPATFULL

Component	Component Percent	Component Registry Number
Al	88 - 97	7429-90-5
Mg	2.7 - 6	7439-95-4
Zn	0.1 - 1.5	7440-66-6
Mn	0.4 - 1.4	7439-96-5
Si	0 - 1.4	7440-21-3
Fe	0 - 1	7439-89-6
Sc	0 - 0.3	7440-20-2
V	0 - 0.3	7440-62-2
Zr	0 - 0.3	7440-67-7
Ti	0 - 0.2	7440-32-6

1 REFERENCES IN FILE CA (1907 TO DATE)
1 REFERENCES IN FILE CAPLUS (1907 TO DATE)

=> d 114 22

L14 ANSWER 22 OF 32 REGISTRY COPYRIGHT 2008 ACS on STN
RN 267005-59-4 REGISTRY
ED Entered STN: 26 May 2000
CN Aluminum alloy, base, Al 85-97, Mg 1.5-6, Zn 0.4-5, Mn 0.3-1.4, Fe 0-0.5, Si 0-0.5, Ag 0.4, Cu 0-0.4, Cr 0-0.3, Zr 0-0.3, Ti 0-0.2, V 0-0.2 (9CI) (CA INDEX NAME)
MF Ag . Al . Cr . Cu . Fe . Mg . Mn . Si . Ti . V . Zn . Zr
CI AYS
SR CA
LC STN Files: CA, CAPLUS, USPATFULL

Component	Component Percent	Component Registry Number
Al	85 - 97	7429-90-5
Mg	1.5 - 6	7439-95-4
Zn	0.4 - 5	7440-66-6
Mn	0.3 - 1.4	7439-96-5
Fe	0 - 0.5	7439-89-6
Si	0 - 0.5	7440-21-3

Ag	0.4		7440-22-4
Cu	0	- 0.4	7440-50-8
Cr	0	- 0.3	7440-47-3
Zr	0	- 0.3	7440-67-7
Ti	0	- 0.2	7440-32-6
V	0	- 0.2	7440-62-2

1 REFERENCES IN FILE CA (1907 TO DATE)
1 REFERENCES IN FILE CAPLUS (1907 TO DATE)

=> d 114 23

L14 ANSWER 23 OF 32 REGISTRY COPYRIGHT 2008 ACS on STN
RN 224648-15-1 REGISTRY
ED Entered STN: 11 Jun 1999
CN Aluminum alloy, base, Al 92-100,Mg 0-5,Mn 0.1-1.6,Si 0-0.5,Zr 0.1-0.4,Sc
0-0.4,Ti 0-0.3,Zn 0.1,Fe 0-0.1 (9CI) (CA INDEX NAME)
MF Al . Fe . Mg . Mn . Sc . Si . Ti . Zn . Zr
CI AYS
SR CA
LC STN Files: CA, CAPLUS

Component	Component Percent	Component Registry Number
Al	92 - 100	7429-90-5
Mg	0 - 5	7439-95-4
Mn	0.1 - 1.6	7439-96-5
Si	0 - 0.5	7440-21-3
Zr	0.1 - 0.4	7440-67-7
Sc	0 - 0.4	7440-20-2
Ti	0 - 0.3	7440-32-6
Zn	0.1	7440-66-6
Fe	0 - 0.1	7439-89-6

1 REFERENCES IN FILE CA (1907 TO DATE)
1 REFERENCES IN FILE CAPLUS (1907 TO DATE)

=> d 114 24

L14 ANSWER 24 OF 32 REGISTRY COPYRIGHT 2008 ACS on STN
RN 210692-17-4 REGISTRY
ED Entered STN: 02 Sep 1998
CN Aluminum alloy, base, Al 80-99,Zn 0.8-9.7,Mg 0.1-3.7,Cu 0-2.6,Fe
0.1-1.4,Mn 0-0.8,Si 0.1-0.7,Zr 0-0.5,Cr 0-0.4,Ti 0-0.2 (9CI) (CA INDEX
NAME)
MF Al . Cr . Cu . Fe . Mg . Mn . Si . Ti . Zn . Zr
CI AYS
SR CA
LC STN Files: CA, CAPLUS

Component	Component Percent	Component Registry Number
Al	80 - 99	7429-90-5
Zn	0.8 - 9.7	7440-66-6
Mg	0.1 - 3.7	7439-95-4
Cu	0 - 2.6	7440-50-8
Fe	0.1 - 1.4	7439-89-6
Mn	0 - 0.8	7439-96-5

Si	0.1	-	0.7	7440-21-3
Zr	0	-	0.5	7440-67-7
Cr	0	-	0.4	7440-47-3
Ti	0	-	0.2	7440-32-6

1 REFERENCES IN FILE CA (1907 TO DATE)
1 REFERENCES IN FILE CAPLUS (1907 TO DATE)

=> d cost

COST IN U.S. DOLLARS	SINCE FILE ENTRY	TOTAL SESSION
CONNECT CHARGES	5.20	11.51
NETWORK CHARGES	0.78	1.74
SEARCH CHARGES	118.45	204.18
DISPLAY CHARGES	62.00	72.72
	-----	-----
FULL ESTIMATED COST	186.43	290.15
DISCOUNT AMOUNTS (FOR QUALIFYING ACCOUNTS)	SINCE FILE ENTRY	TOTAL SESSION
CA SUBSCRIBER PRICE	0.00	-1.60

IN FILE 'REGISTRY' AT 09:57:19 ON 08 SEP 2008

=> d 114 25

L14 ANSWER 25 OF 32 REGISTRY COPYRIGHT 2008 ACS on STN
RN 197586-37-1 REGISTRY
ED Entered STN: 20 Nov 1997
CN Aluminum alloy, base, Al 85-95,Mg 4.5-7,Zn 0.4-5,Mn 0.4-1.2,Fe 0-0.5,Si 0-0.5,Cu 0-0.4,Cr 0-0.3,Zr 0-0.3,Ti 0-0.2 (9CI) (CA INDEX NAME)
MF Al . Cr . Cu . Fe . Mg . Mn . Si . Ti . Zn . Zr
CI AYS
SR CA
LC STN Files: CA, CAPLUS, USPAT2, USPATFULL

Component	Component Percent	Component Registry Number
Al	85 - 95	7429-90-5
Mg	4.5 - 7	7439-95-4
Zn	0.4 - 5	7440-66-6
Mn	0.4 - 1.2	7439-96-5
Fe	0 - 0.5	7439-89-6
Si	0 - 0.5	7440-21-3
Cu	0 - 0.4	7440-50-8
Cr	0 - 0.3	7440-47-3
Zr	0 - 0.3	7440-67-7
Ti	0 - 0.2	7440-32-6

1 REFERENCES IN FILE CA (1907 TO DATE)
1 REFERENCES IN FILE CAPLUS (1907 TO DATE)

=> d 114 26

L14 ANSWER 26 OF 32 REGISTRY COPYRIGHT 2008 ACS on STN
RN 151789-43-4 REGISTRY
ED Entered STN: 17 Dec 1993
CN Aluminum alloy, base, Al 85-98,Mg 1.5-3.5,Ni 0-3.5,Si 0-2.5,Mn 0-1.8,Fe

0-1.5,Cu 0-0.5,Hf 0-0.5,Cr 0-0.4,Ti 0-0.4,V 0-0.4,Zr 0-0.4 (9CI) (CA
INDEX NAME)
MF Al . Cr . Cu . Fe . Hf . Mg . Mn . Ni . Si . Ti . V . Zr
CI AYS
SR CA
LC STN Files: CA, CAPLUS

Component	Component Percent	Component Registry Number
Al	85 - 98	7429-90-5
Mg	1.5 - 3.5	7439-95-4
Ni	0 - 3.5	7440-02-0
Si	0 - 2.5	7440-21-3
Mn	0 - 1.8	7439-96-5
Fe	0 - 1.5	7439-89-6
Cu	0 - 0.5	7440-50-8
Hf	0 - 0.5	7440-58-6
Cr	0 - 0.4	7440-47-3
Ti	0 - 0.4	7440-32-6
V	0 - 0.4	7440-62-2
Zr	0 - 0.4	7440-67-7

1 REFERENCES IN FILE CA (1907 TO DATE)
1 REFERENCES IN FILE CAPLUS (1907 TO DATE)

=> d 114 27

L14 ANSWER 27 OF 32 REGISTRY COPYRIGHT 2008 ACS on STN
RN 147978-44-7 REGISTRY
ED Entered STN: 08 Jun 1993
CN Aluminum alloy, base, Al 88-98,Mg 2-6,Cu 0-2,Mn 0-2,Cr 0-1,Zn 0-0.5,Zr
0-0.3,Fe 0-0.2,Si 0-0.2,Ti 0-0.2 (9CI) (CA INDEX NAME)
MF Al . Cr . Cu . Fe . Mg . Mn . Si . Ti . Zn . Zr
CI AYS
SR CA
LC STN Files: CA, CAPLUS

Component	Component Percent	Component Registry Number
Al	88 - 98	7429-90-5
Mg	2 - 6	7439-95-4
Cu	0 - 2	7440-50-8
Mn	0 - 2	7439-96-5
Cr	0 - 1	7440-47-3
Zn	0 - 0.5	7440-66-6
Zr	0 - 0.3	7440-67-7
Fe	0 - 0.2	7439-89-6
Si	0 - 0.2	7440-21-3
Ti	0 - 0.2	7440-32-6

1 REFERENCES IN FILE CA (1907 TO DATE)
1 REFERENCES IN FILE CAPLUS (1907 TO DATE)

=> d 114 28

L14 ANSWER 28 OF 32 REGISTRY COPYRIGHT 2008 ACS on STN
RN 129703-76-0 REGISTRY
ED Entered STN: 05 Oct 1990

CN Aluminum alloy, base, Al 76-98,Zn 1-12,Mg 0.5-4,Cu 0-3,Mn 0-1,Cr 0-0.5,Fe 0-0.5,Hf 0-0.5,Sc 0-0.5,Si 0-0.5,Ti 0-0.5,Zr 0-0.5,O 0-0.1 (9CI) (CA INDEX NAME)
 MF Al . Cr . Cu . Fe . Hf . Mg . Mn . O . Sc . Si . Ti . Zn . Zr
 CI AYS
 SR CA
 LC STN Files: CA, CAPLUS, USPATFULL

Component	Component Percent	Component Registry Number
Al	76 - 98	7429-90-5
Zn	1 - 12	7440-66-6
Mg	0.5 - 4	7439-95-4
Cu	0 - 3	7440-50-8
Mn	0 - 1	7439-96-5
Cr	0 - 0.5	7440-47-3
Fe	0 - 0.5	7439-89-6
Hf	0 - 0.5	7440-58-6
Sc	0 - 0.5	7440-20-2
Si	0 - 0.5	7440-21-3
Ti	0 - 0.5	7440-32-6
Zr	0 - 0.5	7440-67-7
O	0 - 0.1	17778-80-2

1 REFERENCES IN FILE CA (1907 TO DATE)
 1 REFERENCES IN FILE CAPLUS (1907 TO DATE)

=> d 114 29

L14 ANSWER 29 OF 32 REGISTRY COPYRIGHT 2008 ACS on STN
 RN 129703-72-6 REGISTRY
 ED Entered STN: 05 Oct 1990
 CN Aluminum alloy, base, Al 76-98,Zn 1-12,Mg 0.5-4,Cu 0-3,Mn 0-1,Cr 0-0.5,Fe 0-0.5,Hf 0-0.5,Sc 0-0.5,Si 0-0.5,Ti 0-0.5,Zr 0-0.5 (9CI) (CA INDEX NAME)
 MF Al . Cr . Cu . Fe . Hf . Mg . Mn . Sc . Si . Ti . Zn . Zr
 CI AYS
 SR CA
 LC STN Files: CA, CAPLUS, USPATFULL

Component	Component Percent	Component Registry Number
Al	76 - 98	7429-90-5
Zn	1 - 12	7440-66-6
Mg	0.5 - 4	7439-95-4
Cu	0 - 3	7440-50-8
Mn	0 - 1	7439-96-5
Cr	0 - 0.5	7440-47-3
Fe	0 - 0.5	7439-89-6
Hf	0 - 0.5	7440-58-6
Sc	0 - 0.5	7440-20-2
Si	0 - 0.5	7440-21-3
Ti	0 - 0.5	7440-32-6
Zr	0 - 0.5	7440-67-7

1 REFERENCES IN FILE CA (1907 TO DATE)
 1 REFERENCES IN FILE CAPLUS (1907 TO DATE)

=> d 114 30

L14 ANSWER 30 OF 32 REGISTRY COPYRIGHT 2008 ACS on STN
 RN 109982-56-1 REGISTRY
 ED Entered STN: 22 Aug 1987
 CN Aluminum alloy, base, Al 79-100,Cu 0-5,Mg 0-5,Zn 0-5,Mn 0-1.5,Si 0-1.2,Fe
 0-1,Cr 0-0.5,Ti 0-0.5,V 0-0.5,Zr 0-0.5 (9CI) (CA INDEX NAME)
 MF Al . Cr . Cu . Fe . Mg . Mn . Si . Ti . V . Zn . Zr
 CI AYS
 SR CA
 LC STN Files: CA, CAPLUS, USPATFULL

Component	Component Percent	Component Registry Number
Al	79 - 100	7429-90-5
Cu	0 - 5	7440-50-8
Mg	0 - 5	7439-95-4
Zn	0 - 5	7440-66-6
Mn	0 - 1.5	7439-96-5
Si	0 - 1.2	7440-21-3
Fe	0 - 1	7439-89-6
Cr	0 - 0.5	7440-47-3
Ti	0 - 0.5	7440-32-6
V	0 - 0.5	7440-62-2
Zr	0 - 0.5	7440-67-7

1 REFERENCES IN FILE CA (1907 TO DATE)
 1 REFERENCES IN FILE CAPLUS (1907 TO DATE)

=> d 114 31

L14 ANSWER 31 OF 32 REGISTRY COPYRIGHT 2008 ACS on STN
 RN 100100-19-4 REGISTRY
 ED Entered STN: 08 Feb 1986
 CN Aluminum alloy, base, Al 25-97,Mg 0.5-8,Li 2.7-5,Be 0-5,Co 0-5,Cr 0-5,Cu
 0-5,Fe 0-5,Hf 0-5,Mn 0-5,Ni 0-5,Sc 0-5,Si 0-5,Ti 0-5,V 0-5,Zr 0.2-2 (9CI)
 (CA INDEX NAME)
 MF Al . Be . Co . Cr . Cu . Fe . Hf . Li . Mg . Mn . Ni . Sc . Si . Ti . V .
 Zr
 CI AYS
 SR CA
 LC STN Files: CA, CAPLUS, USPATFULL

Component	Component Percent	Component Registry Number
Al	25 - 97	7429-90-5
Mg	0.5 - 8	7439-95-4
Li	2.7 - 5	7439-93-2
Be	0 - 5	7440-41-7
Co	0 - 5	7440-48-4
Cr	0 - 5	7440-47-3
Cu	0 - 5	7440-50-8
Fe	0 - 5	7439-89-6
Hf	0 - 5	7440-58-6
Mn	0 - 5	7439-96-5
Ni	0 - 5	7440-02-0
Sc	0 - 5	7440-20-2
Si	0 - 5	7440-21-3
Ti	0 - 5	7440-32-6
V	0 - 5	7440-62-2

Zr 0.2 - 2 7440-67-7

1 REFERENCES IN FILE CA (1907 TO DATE)
1 REFERENCES IN FILE CAPLUS (1907 TO DATE)

=> d 114 32

L14 ANSWER 32 OF 32 REGISTRY COPYRIGHT 2008 ACS on STN
RN 61992-87-8 REGISTRY
ED Entered STN: 16 Nov 1984
CN Aluminum alloy, base, Al 78-89, Zn 7-11, Mg 2.5-4, Cu 1-2.6, Mn 0.2-1.2, Zr
0.1-0.3, B 0-0.3, Cd 0-0.3, Co 0-0.3, Cr 0-0.3, Fe 0-0.3, Si 0-0.3, Ti 0-0.3, V
0-0.3, W 0-0.3, Be 0-0.2, Nb 0-0.2 (9CI) (CA INDEX NAME)
MF Al . B . Be . Cd . Co . Cr . Cu . Fe . Mg . Mn . Nb . Si . Ti . V . W . Zn
. Zr
CI AYS
LC STN Files: CA, CAPLUS

Component	Component Percent	Component Registry Number
Al	78 - 89	7429-90-5
Zn	7 - 11	7440-66-6
Mg	2.5 - 4	7439-95-4
Cu	1 - 2.6	7440-50-8
Mn	0.2 - 1.2	7439-96-5
Zr	0.1 - 0.3	7440-67-7
B	0 - 0.3	7440-42-8
Cd	0 - 0.3	7440-43-9
Co	0 - 0.3	7440-48-4
Cr	0 - 0.3	7440-47-3
Fe	0 - 0.3	7439-89-6
Si	0 - 0.3	7440-21-3
Ti	0 - 0.3	7440-32-6
V	0 - 0.3	7440-62-2
W	0 - 0.3	7440-33-7
Be	0 - 0.2	7440-41-7
Nb	0 - 0.2	7440-03-1

1 REFERENCES IN FILE CA (1907 TO DATE)
1 REFERENCES IN FILE CAPLUS (1907 TO DATE)

=> FIL REGISTRY

COST IN U.S. DOLLARS	SINCE FILE	TOTAL
	ENTRY	SESSION
FULL ESTIMATED COST	203.81	307.53
DISCOUNT AMOUNTS (FOR QUALIFYING ACCOUNTS)	SINCE FILE	TOTAL
	ENTRY	SESSION
CA SUBSCRIBER PRICE	0.00	-1.60

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DICTIONARY FILE UPDATES: 7 SEP 2008 HIGHEST RN 1047406-12-1

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experimental property data in the original document. For information
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<http://www.cas.org/support/stngen/stndoc/properties.html>

=> SET TERMSET E#

SET COMMAND COMPLETED

=> DEL SEL Y

=> SEL L14 31 RN

E1 THROUGH E1 ASSIGNED

=> S E1/RN

L15 1 100100-19-4/RN

=> SET TERMSET LOGIN

SET COMMAND COMPLETED

=> FIL CAPLUS

COST IN U.S. DOLLARS	SINCE FILE	TOTAL
	ENTRY	SESSION
FULL ESTIMATED COST	0.55	308.08
DISCOUNT AMOUNTS (FOR QUALIFYING ACCOUNTS)	SINCE FILE	TOTAL
	ENTRY	SESSION
CA SUBSCRIBER PRICE	0.00	-1.60

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FILE COVERS 1907 - 8 Sep 2008 VOL 149 ISS 11
FILE LAST UPDATED: 7 Sep 2008 (20080907/ED)

Caplus now includes complete International Patent Classification (IPC)
reclassification data for the second quarter of 2008.

Effective October 17, 2005, revised CAS Information Use Policies apply.
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=> S L15

L16 1 L15

=> DIS L16 1 IBIB

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DO YOU WANT TO CONTINUE WITH THIS REQUEST? (Y)/N:Y

L16 ANSWER 1 OF 1 CAPLUS COPYRIGHT 2008 ACS on STN

ACCESSION NUMBER: 1986:54985 CAPLUS
DOCUMENT NUMBER: 104:54985
ORIGINAL REFERENCE NO.: 104:8801a,8804a
TITLE: Low-density aluminum alloys
INVENTOR(S): Skinner, David John; Okazaki, Kenji; Adam, Colin
McClean
PATENT ASSIGNEE(S): Allied Corp., USA
SOURCE: Eur. Pat. Appl., 28 pp.
CODEN: EPXXDW
DOCUMENT TYPE: Patent
LANGUAGE: English
FAMILY ACC. NUM. COUNT: 1
PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
EP 158769	A1	19851023	EP 1985-100476	19850118
EP 158769	B1	19880504		
R: CH, DE, FR, GB, LI				
US 4661172	A	19870428	US 1984-584856	19840229
CA 1228491	A1	19871027	CA 1985-474001	19850211
JP 60208445	A	19851021	JP 1985-40244	19850228
JP 02036661	B	19900820		
JP 01272742	A	19891031	JP 1988-67998	19880322
PRIORITY APPLN. INFO.:			US 1984-584856	A 19840229

=> file caplus

COST IN U.S. DOLLARS	SINCE FILE ENTRY	TOTAL SESSION
FULL ESTIMATED COST	2.17	310.25
DISCOUNT AMOUNTS (FOR QUALIFYING ACCOUNTS)	SINCE FILE ENTRY	TOTAL SESSION
CA SUBSCRIBER PRICE	0.00	-1.60

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FILE COVERS 1907 - 8 Sep 2008 VOL 149 ISS 11
FILE LAST UPDATED: 7 Sep 2008 (20080907/ED)

Caplus now includes complete International Patent Classification (IPC) reclassification data for the second quarter of 2008.

Effective October 17, 2005, revised CAS Information Use Policies apply. They are available for your review at:

<http://www.cas.org/legal/infopolicy.html>

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=> s 1015163-38-8/rn
      1 1015163-38-8
      0 1015163-38-8D
L17    1 1015163-38-8/RN
          (1015163-38-8 (NOTL) 1015163-38-8D )

=> s 1015163-38-8/rn and 528578-87-2/rn and 371165-10-5/rn and 224648-15-1/rn and
109982-56-1/rn
      1 1015163-38-8
      0 1015163-38-8D
      1 1015163-38-8/RN
          (1015163-38-8 (NOTL) 1015163-38-8D )
      1 528578-87-2
      0 528578-87-2D
      1 528578-87-2/RN
          (528578-87-2 (NOTL) 528578-87-2D )
      1 371165-10-5
      0 371165-10-5D
      1 371165-10-5/RN
          (371165-10-5 (NOTL) 371165-10-5D )
      1 224648-15-1
      0 224648-15-1D
      1 224648-15-1/RN
          (224648-15-1 (NOTL) 224648-15-1D )
      1 109982-56-1
      0 109982-56-1D
      1 109982-56-1/RN
          (109982-56-1 (NOTL) 109982-56-1D )
L18    0 1015163-38-8/RN AND 528578-87-2/RN AND 371165-10-5/RN AND 224648
      -15-1/RN AND 109982-56-1/RN

=> s 1015163-38-8/rn OR 528578-87-2/rn OR 371165-10-5/rn OR 224648-15-1/rn OR
109982-56-1/rn
      1 1015163-38-8
      0 1015163-38-8D
      1 1015163-38-8/RN
          (1015163-38-8 (NOTL) 1015163-38-8D )
      1 528578-87-2
      0 528578-87-2D
      1 528578-87-2/RN
          (528578-87-2 (NOTL) 528578-87-2D )
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1 371165-10-5
 0 371165-10-5D
 1 371165-10-5/RN
 (371165-10-5 (NOTL) 371165-10-5D)
 1 224648-15-1
 0 224648-15-1D
 1 224648-15-1/RN
 (224648-15-1 (NOTL) 224648-15-1D)
 1 109982-56-1
 0 109982-56-1D
 1 109982-56-1/RN
 (109982-56-1 (NOTL) 109982-56-1D)
 L19 5 1015163-38-8/RN OR 528578-87-2/RN OR 371165-10-5/RN OR 224648-15-1/RN OR 109982-56-1/RN

=> d 119

L19 ANSWER 1 OF 5 CAPLUS COPYRIGHT 2008 ACS on STN
 AN 2008:417065 CAPLUS
 DN 148:407592
 TI Method for evaluation of stress corrosion cracking (SCC) of aluminum alloys and aluminum alloys with excellent resistance to SCC
 IN Sakashita, Shinji; Tanaka, Toshiyuki
 PA Kobe Steel, Ltd., Japan
 SO Jpn. Kokai Tokkyo Koho, 17pp.
 CODEN: JKXXAF
 DT Patent
 LA Japanese
 FAN.CNT 1

	PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
PI	JP 2008076297	A	20080403	JP 2006-257531	20060922
PRAI	JP 2006-257531		20060922		

=> d 119 2

L19 ANSWER 2 OF 5 CAPLUS COPYRIGHT 2008 ACS on STN
 AN 2003:391115 CAPLUS
 DN 138:389217
 TI Manufacture of aluminum alloy billets by semisolid forging for transportation equipments
 IN Mikubo, Shigeru; Mizouchi, Masafumi; Murayama, Yasuyuki; Iwashita, Tsunaki
 PA Kyushu Mitsui Aluminium Co., Ltd., Japan
 SO Jpn. Kokai Tokkyo Koho, 5 pp.
 CODEN: JKXXAF
 DT Patent
 LA Japanese
 FAN.CNT 1

	PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
PI	JP 2003147497	A	20030521	JP 2001-337404	20011102
	JP 3802796	B2	20060726		
PRAI	JP 2001-337404		20011102		

=> d 119 3

L19 ANSWER 3 OF 5 CAPLUS COPYRIGHT 2008 ACS on STN
 AN 2001:808251 CAPLUS
 DN 135:347609
 TI Manufacture of nanosize aluminum alloy powders by attrition milling with a

surfactant
IN Upadhy, Kamleshwar; Hoffman, Wesley P.
PA United States Dept. of the Air Force, USA
SO U.S., 6 pp.
CODEN: USXXAM

DT Patent
LA English

FAN.CNT 1

	PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
PI	US 6312643	B1	20011106	US 1997-957013	19971024
PRAI	US 1997-957013		19971024		

RE.CNT 8 THERE ARE 8 CITED REFERENCES AVAILABLE FOR THIS RECORD
ALL CITATIONS AVAILABLE IN THE RE FORMAT

=> d 119 4

L19 ANSWER 4 OF 5 CAPLUS COPYRIGHT 2008 ACS on STN

AN 1999:344803 CAPLUS

DN 130:355579

TI Pressure-cast aluminum alloy structural parts

IN Winkler, Reinhard; Wust, Jurgen

PA Alusuisse Technology & Management AG, Switz.; Alcan Technology & Management AG

SO Eur. Pat. Appl., 6 pp.

CODEN: EPXXDW

DT Patent

LA German

FAN.CNT 1

	PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
PI	EP 918095	A1	19990526	EP 1997-810884	19971120
	EP 918095	B1	20030326		
	R: AT, BE, CH, DE, DK, ES, FR, GB, GR, IT, LI, LU, NL, SE, MC, PT, IE, SI, LT, LV, FI, RO				
	PT 918095	T	20030630	PT 1997-810884	19971120
	ES 2192257	T3	20031001	ES 1997-810884	19971120
	HU 9802626	A1	19990928	HU 1998-2626	19981112
	HU 220128	B	20011128		
	PL 186936	B1	20040430	PL 1998-329760	19981118
	BR 9804709	A	19991109	BR 1998-4709	19981119
PRAI	EP 1997-810884	A	19971120		

RE.CNT 4 THERE ARE 4 CITED REFERENCES AVAILABLE FOR THIS RECORD
ALL CITATIONS AVAILABLE IN THE RE FORMAT

=> d 119 5

L19 ANSWER 5 OF 5 CAPLUS COPYRIGHT 2008 ACS on STN

AN 1987:501308 CAPLUS

DN 107:101308

OREF 107:16460h,16461a

TI Metallic gasket

IN Sakai, Yakichi

PA Nippon Gakki Co., Ltd., Japan; Hamamatsu Gasket Seisakusho Ltd.

SO Ger. Offen., 7 pp.

CODEN: GWXXBX

DT Patent

LA German

FAN.CNT 1

	PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
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PI	DE 3633988	A1	19870416	DE 1986-3633988	19861006
	DE 3633988	C2	19900613		
	US 4810591	A	19890307	US 1986-916293	19861007
PRAI	JP 1985-224777	A	19851011		

=> logoff

ALL L# QUERIES AND ANSWER SETS ARE DELETED AT LOGOFF

LOGOFF? (Y)/N/HOLD:y

COST IN U.S. DOLLARS

SINCE FILE

TOTAL

ENTRY

SESSION

FULL ESTIMATED COST

66.13

376.38

DISCOUNT AMOUNTS (FOR QUALIFYING ACCOUNTS)

SINCE FILE

TOTAL

ENTRY

SESSION

CA SUBSCRIBER PRICE

0.00

-1.60

STN INTERNATIONAL LOGOFF AT 10:16:55 ON 08 SEP 2008

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LOGINID:ssptamls1742

PASSWORD:

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NEWS	4	NOV 26	CHEMSAFE now available on STN Easy
NEWS	5	NOV 26	Two new SET commands increase convenience of STN searching
NEWS	6	DEC 01	ChemPort single article sales feature unavailable
NEWS	7	DEC 12	GBFULL now offers single source for full-text coverage of complete UK patent families
NEWS	8	DEC 17	Fifty-one pharmaceutical ingredients added to PS
NEWS	9	JAN 06	The retention policy for unread STNmail messages will change in 2009 for STN-Columbus and STN-Tokyo
NEWS	10	JAN 07	WPIDS, WPINDEX, and WPIX enhanced Japanese Patent Classification Data
NEWS	11	FEB 02	Simultaneous left and right truncation (SLART) added for CERAB, COMPUAB, ELCOM, and SOLIDSTATE
NEWS	12	FEB 02	GENBANK enhanced with SET PLURALS and SET SPELLING
NEWS	13	FEB 06	Patent sequence location (PSL) data added to USGENE
NEWS	14	FEB 10	COMPENDEX reloaded and enhanced
NEWS	15	FEB 11	WTEXTILES reloaded and enhanced
NEWS	16	FEB 19	New patent-examiner citations in 300,000 CA/Caplus patent records provide insights into related prior art

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NEWS 20 FEB 23 TOXCENTER updates mirror those of MEDLINE - more precise author group fields and 2009 MeSH terms

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NEWS 25 MAR 11 ESBIODATABASE reloaded and enhanced

NEWS 26 MAR 20 CAS databases on STN enhanced with new super role for nanomaterial substances

NEWS 27 MAR 23 CA/CAPLUS enhanced with more than 250,000 patent equivalents from China

NEWS 28 MAR 30 IMSPATENTS reloaded and enhanced

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FULL ESTIMATED COST	0.22	0.22

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=> s 3.5-4.5 Mg/mac and 0.8-1.5 Mn/mac and 70-100 Al/mac

184935 3.5-4.5/MAC

82170 MG/MAC

5622 3.5-4.5 MG/MAC

(3.5-4.5/MAC (P) MG/MAC)

430489 0.8-1.5/MAC

395965 MN/MAC

138266 0.8-1.5 MN/MAC

(0.8-1.5/MAC (P) MN/MAC)

643943 70-100/MAC

270098 AL/MAC

82453 70-100 AL/MAC

(70-100/MAC (P) AL/MAC)

L1 687 3.5-4.5 MG/MAC AND 0.8-1.5 MN/MAC AND 70-100 AL/MAC

=> s l1 and (silicon or si) and (iron or fe)

112104 SILICON

477617 SI

16367 SIS

493822 SI

(SI OR SIS)

642234 IRON

671421 FE

426 FES

671828 FE

(FE OR FES)

L2 376 L1 AND (SILICON OR SI) AND (IRON OR FE)

=> s 3.5-4.5 Mg/mac and 0.8-1.5 Mn/mac and 70-100 Al/mac and 0-0.5 Fe/mac and 0-0.5 Si/mac

184935 3.5-4.5/MAC

82170 MG/MAC

5622 3.5-4.5 MG/MAC

(3.5-4.5/MAC (P) MG/MAC)

430489 0.8-1.5/MAC

395965 MN/MAC

138266 0.8-1.5 MN/MAC

(0.8-1.5/MAC (P) MN/MAC)

643943 70-100/MAC

270098 AL/MAC

82453 70-100 AL/MAC

(70-100/MAC (P) AL/MAC)

543376 0-0.5/MAC

562021 FE/MAC

47574 0-0.5 FE/MAC

(0-0.5/MAC (P) FE/MAC)

543376 0-0.5/MAC

423505 SI/MAC

222028 0-0.5 SI/MAC

(0-0.5/MAC (P) SI/MAC)

L3 268 3.5-4.5 MG/MAC AND 0.8-1.5 MN/MAC AND 70-100 AL/MAC AND 0-0.5
 FE/MAC AND 0-0.5 SI/MAC

=> s l3 and (titanium or ti) and (zirconium or zr)

204066 TITANIUM
255410 TI
55143 TIS
310553 TI
 (TI OR TIS)
114282 ZIRCONIUM
123063 ZR
2 ZRS
123065 ZR
 (ZR OR ZRS)

L4 62 L3 AND (TITANIUM OR TI) AND (ZIRCONIUM OR ZR)

=> file (hcaplus, inspec, aluminium, compendex, confsci, corrosion, ema, epfull,
gbfull, mdf, metadex, scisearch, uspatall)

COST IN U.S. DOLLARS	SINCE FILE ENTRY	TOTAL SESSION
FULL ESTIMATED COST	89.44	89.66

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FILE COVERS 1907 - 30 Mar 2009 VOL 150 ISS 14
FILE LAST UPDATED: 29 Mar 2009 (20090329/ED)

HCAplus now includes complete International Patent Classification (IPC) reclassification data for the third quarter of 2008.

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This file contains CAS Registry Numbers for easy and accurate substance identification.

=> s l4

L5 110 L4

=> dup rem l5

PROCESSING COMPLETED FOR L5

L6 110 DUP REM L5 (0 DUPLICATES REMOVED)

=> s l6 and (aluminum or aluminium)

L7 110 S L6
1088426 ALUMINUM
312 ALUMINUMS

1088490 ALUMINUM
(ALUMINUM OR ALUMINUMS)
17807 ALUMINIUM
36 ALUMINIUMS
17836 ALUMINIUM
(ALUMINIUM OR ALUMINIUMS)
L8 109 L7 AND (ALUMINUM OR ALUMINIUM)

=> d scan 18

L8 109 ANSWERS HCAPLUS COPYRIGHT 2009 ACS on STN
CC 56-3 (Nonferrous Metals and Alloys)
TI Aluminum-based alloy for aviation and shipbuilding
ST aluminum magnesium manganese alloy armor aviation
IT Aerospace industry
(aviation and aeronautics; aluminum-based alloy for aviation
and shipbuilding)
IT Armor
(plate; aluminum-based alloy for aviation and shipbuilding)
IT 902164-10-7 902164-12-9 902164-15-2 902164-18-5
RL: PRP (Properties); TEM (Technical or engineered material use); USES
(Uses)
(aluminum armor alloy; aluminum-based alloy for
aviation and shipbuilding)
IT 902164-07-2
RL: TEM (Technical or engineered material use); USES (Uses)
(aluminum armor alloy; aluminum-based alloy for
aviation and shipbuilding)

HOW MANY MORE ANSWERS DO YOU WISH TO SCAN? (1):1

L8 109 ANSWERS HCAPLUS COPYRIGHT 2009 ACS on STN
CC 56-9 (Nonferrous Metals and Alloys)
TI CO2 laser welding of aluminium shipbuilding industry alloys. AA
5083, AA 5383, AA 5059, and AA 6082
ST aluminum alloy laser welding
IT Microstructure
(CO2 laser welding of Al shipbuilding industry alloys)
IT Welding of metals
(laser; CO2 laser welding of Al shipbuilding industry alloys)
IT 12616-86-3, AA 5083 12732-13-7, AA 6082 269058-32-4, AA 5059
327622-69-5, AA 5383
RL: CPS (Chemical process); PEP (Physical, engineering or chemical
process); TEM (Technical or engineered material use); PROC (Process); USES
(Uses)
(CO2 laser welding of Al shipbuilding industry alloys)
IT 124-38-9, Carbon dioxide, uses
RL: NUU (Other use, unclassified); USES (Uses)
(CO2 laser welding of Al shipbuilding industry alloys)

HOW MANY MORE ANSWERS DO YOU WISH TO SCAN? (1):1

L8 109 ANSWERS HCAPLUS COPYRIGHT 2009 ACS on STN
CC 56-12 (Nonferrous Metals and Alloys)
TI Features of superplastic deformation of some aluminum alloys
ST superplastic deformation aluminum alloy
IT Plastic deformation
(superplastic; features of superplastic deformation of aluminum
alloys)
IT 12672-17-2, D20 39331-96-9, D19 39461-63-7, AA7475 81159-87-7
, AMg4 110414-16-9, Neopral
RL: PEP (Physical, engineering or chemical process); PRP (Properties);

PROC (Process)
(features of superplastic deformation of aluminum alloys)

HOW MANY MORE ANSWERS DO YOU WISH TO SCAN? (1):1

L8 109 ANSWERS HCAPLUS COPYRIGHT 2009 ACS on STN
CC 56-9 (Nonferrous Metals and Alloys)
TI Properties of AlZn4.5Mg1 joints welded with different filler materials
ST welding aluminum alloy filler metal property
IT Welding
Welds
(properties of AlZn4.5Mg1 joints welded with different filler materials)
IT 12616-87-4 37268-39-6, SG-AlMg5 75686-78-1
RL: NUU (Other use, unclassified); PEP (Physical, engineering or chemical process); PROC (Process); USES (Uses)
(filler metal; properties of AlZn4.5Mg1 joints welded with different filler materials)
IT 12675-83-1, AlZn4.5Mg1
RL: PEP (Physical, engineering or chemical process); PRP (Properties); PROC (Process)
(properties of AlZn4.5Mg1 joints welded with different filler materials)

HOW MANY MORE ANSWERS DO YOU WISH TO SCAN? (1):1

L8 109 ANSWERS HCAPLUS COPYRIGHT 2009 ACS on STN
IC ICM C22F001-04
ICS C22F001-053; C22C021-10
CC 56-11 (Nonferrous Metals and Alloys)
TI Processing of nonrecrystallized aluminum alloy sheets and plates.
ST aluminum alloy plate sheet processing; heat treatment aluminum alloy
IT Aluminum alloy, base
RL: USES (Uses)
(processing of nonrecrystd., for toughness)
IT 129703-71-5, Aluminum 87, copper 1.5, magnesium 1.8, zinc 10, zirconium 0.1 129703-72-6 129703-73-7 129703-74-8 129703-75-9 129703-76-0
RL: USES (Uses)
(processing of nonrecrystd., for toughness)

HOW MANY MORE ANSWERS DO YOU WISH TO SCAN? (1):1

L8 109 ANSWERS HCAPLUS COPYRIGHT 2009 ACS on STN
CC 56-11 (Nonferrous Metals and Alloys)
TI Al-Mg alloy suitable for armor plate applications
ST aluminum magnesium alloy rolling annealing strength corrosion armor plate
IT Annealing
Casting of metals
Cold rolling
Homogenization
Metalworking
(Al-Mg alloy product suitable for armor plate applications)
IT Rolling (metals)
(hot; Al-Mg alloy product suitable for armor plate applications)
IT Armor
(plate, military; Al-Mg alloy product suitable for armor plate applications)
IT Heating

(preheating; Al-Mg alloy product suitable for armor plate applications)
IT 951323-60-7 1047982-87-5
RL: PEP (Physical, engineering or chemical process); TEM (Technical or
engineered material use); PROC (Process); USES (Uses)
(Al-Mg alloy product suitable for armor plate applications)

HOW MANY MORE ANSWERS DO YOU WISH TO SCAN? (1):0

=> e toyoda yusuke/au,in

E1	1	TOYODA YUMIKO/AU
E2	5	TOYODA YURIKO/AU
E3	14 -->	TOYODA YUSUKE/AU
E4	5	TOYODA YUSUKE/IN
E5	3	TOYODA YUTA/AU
E6	77	TOYODA YUTAKA/AU
E7	5	TOYODA YUTAKA/IN
E8	1	TOYODA YUUJI/AU
E9	3	TOYODA YUUJIROU/AU
E10	1	TOYODA YUUJIROU/AU
E11	1	TOYODA YUUJIROU/IN
E12	1	TOYODA YUUSUKE/AU

=> s e3-e4

	14	"TOYODA YUSUKE"/AU
	5	"TOYODA YUSUKE"/IN
L9	14	("TOYODA YUSUKE"/AU OR "TOYODA YUSUKE"/IN)

=> e mizukami takahiro/au,in

E1	8	MIZUKAMI TAKAAKI/AU
E2	6	MIZUKAMI TAKAAKI/IN
E3	2 -->	MIZUKAMI TAKAHIRO/AU
E4	1	MIZUKAMI TAKAHIRO/IN
E5	8	MIZUKAMI TAKAO/AU
E6	6	MIZUKAMI TAKAO/IN
E7	2	MIZUKAMI TAKASHI/AU
E8	5	MIZUKAMI TAKAYOSHI/AU
E9	5	MIZUKAMI TAKAYOSHI/IN
E10	4	MIZUKAMI TAKAYUKI/AU
E11	17	MIZUKAMI TAKESHI/AU
E12	5	MIZUKAMI TAKEYUKI/AU

=> s e3-e4

	2	"MIZUKAMI TAKAHIRO"/AU
	1	"MIZUKAMI TAKAHIRO"/IN
L10	2	("MIZUKAMI TAKAHIRO"/AU OR "MIZUKAMI TAKAHIRO"/IN)

=> e fukuchi fumiaka/au,in

E1	36	FUKUCHI ETSUO/AU
E2	36	FUKUCHI ETSUO/IN
E3	0 -->	FUKUCHI FUMIAKA/AU
E4	0	FUKUCHI FUMIAKA/IN
E5	14	FUKUCHI FUMIAKI/AU
E6	12	FUKUCHI FUMIAKI/IN
E7	6	FUKUCHI FUMIHIKO/AU
E8	1	FUKUCHI FUMIO/AU
E9	1	FUKUCHI FUMIO/IN
E10	1	FUKUCHI FUSAICHI/AU
E11	1	FUKUCHI FUSATOSHI/AU
E12	1	FUKUCHI FUSAYASU/AU

=> s e5-e6

	14	"FUKUCHI FUMIAKI"/AU
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12 "FUKUCHI FUMIAKI"/IN
L11      14 ("FUKUCHI FUMIAKI"/AU OR "FUKUCHI FUMIAKI"/IN)

=> e hata tsunehisa/au,in
E1      262      HATA TSUJI/IAKI/AU
E2      23       HATA TSUJI/IAKI/IN
E3      5  -->   HATA TSUNEHISA/AU
E4      5       HATA TSUNEHISA/IN
E5      1       HATA TSUNEO/AU
E6      1       HATA TSUNEO/IN
E7      1       HATA TSURU/AU
E8      1       HATA TSURU/IN
E9      1       HATA TSUTOMU/AU
E10     1       HATA TSUTOMU/IN
E11     16      HATA TSUYOSHI/AU
E12     9       HATA TSUYOSHI/IN

=> s e3-e4
5 "HATA TSUNEHISA"/AU
5 "HATA TSUNEHISA"/IN
L12      5 ("HATA TSUNEHISA"/AU OR "HATA TSUNEHISA"/IN)

=> e shibata katsuhiro/au,in
E1      27       SHIBATA KATSUHIKO/AU
E2      25       SHIBATA KATSUHIKO/IN
E3      133  -->  SHIBATA KATSUHIRO/AU
E4      78       SHIBATA KATSUHIRO/IN
E5      4        SHIBATA KATSUHISA/AU
E6      1        SHIBATA KATSUI/AU
E7      1        SHIBATA KATSUI/IN
E8      146      SHIBATA KATSUJI/AU
E9      126      SHIBATA KATSUJI/IN
E10     14       SHIBATA KATSUKI/AU
E11     12       SHIBATA KATSUKI/IN
E12     4        SHIBATA KATSUMASA/AU

=> s e3-e4
133 "SHIBATA KATSUHIRO"/AU
78  "SHIBATA KATSUHIRO"/IN
L13  133 ("SHIBATA KATSUHIRO"/AU OR "SHIBATA KATSUHIRO"/IN)

=> s (l9 or l10 or l11 or l12 or l13)
L14  159 (L9 OR L10 OR L11 OR L12 OR L13)

=> dup rem l14
PROCESSING COMPLETED FOR L14
L15  159 DUP REM L14 (0 DUPLICATES REMOVED)

=> d scan l15

L15  159 ANSWERS      HCAPLUS      COPYRIGHT 2009 ACS on STN
IC   ICM  G01N021-64
ICS  G01N021-05; G01N021-15; G01N021-53; G01N021-59
TI   Seaweed density measurement system. [Machine Translation].

HOW MANY MORE ANSWERS DO YOU WISH TO SCAN? (1):1

L15  159 ANSWERS      HCAPLUS      COPYRIGHT 2009 ACS on STN
CC   11F (Biological Chemistry: Physiology)
TI   Thyroid uptake of iodine-131. IV. Effect of the removal of some endocrine
      glands and of corticoid replacement
IT   Thymus gland

```


(in iodine-131 metabolism by thyroid)
IT Hormones
(sex, I metabolism by thyroid and)
IT Corticosteroids
(thyroid metabolic response to)

HOW MANY MORE ANSWERS DO YOU WISH TO SCAN? (1):1

L15 159 ANSWERS HCAPLUS COPYRIGHT 2009 ACS on STN
IC ICM H01L027-148
ICS H01L021-339; H01L029-762; H04N005-335
TI Electric charge transfer device and solid-state image sensing device
[Machine Translation].

HOW MANY MORE ANSWERS DO YOU WISH TO SCAN? (1):1

L15 159 ANSWERS HCAPLUS COPYRIGHT 2009 ACS on STN
CC 72 (Hormones and Related Substances)
TI Emergence of infection in rats after administration of corticosteroids. I.
Symptoms, autopsy findings, and bacteriological observations
IT Corynebacterium pseudotuberculosis
Corynebacterium pseudotuberculosis
(antibiotic sensitivity of)
IT Infections
Infections
(corticosteroid-lowering of resistance to)
IT Corticosteroids
(infection resistance lowering by)
IT Antibiotic substances
(Corynebacterium pseudotuberculosis sensitivity to)
IT 50-23-7, Cortisol
(infection activation by)
IT 50-02-2, Pregna-1,4-diene-3,20-dione,
9-fluoro-11 β ,17,21-trihydroxy-16 α -methyl- 50-22-6,
Corticosterone 50-24-8, Pregna-1,4-diene-3,20-dione,
11 β ,17,21-trihydroxy- 53-06-5, Cortisone 124-94-7,
Pregna-1,4-diene-3,20-dione, 9-fluoro-11 β ,16 α ,17,21-
tetrahydroxy- 127-31-1, Pregn-4-ene-3,20-dione,
9-fluoro-11 β ,17,21-trihydroxy- 302-25-0,
Pregna-1,4-diene-3,20-dione, 11 β ,17,21-trihydroxy-(prednisolone),
phosphate
(infection resistance lowering by)

HOW MANY MORE ANSWERS DO YOU WISH TO SCAN? (1):0

=> s l15 and (aluminium or aluminum or al) and (alloy or die casting or die-casting
or casting)

L16 159 S L15
17807 ALUMINIUM
36 ALUMINIUMS
17836 ALUMINIUM
(ALUMINIUM OR ALUMINIUMS)
1088426 ALUMINUM
312 ALUMINUMS
1088490 ALUMINUM
(ALUMINUM OR ALUMINUMS)
1060837 AL
6931 ALS
1067419 AL
(AL OR ALS)
747215 ALLOY
560056 ALLOYS

933573 ALLOY
 (ALLOY OR ALLOYS)
 91168 DIE
 22194 DIES
 1630 DICE
 150 DICES
 101211 DIE
 (DIE OR DIES OR DICE OR DICES)
 164042 CASTING
 34741 CASTINGS
 177263 CASTING
 (CASTING OR CASTINGS)
 7552 DIE CASTING
 (DIE(W)CASTING)
 91168 DIE
 22194 DIES
 1630 DICE
 150 DICES
 101211 DIE
 (DIE OR DIES OR DICE OR DICES)
 164042 CASTING
 34741 CASTINGS
 177263 CASTING
 (CASTING OR CASTINGS)
 7552 DIE-CASTING
 (DIE(W)CASTING)
 164042 CASTING
 34741 CASTINGS
 177263 CASTING
 (CASTING OR CASTINGS)
 L17 31 L16 AND (ALUMINIUM OR ALUMINUM OR AL) AND (ALLOY OR DIE CASTING
 OR DIE-CASTING OR CASTING)

=> d scan 117

L17 31 ANSWERS HCAPLUS COPYRIGHT 2009 ACS on STN
 IC ICM C22C021-02
 ICS B22D021-04; B22D029-00; C22F001-043; C22F001-00
 CC 56-2 (Nonferrous Metals and Alloys)
 TI High-toughness aluminum alloy casting and
 its production method
 ST high toughness aluminum alloy casting
 IT Casting of metals
 Tensile strength
 Toughness
 Yield strength
 (high-toughness aluminum alloy casting
 and its production method)
 IT 870462-24-1 870462-25-2 870462-26-3 870462-27-4 870462-28-5
 870462-29-6 870462-30-9
 RL: PEP (Physical, engineering or chemical process); PRP (Properties); PYP
 (Physical process); PROC (Process)
 (high-toughness aluminum alloy casting
 and its production method)

HOW MANY MORE ANSWERS DO YOU WISH TO SCAN? (1):1

L17 31 ANSWERS HCAPLUS COPYRIGHT 2009 ACS on STN
 IC ICM C22C021-02
 ICS B22D017-00; C22C001-02
 CC 56-2 (Nonferrous Metals and Alloys)
 TI Manufacture of Al-Si alloy cast having high toughness

and stress corrosion cracking resistance
 ST aluminum silicon alloy casting toughness
 stress corrosion cracking resistance
 IT Casting of metals
 Impact strength
 (manufacture of Al-Si alloy cast having high toughness
 and stress corrosion cracking resistance)
 IT Stress corrosion cracking
 (resistance; manufacture of Al-Si alloy cast having high
 toughness and stress corrosion cracking resistance)
 IT 804566-20-9P 804566-22-1P 804566-25-4P
 RL: IMF (Industrial manufacture); PEP (Physical, engineering or chemical
 process); PYP (Physical process); PREP (Preparation); PROC (Process)
 (manufacture of Al-Si alloy cast having high toughness
 and stress corrosion cracking resistance)

HOW MANY MORE ANSWERS DO YOU WISH TO SCAN? (1):1

L17 31 ANSWERS HCAPLUS COPYRIGHT 2009 ACS on STN
 IC ICM B22D017-00
 ICS B22C009-06; B22D017-22; C22C021-06
 CC 56-2 (Nonferrous Metals and Alloys)
 TI Die-cast aluminum-magnesium alloy products having ribs
 ST die casting aluminum magnesium alloy
 product rib
 IT Casting of metals
 (die; die-cast Al-Mg alloy products having ribs
 with high strength and toughness)
 IT 607356-70-7 607356-71-8 607356-72-9 607356-73-0 607356-74-1
 607356-75-2 607356-76-3 607356-77-4 607356-78-5 607356-79-6
 RL: PEP (Physical, engineering or chemical process); PYP (Physical
 process); TEM (Technical or engineered material use); PROC (Process); USES
 (Uses)
 (die-cast Al-Mg alloy products having ribs with
 high strength and toughness)

HOW MANY MORE ANSWERS DO YOU WISH TO SCAN? (1):1

L17 31 ANSWERS HCAPLUS COPYRIGHT 2009 ACS on STN
 IC ICM C09K003-14
 ICS C09K003-14; C08J005-14; F16D069-02
 CC 57-9 (Ceramics)
 TI Fiber-reinforced frictional materials having improved wear resistance
 ST fiber reinforced friction material; brake pad friction material
 IT Polyamide fibers, uses
 RL: TEM (Technical or engineered material use); USES (Uses)
 (aramid; wear-resistant friction materials containing)
 IT Cashew (Anacardium occidentale)
 (dust; wear-resistant friction materials containing)
 IT Brakes (mechanical)
 (linings; wear-resistant friction materials for)
 IT Carbon fibers, uses
 Phenolic resins, uses
 Synthetic fibers
 RL: TEM (Technical or engineered material use); USES (Uses)
 (wear-resistant friction materials containing)
 IT Friction materials
 (wear-resistant friction materials reinforced with fibers)
 IT 7429-90-5, Aluminum, uses 7439-89-6, Iron, uses 7440-02-0,
 Nickel, uses 7440-47-3, Chromium, uses 7440-50-8, Copper, uses
 7440-66-6, Zinc, uses 7782-42-5, Graphite, uses 11143-56-9
 12597-68-1, Stainless steel, uses

RL: TEM (Technical or engineered material use); USES (Uses)
(wear-resistant friction materials containing)

HOW MANY MORE ANSWERS DO YOU WISH TO SCAN? (1):1

L17 31 ANSWERS HCAPLUS COPYRIGHT 2009 ACS on STN
CC 56-2 (Nonferrous Metals and Alloys)
TI Aluminum alloys for mushy-state casting of
automotive chassis
ST aluminum silicon alloy casting cooling
automobile chassis
IT Casting of metals
Cooling
(aluminum alloys for mushy-state casting
of automotive chassis)
IT Cast alloys
RL: DEV (Device component use); USES (Uses)
(aluminum; aluminum alloys for
mushy-state casting of automotive chassis)
IT Automobiles
(chassis; aluminum alloys for mushy-state
casting of automotive chassis)
IT 11145-29-2 12609-50-6, Aluminum 97, silicon 3 12686-71-4
12727-35-4
RL: DEV (Device component use); PEP (Physical, engineering or chemical
process); PYP (Physical process); PROC (Process); USES (Uses)
(aluminum alloys for mushy-state casting
of automotive chassis)

HOW MANY MORE ANSWERS DO YOU WISH TO SCAN? (1):0

=> s (l8 or l17)

L18 139 (L8 OR L17)

=> dup rem l18

PROCESSING COMPLETED FOR L18

L19 139 DUP REM L18 (0 DUPLICATES REMOVED)

=> d cost

COST IN U.S. DOLLARS	SINCE FILE ENTRY	TOTAL SESSION
CONNECT CHARGES	36.14	39.57
NETWORK CHARGES	0.91	1.54
SEARCH CHARGES	0.00	85.60
	-----	-----
FULL ESTIMATED COST	37.05	126.71

IN FILE 'HCAPLUS' AT 09:16:52 ON 30 MAR 2009

=> s l19 and (magnesium or mg) and (manganese or mn) and (titanium or zirconium)

L20 139 S L19

552337 MAGNESIUM

91 MAGNESIUMS

552372 MAGNESIUM

(MAGNESIUM OR MAGNESIUMS)

1548880 MG

1729 MGS

1550063 MG

(MG OR MGS)

420771 MANGANESE

116 MANGANESES

```

420783 MANGANESE
      (MANGANESE OR MANGANESES)
456674 MN
      5539 MNS
459951 MN
      (MN OR MNS)
569677 TITANIUM
      82 TITANIUMS
569686 TITANIUM
      (TITANIUM OR TITANIUMS)
238488 ZIRCONIUM
      23 ZIRCONIUMS
238491 ZIRCONIUM
      (ZIRCONIUM OR ZIRCONIUMS)
L21      12 L20 AND (MAGNESIUM OR MG) AND (MANGANESE OR MN) AND (TITANIUM
      OR ZIRCONIUM)

=> s l19 and (magnesium or mg) and (manganese or mn)
L22      139 S L19
      552337 MAGNESIUM
      91 MAGNESIUMS
      552372 MAGNESIUM
      (MAGNESIUM OR MAGNESIUMS)
1548880 MG
      1729 MGS
1550063 MG
      (MG OR MGS)
420771 MANGANESE
      116 MANGANESES
420783 MANGANESE
      (MANGANESE OR MANGANESES)
456674 MN
      5539 MNS
459951 MN
      (MN OR MNS)
L23      54 L22 AND (MAGNESIUM OR MG) AND (MANGANESE OR MN)

=> d scan l23

L23      54 ANSWERS      HCAPLUS      COPYRIGHT 2009 ACS on STN
IC      ICM      B23K035-28
CC      56-9 (Nonferrous Metals and Alloys)
TI      Cored wire electrode for the joint welding of high-strength
aluminum alloys
ST      cored wire electrode joint welding aluminum alloy; welding light
metal aerospace industry cored wire electrode; automobile light metal
construction cored wire electrode welding
IT      Aerospace industry
      (aviation and aeronautics; cored wire electrode for the joint welding
of high-strength aluminum alloys applied for)
IT      Automobiles
      (cored wire electrode for the joint welding of high-strength
aluminum alloys applied for)
IT      Welding of metals
      (flux-cored arc, electrodes; for the joint welding of high-strength
aluminum alloys)
IT      841260-31-9      841260-32-0 841260-33-1
      RL: CPS (Chemical process); DEV (Device component use); PEP (Physical,
engineering or chemical process); PROC (Process); USES (Uses)
      (cored wire electrode for the joint welding of high-strength
aluminum alloys)
IT      11145-78-1, AlMg3Mn      12616-86-3, AlMg4.5Mn0.7      12720-80-8, AlMg4

```

37202-63-4, AlMg4.5Mn0.4

RL: CPS (Chemical process); DEV (Device component use); PEP (Physical, engineering or chemical process); PROC (Process); USES (Uses)
(filler material for cored wire electrode for the joint welding of high-strength aluminum alloys)

HOW MANY MORE ANSWERS DO YOU WISH TO SCAN? (1):0

=> d scan 121

L21 12 ANSWERS HCAPLUS COPYRIGHT 2009 ACS on STN

CC 56-13 (Nonferrous Metals and Alloys)

TI XRFs determination of 10 alloying elements in superhard aluminum alloys

ST alloying element aluminum alloy compn

IT 7439-89-6, Iron, analysis 7439-95-4, Magnesium, analysis

7439-96-5, Manganese, analysis 7440-02-0, Nickel, analysis

7440-21-3, Silicon, analysis 7440-32-6, Titanium, analysis

7440-47-3, Chromium, analysis 7440-66-6, Zinc, analysis 7440-67-7, Zirconium, analysis

RL: ANT (Analyte); ANST (Analytical study)

(XRFs determination of 10 alloying elements in superhard aluminum alloys)

IT 918789-28-3, Aluminum 81-98, chromium 0.1-0.4, copper

0.1-3, iron 0.1-0.7, magnesium 0.3-4, manganese

0.1-0.9, nickel 0-0.2, silicon 0.1-0.8, titanium 0-0.2, zinc

1.4-8.4, zirconium 0-0.3

RL: NUU (Other use, unclassified); USES (Uses)

(sample; XRFs determination of 10 alloying elements in superhard aluminum alloys)

HOW MANY MORE ANSWERS DO YOU WISH TO SCAN? (1):0

=> s 119 and (aluminum or aluminium) and (alloy or casting or die-casting)

L24 139 S L19

1088426 ALUMINUM

312 ALUMINIUMS

1088490 ALUMINUM

(ALUMINUM OR ALUMINIUMS)

17807 ALUMINIUM

36 ALUMINIUMS

17836 ALUMINIUM

(ALUMINIUM OR ALUMINIUMS)

747215 ALLOY

560056 ALLOYS

933573 ALLOY

(ALLOY OR ALLOYS)

164042 CASTING

34741 CASTINGS

177263 CASTING

(CASTING OR CASTINGS)

91168 DIE

22194 DIES

1630 DICE

150 DICES

101211 DIE

(DIE OR DIES OR DICE OR DICES)

164042 CASTING

34741 CASTINGS

177263 CASTING

(CASTING OR CASTINGS)

7552 DIE-CASTING

(DIE(W)CASTING)
L25 139 L24 AND (ALUMINUM OR ALUMINIUM) AND (ALLOY OR CASTING OR DIE-CAS
TING)

=> s l24 and high toughness
4576285 HIGH
665 HIGHS
4576666 HIGH
(HIGH OR HIGHS)
88073 TOUGHNESS
549 TOUGHNESSES
88190 TOUGHNESS
(TOUGHNESS OR TOUGHNESSES)
6608 HIGH TOUGHNESS
(HIGH(W)TOUGHNESS)

L26 7 L24 AND HIGH TOUGHNESS

=> d l26

L26 ANSWER 1 OF 7 HCAPLUS COPYRIGHT 2009 ACS on STN
AN 2006:122425 HCAPLUS
DN 144:175087
TI Aluminum alloys for mushy-state casting of
automotive chassis
IN Minakami, Takahiro; Toyota, Yusuke; Shibata, Katsuhiko;
Murakata, Ryoichi
PA Honda Motor Co., Ltd., Japan
SO Jpn. Kokai Tokkyo Koho, 13 pp.
CODEN: JKXXAF

DT Patent
LA Japanese
FAN.CNT 1

	PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
	-----	----	-----	-----	-----
PI	JP 2006037190	A	20060209	JP 2004-221225	20040729
PRAI	JP 2004-221225		20040729		

=> 1

1 IS NOT A RECOGNIZED COMMAND

The previous command name entered was not recognized by the system.
For a list of commands available to you in the current file, enter
"HELP COMMANDS" at an arrow prompt (=>).

=> d l26

L26 ANSWER 1 OF 7 HCAPLUS COPYRIGHT 2009 ACS on STN
AN 2006:122425 HCAPLUS
DN 144:175087
TI Aluminum alloys for mushy-state casting of
automotive chassis
IN Minakami, Takahiro; Toyota, Yusuke; Shibata, Katsuhiko;
Murakata, Ryoichi
PA Honda Motor Co., Ltd., Japan
SO Jpn. Kokai Tokkyo Koho, 13 pp.
CODEN: JKXXAF

DT Patent
LA Japanese
FAN.CNT 1

	PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
	-----	----	-----	-----	-----
PI	JP 2006037190	A	20060209	JP 2004-221225	20040729

PRAI JP 2004-221225

20040729

=> d 126 2

L26 ANSWER 2 OF 7 HCAPLUS COPYRIGHT 2009 ACS on STN
AN 2005:1285265 HCAPLUS
DN 144:25687
TI High toughness aluminum alloy cast
for automobile parts
IN Toyota, Yusuke; Shibata, Katsuhiko; Minakami, Takahiro;
Murakashi, Ryoichi
PA Honda Motor Co., Ltd., Japan
SO Jpn. Kokai Tokkyo Koho, 8 pp.
CODEN: JKXXAF
DT Patent
LA Japanese
FAN.CNT 1

	PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
	-----	----	-----	-----	-----
PI	JP 2005336569	A	20051208	JP 2004-158760	20040528
	JP 4238181	B2	20090311		
PRAI	JP 2004-158760		20040528		

=> d 126 3

L26 ANSWER 3 OF 7 HCAPLUS COPYRIGHT 2009 ACS on STN
AN 2005:1283373 HCAPLUS
DN 144:25665
TI High-toughness aluminum alloy
casting and its production method
IN Toyota, Yusuke; Shibata, Katsuhiko; Minakami, Takahiro;
Murakashi, Ryoichi
PA Honda Motor Co., Ltd., Japan
SO Jpn. Kokai Tokkyo Koho, 10 pp.
CODEN: JKXXAF
DT Patent
LA Japanese
FAN.CNT 1

	PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
	-----	----	-----	-----	-----
PI	JP 2005336568	A	20051208	JP 2004-158757	20040528
	JP 4238180	B2	20090311		
PRAI	JP 2004-158757		20040528		

=> d 126 4

L26 ANSWER 4 OF 7 HCAPLUS COPYRIGHT 2009 ACS on STN
AN 2005:586890 HCAPLUS
DN 143:101204
TI Cast aluminum alloys with high
toughness and their manufacture
IN Toyota, Yusuke; Minakami, Takahiro; Shibata, Katsuhiko
PA Honda Motor Co., Ltd., Japan
SO Jpn. Kokai Tokkyo Koho, 15 pp.
CODEN: JKXXAF
DT Patent
LA Japanese
FAN.CNT 1

	PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
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	PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
PI	JP 2005177791	A	20050707	JP 2003-420405	20031218
PRAI	JP 2003-420405		20031218		

=> d 126 5

L26 ANSWER 5 OF 7 HCAPLUS COPYRIGHT 2009 ACS on STN
 AN 2004:1125675 HCAPLUS
 DN 142:60742
 TI Aluminum-silicon-base alloy cast products with
 high toughness and stress corrosion cracking resistance
 and their manufacture
 IN Nakamura, Takeyoshi; Shibata, Katsuhiko
 PA Honda Motor Co., Ltd., Japan
 SO Jpn. Kokai Tokkyo Koho, 9 pp.
 CODEN: JKXXAF
 DT Patent
 LA Japanese
 FAN.CNT 1

	PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
PI	JP 2004359988	A	20041224	JP 2003-157903	20030603
	JP 4092255	B2	20080528		
PRAI	JP 2003-157903		20030603		

=> d 126 6

L26 ANSWER 6 OF 7 HCAPLUS COPYRIGHT 2009 ACS on STN
 AN 2004:1058568 HCAPLUS
 DN 142:42199
 TI Manufacture of Al-Si alloy cast having high
 toughness and stress corrosion cracking resistance
 IN Nakamura, Takeyoshi; Shibata, Katsuhiko; Minakami, Takahiro
 PA Honda Motor Co., Ltd., Japan
 SO Jpn. Kokai Tokkyo Koho, 11 pp.
 CODEN: JKXXAF
 DT Patent
 LA Japanese
 FAN.CNT 1

	PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
PI	JP 2004346408	A	20041209	JP 2003-147850	20030526
PRAI	JP 2003-147850		20030526		

=> d 126 7

L26 ANSWER 7 OF 7 HCAPLUS COPYRIGHT 2009 ACS on STN
 AN 2003:972274 HCAPLUS
 DN 140:7577
 TI Die casting having high toughness
 IN Toyoda, Yusuke; Mizukami, Takahiro; Fukuchi,
 Fumiaki; Hata, Tsunehisa; Shibata, Katsuhiko
 PA Honda Giken Kogyo Kabushiki Kaisha, Japan
 SO PCT Int. Appl., 19 pp.
 CODEN: PIXXD2
 DT Patent
 LA Japanese
 FAN.CNT 2

	PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
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PI   WO 2003102257      A1      20031211      WO 2003-JP5993      20030514
      W:  AE, AG, AL, AM, AT, AU, AZ, BA, BB, BG, BR, BY, BZ, CA, CH, CN,
          CO, CR, CU, CZ, DE, DK, DM, DZ, EC, EE, ES, FI, GB, GD, GE, GH,
          GM, HR, HU, ID, IL, IN, IS, KE, KG, KP, KR, KZ, LC, LK, LR, LS,
          LT, LU, LV, MA, MD, MG, MK, MN, MW, MX, MZ, NI, NO, NZ, OM, PH,
          PL, PT, RO, RU, SC, SD, SE, SG, SK, SL, TJ, TM, TN, TR, TT, TZ,
          UA, UG, US, UZ, VC, VN, YU, ZA, ZM, ZW
      RW:  GH, GM, KE, LS, MW, MZ, SD, SL, SZ, TZ, UG, ZM, ZW, AM, AZ, BY,
          KG, KZ, MD, RU, TJ, TM, AT, BE, BG, CH, CY, CZ, DE, DK, EE, ES,
          FI, FR, GB, GR, HU, IE, IT, LU, MC, NL, PT, RO, SE, SI, SK, TR,
          BF, BJ, CF, CG, CI, CM, GA, GN, GQ, GW, ML, MR, NE, SN, TD, TG
      JP 2003342664      A      20031203      JP 2002-157329      20020530
      JP 4092138         B2      20080528
      JP 2004001010      A      20040108      JP 2002-157328      20020530
      JP 4210473         B2      20090121
      AU 2003235302      A1      20031219      AU 2003-235302      20030514
      EP 1508627         A1      20050223      EP 2003-723374      20030514
          R:  AT, BE, CH, DE, DK, ES, FR, GB, GR, IT, LI, LU, NL, SE, MC, PT,
              IE, SI, LT, LV, FI, RO, MK, CY, AL, TR, BG, CZ, EE, HU, SK
      US 20060137848      A1      20060629      US 2005-518151      20050927
PRAI JP 2002-157328      A      20020530
      JP 2002-157329      A      20020530
      WO 2003-JP5993      W      20030514
RE.CNT 5      THERE ARE 5 CITED REFERENCES AVAILABLE FOR THIS RECORD
          ALL CITATIONS AVAILABLE IN THE RE FORMAT

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=> d 126 7 it st cc

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L26  ANSWER 7 OF 7  HCAPLUS  COPYRIGHT 2009 ACS on STN
IT   Casting of metals
      Toughness
          (die casting having high
            toughness of aluminum-magnesium alloy)
IT   116658-27-6  627892-55-1 627892-56-2  627892-57-3
      627892-58-4 627892-59-5 627892-60-8
      627892-61-9  627892-62-0
      RL: PEP (Physical, engineering or chemical process); PRP (Properties); PYP
          (Physical process); PROC (Process)
          (die casting having high
            toughness of aluminum-magnesium alloy)
ST   aluminum magnesium alloy die casting
      high toughness
CC   56-2 (Nonferrous Metals and Alloys)

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=> FIL REGISTRY

COST IN U.S. DOLLARS	SINCE FILE ENTRY	TOTAL SESSION
FULL ESTIMATED COST	67.38	157.04

FILE 'REGISTRY' ENTERED AT 09:21:27 ON 30 MAR 2009
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 provided by InfoChem.

STRUCTURE FILE UPDATES: 27 MAR 2009 HIGHEST RN 1128305-29-2

DICTIONARY FILE UPDATES: 27 MAR 2009 HIGHEST RN 1128305-29-2

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TSCA INFORMATION NOW CURRENT THROUGH January 9, 2009.

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REGISTRY includes numerically searchable data for experimental and
predicted properties as well as tags indicating availability of
experimental property data in the original document. For information
on property searching in REGISTRY, refer to:

<http://www.cas.org/support/stngen/stndoc/properties.html>

=> S 116658-27-6/RN

L27 1 116658-27-6/RN

=> SET NOTICE 1 DISPLAY

NOTICE SET TO 1 U.S. DOLLAR FOR DISPLAY COMMAND
SET COMMAND COMPLETED

=> D L27 SQIDE 1-

YOU HAVE REQUESTED DATA FROM 1 ANSWERS - CONTINUE? Y/(N):y
THE ESTIMATED COST FOR THIS REQUEST IS 6.85 U.S. DOLLARS
DO YOU WANT TO CONTINUE WITH THIS REQUEST? (Y)/N:y

L27 ANSWER 1 OF 1 REGISTRY COPYRIGHT 2009 ACS on STN
RN 116658-27-6 REGISTRY
CN Aluminum alloy, base, Al 98,Mg 1,Mn 1,Fe 0.2,Si 0.2 (9CI) (CA INDEX NAME)
MF Al . Fe . Mg . Mn . Si
CI AYS
SR CA
LC STN Files: CA, CAPLUS, USPATFULL
DT.CA CAplus document type: Patent
RL.P Roles from patents: MSC (Miscellaneous); PROC (Process); PRP
(Properties); USES (Uses)

Component	Component Percent	Component Registry Number
Al	98	7429-90-5
Mg	1	7439-95-4
Mn	1	7439-96-5
Fe	0.2	7439-89-6
Si	0.2	7440-21-3

PROPERTY DATA AVAILABLE IN THE 'PROP' FORMAT

4 REFERENCES IN FILE CA (1907 TO DATE)
4 REFERENCES IN FILE CAPLUS (1907 TO DATE)

=> SET NOTICE LOGIN DISPLAY

NOTICE SET TO OFF FOR DISPLAY COMMAND
SET COMMAND COMPLETED

=>

=> s l19 and (toughness or die-casting or casting or manganese or mn or magnesium or mg)

L19 CANNOT BE SEARCHED IN REGISTRY

The L-number cannot be used because it does not contain a query.

Enter DISPLAY HISTORY to see the sequence of commands that created this L-number.

=> file (hcaplus, inspec, aluminium, compendex, confsci, corrosion, ema, epfull, gbfull, mdf, metadex, scisearch, uspatall)

COST IN U.S. DOLLARS	SINCE FILE ENTRY	TOTAL SESSION
FULL ESTIMATED COST	3.49	160.53

FILE 'HCAPLUS' ENTERED AT 09:23:11 ON 30 MAR 2009

USE IS SUBJECT TO THE TERMS OF YOUR STN CUSTOMER AGREEMENT.

PLEASE SEE "HELP USAGETERMS" FOR DETAILS.

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FILE COVERS 1907 - 30 Mar 2009 VOL 150 ISS 14

FILE LAST UPDATED: 29 Mar 2009 (20090329/ED)

HCAplus now includes complete International Patent Classification (IPC) reclassification data for the third quarter of 2008.

CAS Information Use Policies apply and are available at:

<http://www.cas.org/legal/infopolicy.html>

This file contains CAS Registry Numbers for easy and accurate substance identification.

=> s l19 and (toughness or die-casting or casting or manganese or mn or magnesium or mg)

L28 139 S L19
 88073 TOUGHNESS
 549 TOUGHNESSES
 88190 TOUGHNESS
 (TOUGHNESS OR TOUGHNESSES)
 91168 DIE
 22194 DIES
 1630 DICE
 150 DICES
 101211 DIE
 (DIE OR DIES OR DICE OR DICES)
 164042 CASTING
 34741 CASTINGS
 177263 CASTING
 (CASTING OR CASTINGS)

7552 DIE-CASTING
 (DIE(W)CASTING)
 164042 CASTING
 34741 CASTINGS
 177263 CASTING
 (CASTING OR CASTINGS)
 420771 MANGANESE
 116 MANGANESES
 420783 MANGANESE
 (MANGANESE OR MANGANESES)
 456674 MN
 5539 MNS
 459951 MN
 (MN OR MNS)
 552337 MAGNESIUM
 91 MAGNESIUMS
 552372 MAGNESIUM
 (MAGNESIUM OR MAGNESIUMS)
 1548880 MG
 1729 MGS
 1550063 MG
 (MG OR MGS)
 L29 96 L28 AND (TOUGHNESS OR DIE-CASTING OR CASTING OR MANGANESE OR MN
 OR MAGNESIUM OR MG)

=> dup rem l29
 PROCESSING COMPLETED FOR L29
 L30 96 DUP REM L29 (0 DUPLICATES REMOVED)

=> d l30 1-96 ibib, abs it

L30 ANSWER 1 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN
 ACCESSION NUMBER: 2009:52639 HCAPLUS
 DOCUMENT NUMBER: 150:103298
 TITLE: Aluminum-magnesium-silicon alloy
 sheets for warm forming, and their manufacture
 INVENTOR(S): Kudo, Takeshi; Matsumoto, Katsushi; Ariga, Yasuhiro
 PATENT ASSIGNEE(S): Kobe Steel, Ltd., Japan
 SOURCE: Jpn. Kokai Tokkyo Koho, 22pp.
 CODEN: JKXXAF
 DOCUMENT TYPE: Patent
 LANGUAGE: Japanese
 FAMILY ACC. NUM. COUNT: 1
 PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
-----	---	-----	-----	-----
JP 2009007617	A	20090115	JP 2007-169220	20070627
PRIORITY APPLN. INFO.:			JP 2007-169220	20070627

AB The Al alloy sheets contain Mg 0.57-4.5 and Si 0.33-2.5 weight% by
 satisfying condition A for Mg 0.57-3.8 weight% and (0.578 +
 Mg) ≤ Si ≤ [(0.578 + Mg) + 0.3] and
 condition B for Mg 0.57-4.5 weight% and Si 0.33-2.2 weight% and
 [(0.578 + Mg) - 0.4] ≤ Si ≤ [(0.578 +
 Mg)] and have texture showing average total area ratio 20-65% for Cube
 orientation, Brass orientation, S orientation, and Cu orientation, average
 area ratio for Cube orientation 5-15%, and average grain size 10-50 μm.
 Optionally, the Al alloy sheets contain (1) Cu 0.05-0.5 and/or (2) Fe
 ≤ 1.5, Ti ≤ 0.2, Mn ≤ 1.0, Cr ≤ 0.5, Zr
 ≤ 0.5, V ≤ 0.3, and Zn ≤ 1.5 weight%. The sheets are
 manufactured by casting Al alloy ingots containing the above compns. and
 having thickness ≤ 100 mm, homogenizing, hot rolling for draft

≤92%, and then cold rolling for draft ≤92% to give
≤2.0 mm thickness.

IT Rolling (metals)
(hot; manufacture of aluminum-magnesium-silicon alloy
sheets for warm forming)

IT Casting of metals
Cold rolling
Homogenization
Texture (metallographic)
(manufacture of aluminum-magnesium-silicon alloy sheets
for warm forming)

IT 39299-11-1 53208-42-7 71045-22-2 1095751-97-5 1095751-98-6
1095751-99-7 1095752-00-3
RL: PEP (Physical, engineering or chemical process); TEM (Technical or
engineered material use); PROC (Process); USES (Uses)
(manufacture of aluminum-magnesium-silicon alloy sheets
for warm forming)

L30 ANSWER 2 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2008:1244978 HCAPLUS

DOCUMENT NUMBER: 149:476259

TITLE: Method for producing aluminum alloy thick
plate and aluminum alloy thick plate

INVENTOR(S): Kobayashi, Kazunori; Tokuda, Kenji; Kato, Tomoharu;
Inaba, Takashi

PATENT ASSIGNEE(S): Kabushiki Kaisha Kobe Seiko Sho, Japan

SOURCE: PCT Int. Appl., 105pp.

CODEN: PIXXD2

DOCUMENT TYPE: Patent

LANGUAGE: Japanese

FAMILY ACC. NUM. COUNT: 1

PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
WO 2008123355	A1	20081016	WO 2008-JP55873	20080327
W:	AE, AG, AL, AM, AO, AT, AU, AZ, BA, BB, BG, BH, BR, BW, BY, BZ, CA, CH, CN, CO, CR, CU, CZ, DE, DK, DM, DO, DZ, EC, EE, EG, ES, FI, GB, GD, GE, GH, GM, GT, HN, HR, HU, ID, IL, IN, IS, KE, KG, KM, KN, KP, KR, KZ, LA, LC, LK, LR, LS, LT, LU, LY, MA, MD, ME, MG, MK, MN, MW, MX, MY, MZ, NA, NG, NI, NO, NZ, OM, PG, PH, PL, PT, RO, RS, RU, SC, SD, SE, SG, SK, SL, SM, SV, SY, TJ, TM, TN, TR, TT, TZ, UA, UG, US, UZ, VC, VN, ZA, ZM, ZW			
RW:	AT, BE, BG, CH, CY, CZ, DE, DK, EE, ES, FI, FR, GB, GR, HR, HU, IE, IS, IT, LT, LU, LV, MC, MT, NL, NO, PL, PT, RO, SE, SI, SK, TR, BF, BJ, CF, CG, CI, CM, GA, GN, GQ, GW, ML, MR, NE, SN, TD, TG, BW, GH, GM, KE, LS, MW, MZ, NA, SD, SL, SZ, TZ, UG, ZM, ZW, AM, AZ, BY, KG, KZ, MD, RU, TJ, TM			
JP 2008255371	A	20081023	JP 2007-95419	20070330
JP 4231529	B2	20090304		
JP 2008255372	A	20081023	JP 2007-95423	20070330
JP 4231530	B2	20090304		
JP 2008255411	A	20081023	JP 2007-98495	20070404
JP 4242429	B2	20090325		

PRIORITY APPLN. INFO.: JP 2007-95419 A 20070330
JP 2007-95423 A 20070330
JP 2007-98495 A 20070404

AB Disclosed is a method for producing an aluminum alloy thick
plate, which is characterized by sequentially performing, in the following
order, a melting step wherein an aluminum alloy consisting of a
predetd. amount of Mg, at least one of Si, Fe, Cu, Mn,
Cr, Zn, Ti and Zr, and the balance of Al and unavoidable impurities; a

hydrogen gas removing step wherein a hydrogen gas is removed from the melted aluminum alloy; a filtering step wherein inclusions are removed from the aluminum alloy from which a hydrogen gas is removed; a casting step wherein an ingot is produced by casting the aluminum alloy from which inclusions are removed; a slicing step wherein an aluminum alloy thick plate having a predetd. thickness is produced by slicing the ingot; and a heat treatment step wherein the aluminum alloy thick plate having a predetd. thickness is heat-treated by being maintained at a temperature not

less

than 400°C but less than the m.p. for 1 or more hours.

IT Casting of metals

Dehydrogenation

Filtration

Grain size

Heat treatment

Melting

Nonmetallic inclusions

Tensile strength

Yield strength

(method for producing aluminum alloy thick plate and aluminum alloy thick plate)

IT 12616-83-0, Aa5052 12616-86-3, Aa5083 114323-24-9 122208-88-2
138315-49-8 151975-38-1 261381-04-8 910535-43-2 958460-57-6
958460-65-6 958460-66-7 1071140-51-6 1071140-57-2 1071140-62-9
1071140-65-2 1071140-69-6 1071140-73-2 1071140-78-7 1071140-81-2
1071140-86-7 1071140-91-4 1071141-00-8 1071141-04-2 1071141-07-5
1071141-10-0 1071141-14-4 1071141-16-6 1071141-18-8 1071141-21-3
1071141-30-4 1071141-35-9 1071141-39-3 1071141-43-9 1071141-52-0
1071141-55-3 1071141-58-6 1071141-67-7 1071141-68-8
1071141-71-3

RL: PEP (Physical, engineering or chemical process); PRP (Properties); TEM (Technical or engineered material use); PROC (Process); USES (Uses)

(method for producing aluminum alloy thick plate and aluminum alloy thick plate)

REFERENCE COUNT: 7 THERE ARE 7 CITED REFERENCES AVAILABLE FOR THIS RECORD. ALL CITATIONS AVAILABLE IN THE RE FORMAT

L30 ANSWER 3 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2008:1005642 HCAPLUS

DOCUMENT NUMBER: 149:272995

TITLE: Al-Mg alloy suitable for armor plate applications

INVENTOR(S): Kroepfl, Ingo Guenther; Moritz, Claus Juergen; Moldenhauer, Stefan

PATENT ASSIGNEE(S): Aleris Aluminum Koblenz G.m.b.H., Germany

SOURCE: PCT Int. Appl., 21pp.

CODEN: PIXXD2

DOCUMENT TYPE: Patent

LANGUAGE: English

FAMILY ACC. NUM. COUNT: 1

PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
WO 2008098743	A1	20080821	WO 2008-EP1070	20080212
W:	AE, AG, AL, AM, AO, AT, AU, AZ, BA, BB, BG, BH, BR, BW, BY, BZ, CA, CH, CN, CO, CR, CU, CZ, DE, DK, DM, DO, DZ, EC, EE, EG, ES, FI, GB, GD, GE, GH, GM, GT, HN, HR, HU, ID, IL, IN, IS, JP, KE, KG, KM, KN, KP, KR, KZ, LA, LC, LK, LR, LS, LT, LU, LY, MA, MD, ME, MG, MK, MN, MW, MX, MY, MZ, NA, NG, NI, NO, NZ, OM, PG, PH, PL, PT, RO, RS, RU, SC, SD, SE, SG, SK, SL, SM, SV, SY, TJ, TM,			

TN, TR, TT, TZ, UA, UG, US, UZ, VC, VN, ZA, ZM, ZW
 RW: AT, BE, BG, CH, CY, CZ, DE, DK, EE, ES, FI, FR, GB, GR, HR, HU,
 IE, IS, IT, LT, LU, LV, MC, MT, NL, NO, PL, PT, RO, SE, SI, SK,
 TR, BF, BJ, CF, CG, CI, CM, GA, GN, GQ, GW, ML, MR, NE, SN, TD,
 TG, BW, GH, GM, KE, LS, MW, MZ, NA, SD, SL, SZ, TZ, UG, ZM, ZW,
 AM, AZ, BY, KG, KZ, MD, RU, TJ, TM

PRIORITY APPLN. INFO.: US 2007-889386P P 20070212

AB An aluminum alloy plate has improved resistance against incoming kinetic energy projectiles, the plate having a gauge of 10 mm or more and the aluminum alloy has a chemical composition including, in weight percent: Mg 4.0-6.0, Mn 0.2-1.4, Zn 0.9 maximum, Zr< 0.3, Cr< 0.3, Sc< 0.5, Ti< 0.3, Fe< 0.5, Si< 0.45, Ag< 0.4, Cu<0.25, other elements and unavoidable impurities each <0.05, total <0.20, balance aluminum, and where the alloy plate is obtained by a manufacturing process including casting, preheating and/or homogenization, hot rolling, a first cold working operation, an annealing treatment at <350°, followed by a second cold working operation.

IT Annealing
 Casting of metals
 Cold rolling
 Homogenization
 Metalworking
 (Al-Mg alloy product suitable for armor plate applications)

IT Rolling (metals)
 (hot; Al-Mg alloy product suitable for armor plate applications)

IT Armor
 (plate, military; Al-Mg alloy product suitable for armor plate applications)

IT Heating
 (preheating; Al-Mg alloy product suitable for armor plate applications)

IT 951323-60-7 1047982-87-5
 RL: PEP (Physical, engineering or chemical process); TEM (Technical or engineered material use); PROC (Process); USES (Uses)
 (Al-Mg alloy product suitable for armor plate applications)

REFERENCE COUNT: 4 THERE ARE 4 CITED REFERENCES AVAILABLE FOR THIS RECORD. ALL CITATIONS AVAILABLE IN THE RE FORMAT

L30 ANSWER 4 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2008:1009631 HCAPLUS
 DOCUMENT NUMBER: 149:250704
 TITLE: Hot-rolled aluminum-magnesium alloy plates and method for their manufacture
 INVENTOR(S): Kajiwara, Katsura
 PATENT ASSIGNEE(S): Kobe Steel, Ltd., Japan
 SOURCE: Jpn. Kokai Tokkyo Koho, 13pp.
 CODEN: JKXXAF
 DOCUMENT TYPE: Patent
 LANGUAGE: Japanese
 FAMILY ACC. NUM. COUNT: 1
 PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
JP 2008190021	A	20080821	JP 2007-28291	20070207
PRIORITY APPLN. INFO.:			JP 2007-28291	20070207

AB Claimed are 1.5-10 mm-thick Al-Mg alloy plates containing ≥3 and <5 weight% Mg and having average grain size of ≤50 μm, both at the plate surfaces and at the plate center. The sheets may also contain Mn ≤1.0, Fe ≤0.5, Si ≤0.5, Cr ≤0.4, Zn ≤0.5, Zr ≤0.3, Cu ≤0.6, Ti 0.005-0.2,

and/or B 0.0001-0.05 weight%. Method for manufacture of the plates includes rough

hot rolling of soaked ingot and final finish hot rolling at 250-400° under >50% draft and certain strain rate. The thus manufactured plates are especially suitable for use in automobile, ships, electronic appliances,.

IT Grain size
(hot rolling of Al-Mg alloys for preparation of plates with limited grain size)

IT Rolling (metals)
(hot; hot rolling of Al-Mg alloys for preparation of plates with limited grain size)

IT 12686-54-3 141296-66-4 1045685-45-7 1045685-50-4
RL: PEP (Physical, engineering or chemical process); TEM (Technical or engineered material use); PROC (Process); USES (Uses)
(hot rolling of Al-Mg alloys for preparation of plates with limited grain size)

IT 7440-42-8, Boron, uses
RL: MOA (Modifier or additive use); USES (Uses)
(microalloying element; hot rolling of Al-Mg alloys for preparation of plates with limited grain size)

L30 ANSWER 5 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2008:934507 HCAPLUS

DOCUMENT NUMBER: 149:229185

TITLE: Aluminum alloy sheets and method for their manufacture

INVENTOR(S): Kudo, Takeshi; Matsumoto, Katsushi; Ariga, Yasuhiro

PATENT ASSIGNEE(S): Kobe Steel, Ltd., Japan

SOURCE: Jpn. Kokai Tokkyo Koho, 20pp.
CODEN: JKXXAF

DOCUMENT TYPE: Patent

LANGUAGE: Japanese

FAMILY ACC. NUM. COUNT: 1

PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
-----	----	-----	-----	-----
JP 2008179838	A	20080807	JP 2007-12116	20070123
PRIORITY APPLN. INFO.:			JP 2007-12116	20070123

AB The title Al alloy sheet consists of 0.57-4.5 weight% Mg, 0.33-2.5 weight% Si, and balance Al under satisfaction of (A) $0.578\text{Mg} \leq \text{Si} \leq 0.578\text{Mg} + 0.3$, when $\text{Mg} = 0.57-3.8$, or (B) $0.578\text{Mg} - 0.4 \leq \text{Si} < 0.578\text{Mg}$, when $\text{Mg} = 0.57-4.5$ and $\text{Si} = 0.33-2.2$ (the element symbols indicate their weight% contents), and contains 1-5 area% of total of Mg-Si compds. and Si ppts., in which the ratio of Si precipitate therein is ≤ 1.0 . Optionally, the alloys also contain $\text{Fe} \leq 1.5$, $\text{Mn} \leq 1.0$, $\text{Cr} \leq 0.5$, $\text{Zr} \leq 0.5$, $\text{V} \leq 0.3$, $\text{Ti} \leq 0.2$, $\text{Zn} \leq 1.5$, and/or $\text{Cu} \leq 1.5$ weight%.

Method for manufacture of the sheet includes cast preparation of an ingot having the

above given. chemical compns., homogenization, hot rolling within 20 min after finishing the homogenization step, cold rolling, and solution heat treatment. The sheets have high strength and excellent formability and are suitable for vehicles, machineries, constructions, etc.

IT Precipitates
(Mg Si compds.; manufacture of Al-Mg-Si alloy sheets with formability and strength)

IT Rolling (metals)
(hot; manufacture of Al-Mg-Si alloy sheets with formability and strength)

IT Casting of metals
Cold rolling
Homogenization
(manufacture of Al-Mg-Si alloy sheets with formability and strength)

IT Heat treatment
(solution; manufacture of Al-Mg-Si alloy sheets with formability and strength)

IT 39299-11-1 53208-42-7 71045-22-2 96300-79-7 121439-01-8
201218-44-2 1043448-34-5 1043448-36-7 1043448-37-8 1043448-38-9
1043448-39-0
RL: PEP (Physical, engineering or chemical process); TEM (Technical or engineered material use); PROC (Process); USES (Uses)
(manufacture of Al-Mg-Si alloy sheets with formability and strength)

IT 7440-21-3, Silicon, uses
RL: TEM (Technical or engineered material use); USES (Uses)
(precipitate; manufacture of Al-Mg-Si alloy sheets with formability and strength)

L30 ANSWER 6 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2008:417065 HCAPLUS

DOCUMENT NUMBER: 148:407592

TITLE: Method for evaluation of stress corrosion cracking (SCC) of aluminum alloys and aluminum alloys with excellent resistance to SCC

INVENTOR(S): Sakashita, Shinji; Tanaka, Toshiyuki

PATENT ASSIGNEE(S): Kobe Steel, Ltd., Japan

SOURCE: Jpn. Kokai Tokkyo Koho, 17pp.

CODEN: JKXXAF

DOCUMENT TYPE: Patent

LANGUAGE: Japanese

FAMILY ACC. NUM. COUNT: 1

PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
JP 2008076297	A	20080403	JP 2006-257531	20060922
PRIORITY APPLN. INFO.:			JP 2006-257531	20060922

AB The title process is carried out by determination of the anode polarization curve

of Al alloys in an aqueous 5.8 weight% NaCl solution of pH 10 and 30° by 3-electrode method and evaluation of SCC from the average slope of the elec. current/elec. potential under c.d. of 1-10 A/cm². Al-Mg-Si alloys with the said slope of average value $\leq 350 \Omega \cdot \text{cm}^{-2}$ are also claimed. Preferably, the alloys consist of Mg 0.30-5.0, Si 0.20-2.0, Cu 0.01-2.0, Mn 0.01-1.0, Fe 0.01-1.0, Cr 0.01-2.0, Zn 0.005-10.0, optionally Ti 0.001-0.5, B 0.0001-0.05 Nb 0.01-1.0, Zr 0.01-1.0, and/or V 0.01-1.0 weight%, and balance Al.

IT Stress corrosion cracking

Testing of materials

(evaluation of stress corrosion cracking of Al-Mg-Si alloys and those with excellent stress corrosion cracking resistance)

IT 152677-74-2 333799-16-9 1015163-31-1 1015163-32-2 1015163-33-3
1015163-34-4 1015163-35-5 1015163-36-6 1015163-37-7
1015163-38-8

RL: PRP (Properties); TEM (Technical or engineered material use); USES (Uses)

(evaluation of stress corrosion cracking of Al-Mg-Si alloys and those with excellent stress corrosion cracking resistance)

IT 7440-42-8, Boron, uses

RL: MOA (Modifier or additive use); USES (Uses)
(microalloying element; evaluation of stress corrosion cracking of Al-Mg-Si alloys and those with excellent stress corrosion cracking resistance)

L30 ANSWER 7 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2008:91482 HCAPLUS
DOCUMENT NUMBER: 148:173393
TITLE: Aluminum alloys containing nanocomposite phases
INVENTOR(S): Hung, Wei-Peng; Chen, Chien-Tong
PATENT ASSIGNEE(S): Advanced Material Specialty Inc., Taiwan; Nelson Precision Casting Co., Ltd.
SOURCE: Jpn. Kokai Tokkyo Koho, 10pp.
CODEN: JKXXAF
DOCUMENT TYPE: Patent
LANGUAGE: Japanese
FAMILY ACC. NUM. COUNT: 1
PATENT INFORMATION:

	PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
	-----	----	-----	-----	-----
	JP 2008013826	A	20080124	JP 2006-187815	20060707
PRIORITY APPLN. INFO.:				JP 2006-187815	20060707
AB	The title alloy has a chemical composition contain Mn 1.1-7.0, Mg 0.1-6.0, and Sc 0.01-1.5 weight% and includes long cylindrical nanocomposite phases. Optionally, the alloys also contain Si 0.01-0.5, Fe 0.01-0.10, Cu 0.01-0.50, Cr 0.01-0.50, Ni 0.01-0.50, Ti 0.01-0.1, V 0.01-0.1, Co 0.01-0.1, Zn 0.01-0.1, Zr 0.01-0.1, Nb 0.01-0.1, Mo 0.01-0.1, Y 0.01-0.1, W 0.01-0.1, and/or La 0.01-0.1 weight%. The alloys are especially suitable for golf club heads and golf club shafts.				
IT	Nanocomposites (Al-Mn-Mg-Sc alloys containing nanocomposite phases for golf club heads and shafts)				
IT	Microstructure (columnar nanocomposite; Al-Mn-Mg-Sc alloys containing nanocomposite phases for golf club heads and shafts)				
IT	Sporting goods (golf club heads; Al-Mn-Mg-Sc alloys containing nanocomposite phases for golf club heads and shafts)				
IT	Sporting goods (golf club shafts; Al-Mn-Mg-Sc alloys containing nanocomposite phases for golf club heads and shafts)				
IT	Shafts (golf-club; Al-Mn-Mg-Sc alloys containing nanocomposite phases for golf club heads and shafts)				
IT	1001846-01-0 RL: TEM (Technical or engineered material use); USES (Uses) (Al-Mn-Mg-Sc alloys containing nanocomposite phases for golf club heads and shafts)				

L30 ANSWER 8 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2008:51647 HCAPLUS
DOCUMENT NUMBER: 148:219699
TITLE: Aluminum alloy having nanometer compound phase for golf clubs
INVENTOR(S): Hong, Weipeng; Chen, Jiantong
PATENT ASSIGNEE(S): Amspec Material Inc., Peop. Rep. China; Fu Sheng Group
SOURCE: Faming Zhuanli Shenqing Gongkai Shuomingshu, 10pp.
CODEN: CNXXEV
DOCUMENT TYPE: Patent
LANGUAGE: Chinese

FAMILY ACC. NUM. COUNT: 1

PATENT INFORMATION:

	PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
	-----	----	-----	-----	-----
	CN 101100716	A	20080109	CN 2006-10090373	20060703
PRIORITY APPLN. INFO.:				CN 2006-10090373	20060703
AB	The alloy comprises Mn 1.1-7, Mg 0.1-6, Sc 0.01-1.5%, Al bal. The alloy may further contain Si 0.01-0.5, Fe 0.01-0.1, Cu 0.01-0.5, Cr 0.01-0.5, Ni 0.01-0.5, Ti 0.01-0.1, V 0.01-0.1, Co 0.01-0.1, Zn 0.01-0.1, Zr 0.01-0.1, Nb 0.01-0.1, Mo 0.01-0.1, Y 0.01-0.1, W 0.01-0.1, and/or La 0.01-0.1.				
IT	Phase (aluminum alloy having nanometer compound phase for golf clubs)				
IT	Sporting goods (golf clubs; aluminum alloy having nanometer compound phase for golf clubs)				
IT	Elongation at break Microstructure Tensile strength Yield strength (of aluminum alloy having nanometer compound phase for golf clubs)				
IT	1001846-01-0, Mn 1.1-7, Mg 0.1-6, Sc 0-1.5, Si 0-0.5, Fe 0-0.1, Cu 0-0.5, Cr 0-0.5, Ni 0-0.5, Ti 0-0.1, V 0-0.1, Co 0-0.1, Zn 0-0.1, Zr 0-0.1, Nb 0-0.1, Mo 0-0.1, Y 0-0.1, W 0-0.1, La 0-0.1, Al bal. RL: PRP (Properties); TEM (Technical or engineered material use); USES (Uses) (aluminum alloy having nanometer compound phase for golf clubs)				

L30 ANSWER 9 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2007:1024637 HCAPLUS

DOCUMENT NUMBER: 147:369588

TITLE: Process for manufacturing cast aluminum
alloy plate

INVENTOR(S): Morishita, Makoto

PATENT ASSIGNEE(S): Kabushiki Kaisha Kobe Seiko Sho, Japan

SOURCE: PCT Int. Appl., 20pp.

CODEN: PIXXD2

DOCUMENT TYPE: Patent

LANGUAGE: Japanese

FAMILY ACC. NUM. COUNT: 1

PATENT INFORMATION:

	PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
	-----	----	-----	-----	-----
	WO 2007102290	A1	20070913	WO 2007-JP52040	20070206
W:	AE, AG, AL, AM, AT, AU, AZ, BA, BB, BG, BR, BW, BY, BZ, CA, CH, CN, CO, CR, CU, CZ, DE, DK, DM, DZ, EC, EE, EG, ES, FI, GB, GD, GE, GH, GM, GT, HN, HR, HU, ID, IL, IN, IS, KE, KG, KM, KN, KP, KR, KZ, LA, LC, LK, LR, LS, LT, LU, LV, LY, MA, MD, MG, MK, MN, MW, MX, MY, MZ, NA, NG, NI, NO, NZ, OM, PG, PH, PL, PT, RO, RS, RU, SC, SD, SE, SG, SK, SL, SM, SV, SY, TJ, TM, TN, TR, TT, TZ, UA, UG, US, UZ, VC, VN, ZA, ZM, ZW				
RW:	AT, BE, BG, CH, CY, CZ, DE, DK, EE, ES, FI, FR, GB, GR, HU, IE, IS, IT, LT, LU, LV, MC, NL, PL, PT, RO, SE, SI, SK, TR, BF, BJ, CF, CG, CI, CM, GA, GN, GQ, GW, ML, MR, NE, SN, TD, TG, BW, GH, GM, KE, LS, MW, MZ, NA, SD, SL, SZ, TZ, UG, ZM, ZW, AM, AZ, BY, KG, KZ, MD, RU, TJ, TM				
	JP 2007237237	A	20070920	JP 2006-63050	20060308
	JP 4203508	B2	20090107		

AU 2007224070 A1 20070913 AU 2007-224070 20070206
 CA 2637276 A1 20070913 CA 2007-2637276 20070206
 EP 2011587 A1 20090107 EP 2007-713869 20070206
 R: AT, BE, BG, CH, CY, CZ, DE, DK, EE, ES, FI, FR, GB, GR, HU, IE,
 IS, IT, LI, LT, LU, LV, MC, NL, PL, PT, RO, SE, SI, SK, TR, AL,
 BA, HR, MK, RS
 KR 2008096691 A 20081031 KR 2008-721830 20080905
 PRIORITY APPLN. INFO.: JP 2006-63050 A 20060308
 WO 2007-JP52040 W 20070206

AB A process for manufacturing a cast aluminum alloy plate, in which even in the twin-roll continuous casting process of an Al-Mg aluminum alloy being wide in solid-liquid coexisting temperature range, there can be attained inhibition of defects in the center region of plate thickness. In the process for manufacturing cast plate of Al-Mg aluminum alloy containing a specified amount of Mg and having a large cast plate thickness according to twin-roll continuous casting technique, continuous casting is carried out while having a specified relationship satisfied by D (m) referring to the roll diameter of twin roll, v (m/s) referring to the circumferential velocity of the twin roll, s (m) referring to the solidification distance being the length of roll circumference from point of starting of contact by molten metal with the roll to kiss point and d (m) referring to the thickness of cast plate as a roll gap at the kiss point.

IT Casting of metals

(continuous; for manufacturing cast aluminum alloy plate)

IT 949114-95-8 949114-96-9 949114-97-0 949114-98-1 949114-99-2
 , Aluminum 81-97, chromium 0-0.5, copper 0-0.5, iron 0-1,
 magnesium 3-14, manganese 0-1, silicon 0-0.5, titanium
 0-0.5, vanadium 0-0.3, zinc 0-0.5, zirconium 0-0.3

RL: PEP (Physical, engineering or chemical process); TEM (Technical or engineered material use); PROC (Process); USES (Uses)

(process for manufacturing cast aluminum alloy plate)

REFERENCE COUNT: 2 THERE ARE 2 CITED REFERENCES AVAILABLE FOR THIS
 RECORD. ALL CITATIONS AVAILABLE IN THE RE FORMAT

L30 ANSWER 10 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2007:790072 HCAPLUS

DOCUMENT NUMBER: 147:148324

TITLE: Aluminum alloys for high-temperature and
 high-speed forming, processes for production thereof,
 and process for production of aluminum alloy
 forms

INVENTOR(S): Ichitani, Koji; Tagata, Tsutomu; Komatsubara, Toshio;
 Takata, Ken

PATENT ASSIGNEE(S): Furukawa-Sky Aluminum Corp., Japan; Nippon Steel
 Corporation

SOURCE: PCT Int. Appl., 48pp.

CODEN: PIXXD2

DOCUMENT TYPE: Patent

LANGUAGE: Japanese

FAMILY ACC. NUM. COUNT: 1

PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
WO 2007080938	A1	20070719	WO 2007-JP50276	20070111
W:	AE, AG, AL, AM, AT, AU, AZ, BA, BB, BG, BR, BW, BY, BZ, CA, CH, CN, CO, CR, CU, CZ, DE, DK, DM, DZ, EC, EE, EG, ES, FI, GB, GD, GE, GH, GM, GT, HN, HR, HU, ID, IL, IN, IS, KE, KG, KM, KN, KP, KR, KZ, LA, LC, LK, LR, LS, LT, LU, LV, LY, MA, MD, MG, MK, MN, MW, MX, MY, MZ, NA, NG, NI, NO, NZ, OM, PG, PH, PL, PT, RO, RS, RU, SC, SD, SE, SG, SK, SL, SM, SV, SY, TJ, TM, TN, TR, TT, TZ,			

UA, UG, US, UZ, VC, VN, ZA, ZM, ZW
 RW: AT, BE, BG, CH, CY, CZ, DE, DK, EE, ES, FI, FR, GB, GR, HU, IE,
 IS, IT, LT, LU, LV, MC, NL, PL, PT, RO, SE, SI, SK, TR, BF, BJ,
 CF, CG, CI, CM, GA, GN, GQ, GW, ML, MR, NE, SN, TD, TG, BW, GH,
 GM, KE, LS, MW, MZ, NA, SD, SL, SZ, TZ, UG, ZM, ZW, AM, AZ, BY,
 KG, KZ, MD, RU, TJ, TM

JP 2007186747 A 20070726 JP 2006-5406 20060112
 JP 2007186748 A 20070726 JP 2006-5415 20060112
 EP 1975263 A1 20081001 EP 2007-706623 20070111

R: DE, FR, GB, IT

US 20080257462 A1 20081023 US 2008-171380 20080711

PRIORITY APPLN. INFO.:

JP 2006-5406 A 20060112
 JP 2006-5415 A 20060112
 WO 2007-JP50276 W 20070111

AB An aluminum alloy for high-temperature and high-speed forming contains
 Mg: 2.0 to 8.0, Mn: 0.05 to 1.0, Zr: 0.01 to 0.3, Si:
 0.06 to 0.4, and Fe: 0.06 to 0.4%. The alloy is subjected to forming at
 200-550°C and a strain rate of 10-2-10/s and subsequent cooling to
 room temperature at a rate of ≥20°C/min. In the alloy, Cr-containing
 intermetallic compds. formed in melting and casting have sizes
 of ≤20 μm or below and intermetallic compound particles of 50 to
 1000 nm are present in an amount of 350,000 particles/mm² or above as
 Mn- and Cr-containing ppts.

IT Intermetallic compounds

RL: MOA (Modifier or additive use); USES (Uses)

(Cr-containing; aluminum alloys for high-temperature and high-speed
 forming, processes for production thereof, and process for production of
 aluminum alloy forms)

IT Cooling

Precipitates

Strain

(aluminum alloys for high-temperature and high-speed forming,
 processes for production thereof, and process for production of
 aluminum alloy forms)

IT Cast alloys

RL: PEP (Physical, engineering or chemical process); PRP (Properties); TEM
 (Technical or engineered material use); PROC (Process); USES (Uses)

(aluminum; aluminum alloys for high-temperature and
 high-speed forming, processes for production thereof, and process for
 production of aluminum alloy forms)

IT Particle size

(intermetallic compds.; aluminum alloys for high-temperature and
 high-speed forming, processes for production thereof, and process for
 production of aluminum alloy forms)

IT 943736-16-1 943736-17-2 943736-18-3 943736-19-4 943736-20-7
 943736-21-8 943736-22-9 943736-23-0 943736-24-1 943736-25-2
 943736-27-4 943736-29-6 943736-32-1 943736-34-3 943736-37-6
 943736-40-1 943736-42-3 943736-44-5

RL: PRP (Properties); TEM (Technical or engineered material use); USES
 (Uses)

(aluminum alloys for high-temperature and high-speed forming,
 processes for production thereof, and process for production of
 aluminum alloy forms)

REFERENCE COUNT:

9

THERE ARE 9 CITED REFERENCES AVAILABLE FOR THIS
 RECORD. ALL CITATIONS AVAILABLE IN THE RE FORMAT

L30 ANSWER 11 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2007:201658 HCAPLUS

DOCUMENT NUMBER: 146:278835

TITLE: High strength weldable Al-Mg alloy

INVENTOR(S): Telioui, Nadia; Normann, Andrew

PATENT ASSIGNEE(S): Corus Aluminium Walzprodukte G.m.b.H., Germany;

SOURCE: Meijers, Steven Dirk
PCT Int. Appl., 18pp.
CODEN: PIXXD2
DOCUMENT TYPE: Patent
LANGUAGE: English
FAMILY ACC. NUM. COUNT: 1
PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
WO 2007020041	A2	20070222	WO 2006-EP8030	20060814
WO 2007020041	A3	20070510		
W: AE, AG, AL, AM, AT, AU, AZ, BA, BB, BG, BR, BW, BY, BZ, CA, CH, CN, CO, CR, CU, CZ, DE, DK, DM, DZ, EC, EE, EG, ES, FI, GB, GD, GE, GH, GM, HN, HR, HU, ID, IL, IN, IS, JP, KE, KG, KM, KN, KP, KR, KZ, LA, LC, LK, LR, LS, LT, LU, LV, LY, MA, MD, MG, MK, MN, MW, MX, MZ, NA, NG, NI, NO, NZ, OM, PG, PH, PL, PT, RO, RS, RU, SC, SD, SE, SG, SK, SL, SM, SY, TJ, TM, TN, TR, TT, TZ, UA, UG, US, UZ, VC, VN, ZA, ZM, ZW RW: AT, BE, BG, CH, CY, CZ, DE, DK, EE, ES, FI, FR, GB, GR, HU, IE, IS, IT, LT, LU, LV, MC, NL, PL, PT, RO, SE, SI, SK, TR, BF, BJ, CF, CG, CI, CM, GA, GN, GQ, GW, ML, MR, NE, SN, TD, TG, BW, GH, GM, KE, LS, MW, MZ, NA, SD, SL, SZ, TZ, UG, ZM, ZW, AM, AZ, BY, KG, KZ, MD, RU, TJ, TM, AP, EA, EP, OA FR 2889852 A1 20070223 FR 2006-7305 20060811 CA 2617528 A1 20070222 CA 2006-2617528 20060814 EP 1917373 A2 20080507 EP 2006-776840 20060814 R: AT, BE, BG, CH, CY, CZ, DE, DK, EE, ES, FI, FR, GB, GR, HU, IE, IS, IT, LI, LT, LU, LV, MC, NL, PL, PT, RO, SE, SI, SK, TR, AL, BA, HR, MK, RS JP 2009504918 T 20090205 JP 2008-526421 20060814 CN 101233252 A 20080730 CN 2006-80028105 20080131 IN 2008CN00756 A 20081128 IN 2008-CN756 20080213 PRIORITY APPLN. INFO.: EP 2005-76898 A 20050816 WO 2006-EP8030 W 20060814				

AB An aluminum alloy product having high strength, excellent corrosion resistance and weldability, having the following composition in weight%:

Mg 3.5 to 6.0, Mn 0.4 to 1.2, Fe < 0.5, Si < 0.5, Cu < 0.15, Zr < 0.5, Cr < 0.3, Ti 0.03 to 0.2, Sc < 0.5 Zn, < 1.7 Li, < 0.5, and Ag < 0.4, optionally one or more of the following dispersoid forming elements selected from the group consisting of erbium, yttrium, hafnium, vanadium, each < 0.5 wt%, and impurities or incidental elements each < 0.05, total < 0.15 and the balance being aluminum.

IT Tensile strength
Welding of metals
(high strength weldable Al-Mg alloy with excellent corrosion resistance and weldability)
IT Corrosion
(resistance; high strength weldable Al-Mg alloy with excellent corrosion resistance and weldability)
IT 7440-52-0, Erbium, properties 7440-58-6, Hafnium, properties 7440-62-2, Vanadium, properties 7440-65-5, Yttrium, properties
RL: MOA (Modifier or additive use); PRP (Properties); USES (Uses)
(dispersoid forming element; high strength weldable Al-Mg alloy with excellent corrosion resistance and weldability)
IT 926624-86-4, Aluminum 88-96, chromium 0-0.3, copper 0-0.2, iron 0-0.5, lithium 0-0.5, magnesium 3.5-6, manganese 0.4-1.2, scandium 0-0.5, silicon 0-0.5, silver 0-0.4, titanium 0-0.2, zinc 0-1.7, zirconium 0-0.5 926624-88-6, Aluminum 91-95, chromium 0-0.1, copper 0-0.2, iron 0-0.5, lithium 0-0.5, magnesium 3.8-4.3, manganese 0.6-0.9, scandium

0.1-0.3, silicon 0-0.5, silver 0-0.4, titanium 0-0.1, zinc 0.4-0.6,
 zirconium 0-0.2 926624-90-0, Aluminum 91-94, chromium 0-0.1,
 copper 0-0.2, iron 0-0.5, lithium 0-0.5, magnesium 5-5.6,
 manganese 0.6-0.9, scandium 0.1-0.3, silicon 0-0.5, silver 0-0.4,
 titanium 0-0.1, zinc 0.4-0.6, zirconium 0-0.2
 RL: PEP (Physical, engineering or chemical process); PRP (Properties);
 PROC (Process)
 (high strength weldable Al-Mg alloy with excellent corrosion
 resistance and weldability)

L30 ANSWER 12 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2007:228052 HCAPLUS
 DOCUMENT NUMBER: 146:300424
 TITLE: Welded aluminum sheets to improve corrosion
 resistance
 INVENTOR(S): Feng, Frank; Christy, William H.
 PATENT ASSIGNEE(S): Can.
 SOURCE: U.S. Pat. Appl. Publ., 7pp.
 CODEN: USXXCO
 DOCUMENT TYPE: Patent
 LANGUAGE: English
 FAMILY ACC. NUM. COUNT: 1
 PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
US 20070045260	A1	20070301	US 2005-215343	20050830
PRIORITY APPLN. INFO.:			US 2005-215343	20050830
AB An edge of an aluminum 5xxx series sheet is welded to another aluminum 5xxx series sheet utilizing a low magnesium content aluminum alloy filler and GMAW welding technique. The bottom weld seam metal is heat dressed as by a TIG and subsequently the bottom and the top, are planished to the same thickness as the sheets. Favorable properties such as the elimination of nail heads, improved corrosion resistance, and bendability are obtained.				
IT Welding of metals (gas metal-arc; welded aluminum sheets to improve corrosion resistance)				
IT Corrosion (resistance; welded aluminum sheets to improve corrosion resistance)				
IT Welding of metals (welded aluminum sheets to improve corrosion resistance)				
IT 11145-78-1, AA5454 12616-83-0, AA5052 12616-86-3, AA5083 12720-80-8, AA5086 65394-05-0, AA5754 259876-44-3, AA5186 327622-69-5, AA5383 661475-83-8, AA5087 RL: PEP (Physical, engineering or chemical process); TEM (Technical or engineered material use); PROC (Process); USES (Uses) (base metal; welded aluminum sheets to improve corrosion resistance)				
IT 37268-39-6, AA5356 55535-47-2, AA1188 113314-85-5 RL: PEP (Physical, engineering or chemical process); TEM (Technical or engineered material use); PROC (Process); USES (Uses) (filler metal; welded aluminum sheets to improve corrosion resistance)				

L30 ANSWER 13 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2007:550709 HCAPLUS
 DOCUMENT NUMBER: 146:505249
 TITLE: Al-Cu-Mg alloy for aerospace industry with
 improved strength under short and long-term elevated
 temps.

INVENTOR(S): Chirkov, E. F.; Kablov, E. N.; Karimova, S. A.
 PATENT ASSIGNEE(S): FGUP "Vserossiiskii Nauchno-Issled. Inst.
 Aviatsionnykh Materialov", Russia
 SOURCE: Russ., 7pp.
 CODEN: RUXXE7
 DOCUMENT TYPE: Patent
 LANGUAGE: Russian
 FAMILY ACC. NUM. COUNT: 1
 PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
RU 2299256	C1	20070520	RU 2005-140799	20051227
PRIORITY APPLN. INFO.:			RU 2005-140799	20051227

AB The invented method relates to aluminum-copper-magnesium system and provides alloy for manufacturing aerospace-destination welded articles capable of working under loadings not only at ambient temps. but also at short and long-term elevated temperature Alloy has following chemical anal., wt %: copper 4.5-7.0, magnesium 1.75-4.5, manganese 0.25-0.8, titanium 0.05-0.45, iron 0.05-0.45, silicon 0.02-0.2, beryllium 0.001-0.07, hydrogen 1.8 + 10⁻⁶-3.1 + 10⁻⁵, calcium 0.0001-0.08, cobalt 0.02-0.45; at least one of the following elements: nickel 0.001-0.05, chromium 0.001-0.05, or zinc 0.001-0.05; one of the following elements: zirconium 0.055-0.45 or vanadium 0.055-0.45; and aluminum - the balance. The developed deformable aluminum-based alloy and articles made therefrom show good weldability, low hot brittleness, and high strength of welded joint at ambient and elevated temps.

IT Brittle fracture
 Welding of metals
 (Al-Cu-Mg alloy for aerospace industry with improved strength under short and long-term elevated temps.)

IT 936561-06-7 936561-07-8
 RL: PEP (Physical, engineering or chemical process); PRP (Properties); TEM (Technical or engineered material use); PROC (Process); USES (Uses)
 (Al-Cu-Mg alloy for aerospace industry with improved strength under short and long-term elevated temps.)

L30 ANSWER 14 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2007:1293299 HCAPLUS
 DOCUMENT NUMBER: 147:526358
 TITLE: Friction stir welding process for Al-Si and Al-Mg alloys with dissimilar shear strengths

INVENTOR(S): Fukuchi, Fumiaki; Sayama, Mitsuru; Miyahara, Tetsuya; Ishida, Eiji
 PATENT ASSIGNEE(S): Honda Motor Co., Ltd., Japan
 SOURCE: Brit. UK Pat. Appl., 21pp.
 CODEN: BAXXDU

DOCUMENT TYPE: Patent
 LANGUAGE: English
 FAMILY ACC. NUM. COUNT: 1
 PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
GB 2438063	A	20071114	GB 2007-8741	20070504
GB 2438063	B	20090304		
US 20070280849	A1	20071206	US 2007-789606	20070425
PRIORITY APPLN. INFO.:			JP 2006-129595	A 20060508

AB The invention relates to the friction stir welding process for joining two members having different shearing strengths, and friction stir welding

structure fabricated by the process. The friction stir welding process includes positioning the first and second welding members such that both members overlap to define an overlapped region before inserting a rotating pin into the overlapped region from the surface of the second welding member, so that the first and second welding members are joined together. The first welding member (the one furthest from the welding tool) has lower shearing strength than the second welding member. The first and the second welding member may be made of Al-Si alloy, and Al-Mg alloy, resp.

IT Welding of metals
(friction; friction stir welding process for dissimilar metal alloys)

IT 37254-75-4 370070-97-6
RL: TEM (Technical or engineered material use); USES (Uses)
(friction stir welding process for dissimilar metal alloys)

REFERENCE COUNT: 3 THERE ARE 3 CITED REFERENCES AVAILABLE FOR THIS RECORD. ALL CITATIONS AVAILABLE IN THE RE FORMAT

L30 ANSWER 15 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2007:1064778 HCAPLUS

DOCUMENT NUMBER: 147:453255

TITLE: Al alloy sheet for train compartment capable of preventing hot rolling crack and machining deformation, and improving strength at non-proportional extension

INVENTOR(S): Zhong, Li; Xu, Zhongyan; Nie, Bo; Wu, Xinfeng; Qi, Yanhua; Tao, Zhimin; Wang, Guojun; Li, Guangyu

PATENT ASSIGNEE(S): Northeast Light Alloy Co., Ltd., Peop. Rep. China

SOURCE: Faming Zhuanli Shenqing Gongkai Shuomingshu, 9pp.
CODEN: CNXXEV

DOCUMENT TYPE: Patent

LANGUAGE: Chinese

FAMILY ACC. NUM. COUNT: 1

PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
CN 101037744	A	20070919	CN 2007-10072154	20070429
PRIORITY APPLN. INFO.:			CN 2007-10072154	20070429

AB The title Al-alloy plate is prepared from (by weight) Si \leq 0.25%, Fe \leq 0.25%, Cu \leq 0.20%, Mn 0.7-1.0%, Mg 4.0-5.2%, Cr 0.25%, Zn \leq 0.40%, Ti \leq 0.15%, Zr \leq 0.20%, and Al in balance by mixing; smelting; ingot casting; homogenization annealing at 450-480° for 35 h; hot rolling at 450-480°; cold rolling; and stabilizing at 80-100° for 6 h. The prepared Al-alloy plate has the advantage of no hot rolling crack and machining deformation, and improved proof strength at non-proportional extension.

IT Annealing
Casting of metals
Cold rolling
Filtration
Homogenization

(Al alloy sheet for train compartment capable of preventing hot rolling crack and machining deformation, and improving strength at non-proportional extension)

IT Filters
(ceramic; Al alloy sheet for train compartment capable of preventing hot rolling crack and machining deformation, and improving strength at non-proportional extension)

IT Ceramics
(filters; Al alloy sheet for train compartment capable of preventing hot rolling crack and machining deformation, and improving strength at

non-proportional extension)
 IT Rolling (metals)
 (hot; Al alloy sheet for train compartment capable of preventing hot rolling crack and machining deformation, and improving strength at non-proportional extension)
 IT 952105-95-2
 RL: PEP (Physical, engineering or chemical process); PRP (Properties); TEM (Technical or engineered material use); PROC (Process); USES (Uses)
 (Al alloy sheet for train compartment capable of preventing hot rolling crack and machining deformation, and improving strength at non-proportional extension)
 IT 12617-27-5 13463-67-7, Titania, uses 16923-95-8, Potassium hexafluoro-zirconate 39364-34-6 39364-47-1
 RL: TEM (Technical or engineered material use); USES (Uses)
 (Al alloy sheet for train compartment capable of preventing hot rolling crack and machining deformation, and improving strength at non-proportional extension)

L30 ANSWER 16 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2006:735523 HCAPLUS

DOCUMENT NUMBER: 145:193265

TITLE: Aluminum-based alloy for aviation and shipbuilding

INVENTOR(S): Popov, V. I.

PATENT ASSIGNEE(S): OAO "Kamensk-Ural'skii Metallurgicheskii Zavod", Russia

SOURCE: Russ., 13 pp.

CODEN: RUXXE7

DOCUMENT TYPE: Patent

LANGUAGE: Russian

FAMILY ACC. NUM. COUNT: 1

PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
RU 2280705	C2	20060727	RU 2004-127634	20040915

PRIORITY APPLN. INFO.: RU 2004-127634 20040915

AB The invention is suitable in metallurgy of aluminum alloys, especially Al-Mg-Mn alloys, for the manufacture of armored semi-finished products and articles for aviation and shipbuilding. The alloy contains Mg 4.2-6.5, Mn 0.5-1.2, Zn ≤0.2, Cr ≤0.2, Ti ≤0.15, Si ≤0.25, Fe ≤0.3, Cu ≤0.1, Zr ≤0.05-0.3 weight%, and at least one element selected from Sc, 0.05-0.3 weight%, Be 0.0001-0.01 weight%, Y 0.001-0.1 weight%, Nd 0.001-0.1 weight%, Ce 0.001-0.1 weight%, and Al in the balance. The resulting alloy and articles made from it have high resistance to ballistic action of various projectiles due to optimal strength characteristics, optimal structure and plasticity characteristics, as well as enhanced corrosion resistance and weldability.

IT Aerospace industry
 (aviation and aeronautics; aluminum-based alloy for aviation and shipbuilding)

IT Armor
 (plate; aluminum-based alloy for aviation and shipbuilding)

IT 902164-10-7 902164-12-9 902164-15-2 902164-18-5
 RL: PRP (Properties); TEM (Technical or engineered material use); USES (Uses)
 (aluminum armor alloy; aluminum-based alloy for aviation and shipbuilding)

IT 902164-07-2
 RL: TEM (Technical or engineered material use); USES (Uses)

(aluminum armor alloy; aluminum-based alloy for
aviation and shipbuilding)

L30 ANSWER 17 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2006:122425 HCAPLUS

DOCUMENT NUMBER: 144:175087

TITLE: Aluminum alloys for mushy-state
casting of automotive chassis

INVENTOR(S): Minakami, Takahiro; Toyota, Yusuke; Shibata,
Katsuhiko; Murakata, Ryoichi

PATENT ASSIGNEE(S): Honda Motor Co., Ltd., Japan

SOURCE: Jpn. Kokai Tokkyo Koho, 13 pp.

CODEN: JKXXAF

DOCUMENT TYPE: Patent

LANGUAGE: Japanese

FAMILY ACC. NUM. COUNT: 1

PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
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JP 2006037190	A	20060209	JP 2004-221225	20040729
PRIORITY APPLN. INFO.:			JP 2004-221225	20040729
AB	The title Al alloys contain 2.0-4.0 weight% Si and are obtained by rapidly cooling mushy-state melt having solid-phase ratio 25-45%. The title process comprises steps of (1) preparing the mushy-state melt at periphery of eutectic point in a container, (2) setting the container to an injection sleeve, (3) closing a die, (4) filling the melt to a cavity by pressing with a plunger, and then (5) cooling at $\geq 5^\circ/\text{s}$. The cast Al alloys provide high toughness.			
IT	Casting of metals Cooling (aluminum alloys for mushy-state casting of automotive chassis)			
IT	Cast alloys RL: DEV (Device component use); USES (Uses) (aluminum; aluminum alloys for mushy-state casting of automotive chassis)			
IT	Automobiles (chassis; aluminum alloys for mushy-state casting of automotive chassis)			
IT	11145-29-2	12609-50-6, Aluminum 97, silicon 3	12686-71-4	12727-35-4
	RL: DEV (Device component use); PEP (Physical, engineering or chemical process); PYP (Physical process); PROC (Process); USES (Uses) (aluminum alloys for mushy-state casting of automotive chassis)			

L30 ANSWER 18 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2006:75775 HCAPLUS

DOCUMENT NUMBER: 144:132643

TITLE: Tough cast aluminum alloys and
method for their manufacture

INVENTOR(S): Toyota, Yusuke; Shibata, Katsuhiko;
Minakami, Takahiro; Murakata, Ryoichi

PATENT ASSIGNEE(S): Honda Motor Co., Ltd., Japan

SOURCE: Jpn. Kokai Tokkyo Koho, 9 pp.

CODEN: JKXXAF

DOCUMENT TYPE: Patent

LANGUAGE: Japanese

FAMILY ACC. NUM. COUNT: 1

PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
JP 2006022385	A	20060126	JP 2004-202713	20040709
PRIORITY APPLN. INFO.:			JP 2004-202713	20040709
AB	The title cast Al alloys consisting of Si 2-4, Mg 0.2-0.5, Cu 0.4-0.8, Ni 0.05-0.3 weight%, and balance Al having tensile strength ≥ 300 MPa, 0.2% yield strength ≥ 210 MPa, and elongation $\geq 10\%$ are claimed. Alloys having the said composition is cast, heat treated at 515-540° and then aged at 165-185° to give the claimed alloys. The alloys are especially suitable for structures in automobiles.			
IT	Aging, materials Heat treatment (manufacture of cast aluminum alloys with balanced elongation and tensile strength)			
IT	873590-90-0	873590-91-1	873590-92-2	873590-93-3
	873590-95-5	873590-96-6	873590-97-7	873590-98-8
	RL: PEP (Physical, engineering or chemical process); PYP (Physical process); TEM (Technical or engineered material use); PROC (Process); USES (Uses) (manufacture of cast aluminum alloys with balanced elongation and tensile strength)			

L30 ANSWER 19 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2008:34587 HCAPLUS

DOCUMENT NUMBER: 148:407301

TITLE: Low-temperature heating changes properties of welded joints of aluminum alloy V92Zr

AUTHOR(S): Svedlin, A. V.

CORPORATE SOURCE: Bradley University, Peoria, IL, USA

SOURCE: Heat Treating, Proceedings of the ASM Heat Treating Society Conference, 23rd, Pittsburgh, PA, United States, Sept. 25-28, 2005 (2006), Meeting Date 2005, 370-375. Editor(s): Herring, Daniel; Hill, Robert. ASM International: Materials Park, Ohio. CODEN: 69KGZS; ISBN: 978-0-87170-832-8

DOCUMENT TYPE: Conference

LANGUAGE: English

AB The high-strength aluminum alloy V92Zr system Al-Zn-Mg is self-quenched alloy. The major alloying elements are 4.2 wt % Mg, 3.2 wt % Zn, 0.6 wt % Mn, and 0.15 wt % Zr. The most acceptable filler wires to weld this alloy are V92W, alloys AMg6, AMg4Zr and No.11 (Al-Zn-Mg). This alloy can be used in the aircraft production. Prolonged heating at 50-70°C can lead to substantial structural changes in precipitation hardening of aluminum alloys due to the transition from zone to phase aging. According to, zone aging of Al-Zn-Mg alloys, particularly the weld seams, with repeated heating at 50-70°C substantially increases the strength and lowers the elongation, reduction in cross-sectional area, toughness, resistance to stress corrosion, and increases susceptibility to cracking. It was shown in that heating at temps. even below the phase aging temperature changes the properties considerably. This article deals with the effect of prolonged low-temperature heating on the mech. properties, sensitivity to cracks in impact bending, and corrosion resistance of semifinished products and weldments of aluminum alloys V92Zr after solution treatment and aging at the room and elevated temps.

IT Bending
Crack (fracture)
Elongation at break
Forging

Fracture toughness
Heating
Polarizability
Tensile strength
Welds
Yield strength

(low-temperature heating changes properties of welded joints of
aluminum alloy V92Zr)

- IT Corrosion
(resistance; low-temperature heating changes properties of welded joints of
aluminum alloy V92Zr)
- IT 12732-16-0 39410-66-7 81159-87-7, AMg4 284685-77-4
1015477-81-2, Aluminum 91, iron 0.2, magnesium 4.4,
manganese 0.8, silicon 0.1, zinc 3.4, zirconium 0.1
1015477-82-3, Aluminum 95, iron 0.1, magnesium 4.1,
manganese 0.4, silicon 0.2, titanium 0.1, zinc 0.1, zirconium 0.1
RL: PEP (Physical, engineering or chemical process); PRP (Properties); TEM
(Technical or engineered material use); PROC (Process); USES (Uses)
(low-temperature heating changes properties of welded joints of
aluminum alloy V92Zr)

REFERENCE COUNT: 8 THERE ARE 8 CITED REFERENCES AVAILABLE FOR THIS
RECORD. ALL CITATIONS AVAILABLE IN THE RE FORMAT

L30 ANSWER 20 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2007:972681 HCAPLUS

DOCUMENT NUMBER: 147:489920

TITLE: Alloy design by spreadsheet

AUTHOR(S): Dupen, Barry

CORPORATE SOURCE: Indiana University - Purdue University, Fort Wayne,
USA

SOURCE: Materials Science & Technology 2006 Conference and
Exhibition, MST&T'06, Cincinnati, OH, United States,
Oct. 15-19, 2006 (2006), facv2/365-facv2/373.
Minerals, Metals & Materials Society: Warrendale, Pa.
CODEN: 69JOEQ

DOCUMENT TYPE: Conference; (computer optical disk)

LANGUAGE: English

AB Com. manufacturers of metallic casting alloys are interested in
composition-dependent materials properties, such as liquidus and solidus
temps., thermal and elec. conductivity, and color (important in dentistry and
jewelry). A two-step method is proposed for predicting compns. which have
the desired properties. First, property data for existing alloys is
entered into a spreadsheet, and matrix algebra is used to calculate the
coeffs. for a multivariable nonlinear regression equation. Second, the
spreadsheet uses the regression equation recursively to predict properties
of all possible alloys within a search field. Results are sorted
according to the target property range. One benefit of using a
spreadsheet is that a small or medium-sized company with limited resources
can develop and run its own alloy property prediction program at
relatively low cost. Unlike com.-available phase diagram prediction
software, this method can be applied to any composition-dependent property of
an alloy system.

IT Electric conductivity
Temperature
Thermal conductivity
(alloy design by spreadsheet)

IT Computer program
(spreadsheet; alloy design by spreadsheet)

IT 954098-28-3, Aluminum 76-99, boron 0-0.1, chromium
0-0.5, copper 0-10, iron 0.1-2, magnesium 0-10,
manganese 0-1.2, nickel 0-2.5, silicon 0-19, titanium 0-0.4,
vanadium 0-0.1, zinc 0-7.7, zirconium 0-1, tin 0-6.2

RL: PRP (Properties)
 (alloy design by spreadsheet)
 REFERENCE COUNT: 5 THERE ARE 5 CITED REFERENCES AVAILABLE FOR THIS
 RECORD. ALL CITATIONS AVAILABLE IN THE RE FORMAT

L30 ANSWER 21 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN
 ACCESSION NUMBER: 2005:1077850 HCAPLUS
 DOCUMENT NUMBER: 143:351105
 TITLE: Cast and weldable Al-Si based alloy
 and alloy member made therefrom
 INVENTOR(S): Fukuchi, Fumiaki; Yahaba, Takanori
 PATENT ASSIGNEE(S): Japan
 SOURCE: U.S. Pat. Appl. Publ., 10 pp.
 CODEN: USXXCO
 DOCUMENT TYPE: Patent
 LANGUAGE: English
 FAMILY ACC. NUM. COUNT: 1
 PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
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US 20050220660	A1	20051006	US 2005-92978	20050330
JP 2005281829	A	20051013	JP 2004-101044	20040330
DE 102005014485	A1	20060105	DE 2005-102005014485	20050330
PRIORITY APPLN. INFO.:			JP 2004-101044	A 20040330

AB An Al-Si based alloy and an alloy member
 made of the alloy are disclosed, in which when alloys
 produced by die casting under high vacuum conditions
 are welded, weldability can be improved without increasing plate thickness
 of welded portions and reducing gas content in die
 casting. The alloy contains Si 7.5-9, Mg
 0.2-0.4, Mn 0.3-0.5, Cu 0.03-0.2, Fe 0.1-0.25, Sr 0.0050.02
 weight%, and aluminum in the balance.

IT Elongation at break
 Impact strength
 (cast Al-Si based alloy and alloy member
 made therefrom)

IT Casting of metals
 (die; cast Al-Si based alloy and alloy
 member made therefrom)

IT Welding of metals
 (gas tungsten-arc; cast Al-Si based alloy and
 alloy member made therefrom)

IT Aging, materials
 (of cast alloy; cast Al-Si based alloy
 and alloy member made therefrom)

IT Tensile strength
 (ultimate; cast Al-Si based alloy and alloy
 member made therefrom)

IT 866035-22-5 866035-23-6 866035-24-7 866035-25-8 866035-26-9
 RL: PEP (Physical, engineering or chemical process); PYP (Physical
 process); TEM (Technical or engineered material use); PROC (Process); USES
 (Uses)
 (cast Al-Si based alloy and alloy member
 made therefrom)

IT 866035-21-4
 RL: PEP (Physical, engineering or chemical process); PYP (Physical
 process); TEM (Technical or engineered material use); PROC (Process); USES
 (Uses)
 (claim 1; cast Al-Si based alloy and alloy
 member made therefrom)

L30 ANSWER 22 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2005:1285265 HCAPLUS
DOCUMENT NUMBER: 144:25687
TITLE: High toughness aluminum
alloy cast for automobile parts
INVENTOR(S): Toyota, Yusuke; Shibata, Katsuhiro;
Minakami, Takahiro; Murakashi, Ryoichi
PATENT ASSIGNEE(S): Honda Motor Co., Ltd., Japan
SOURCE: Jpn. Kokai Tokkyo Koho, 8 pp.
CODEN: JKXXAF
DOCUMENT TYPE: Patent
LANGUAGE: Japanese
FAMILY ACC. NUM. COUNT: 1
PATENT INFORMATION:

	PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
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	JP 2005336569	A	20051208	JP 2004-158760	20040528
	JP 4238181	B2	20090311		
PRIORITY APPLN. INFO.:				JP 2004-158760	20040528
AB	The alloy with tensile strength ≥ 280 MPa, yield strength ≥ 220 MPa, and elongation $\geq 12\%$ comprises Si 2-5, Mg 0.2-0.5, Cu 0.4-0.8, Ge 0.05-0.3%, and Al bal. The alloy may further contain Zr, Ti, and/or B.				
IT	Elongation, mechanical Impact strength Tensile strength Yield strength (of aluminum alloy cast for automobile parts)				
IT	205579-01-7	870462-29-6	870525-27-2	870525-28-3	870525-29-4
	870525-30-7	870525-31-8	870525-32-9	870525-33-0	870525-34-1
	870525-35-2	870525-36-3	870525-37-4	870525-38-5	870525-39-6
	870525-40-9	870525-41-0	870525-42-1	870525-43-2	870525-44-3
	870525-45-4				
	RL: PRP (Properties); TEM (Technical or engineered material use); USES (Uses) (high toughness aluminum alloy cast for automobile parts)				

L30 ANSWER 23 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2005:1283373 HCAPLUS
DOCUMENT NUMBER: 144:25665
TITLE: High-toughness aluminum
alloy casting and its production
method
INVENTOR(S): Toyota, Yusuke; Shibata, Katsuhiro;
Minakami, Takahiro; Murakashi, Ryoichi
PATENT ASSIGNEE(S): Honda Motor Co., Ltd., Japan
SOURCE: Jpn. Kokai Tokkyo Koho, 10 pp.
CODEN: JKXXAF
DOCUMENT TYPE: Patent
LANGUAGE: Japanese
FAMILY ACC. NUM. COUNT: 1
PATENT INFORMATION:

	PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
	-----	----	-----	-----	-----
	JP 2005336568	A	20051208	JP 2004-158757	20040528
	JP 4238180	B2	20090311		
PRIORITY APPLN. INFO.:				JP 2004-158757	20040528
AB	High-toughness Al alloy casting contains Si 2-4, Mg 0.2-0.5, Cu 0.4-0.8, Zr 0.1-0.4%, and				

balance Al and has tensile strength ≥ 80 MPa, 0.2% yield strength ≥ 220 MPa and elongation $\geq 10\%$. Al alloy having the above composition is cast, heated at 500-540°, quenched and aged at 160-185° to obtain high-toughness Al alloy cast product.

IT Casting of metals

Tensile strength

Toughness

Yield strength

(high-toughness aluminum alloy

casting and its production method)

IT 870462-24-1 870462-25-2 870462-26-3 870462-27-4 870462-28-5

870462-29-6 870462-30-9

RL: PEP (Physical, engineering or chemical process); PRP (Properties); PYP (Physical process); PROC (Process)

(high-toughness aluminum alloy

casting and its production method)

L30 ANSWER 24 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2005:586890 HCAPLUS

DOCUMENT NUMBER: 143:101204

TITLE: Cast aluminum alloys with high toughness and their manufacture

INVENTOR(S): Toyota, Yusuke; Minakami, Takahiro; Shibata, Katsuhiko

PATENT ASSIGNEE(S): Honda Motor Co., Ltd., Japan

SOURCE: Jpn. Kokai Tokkyo Koho, 15 pp.

CODEN: JKXXAF

DOCUMENT TYPE: Patent

LANGUAGE: Japanese

FAMILY ACC. NUM. COUNT: 1

PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
JP 2005177791	A	20050707	JP 2003-420405	20031218
PRIORITY APPLN. INFO.:			JP 2003-420405	20031218

AB The alloys consist of Si 2-4, Mg 0.2-0.5, Cu 0.4-0.8, Fe >0.2 and ≤ 0.5 , Ti 0.1-0.3 weight%, and balance Al and are characterized by its metallog. texture including α phase of grain size (d) $d \leq 50 \mu\text{m}$. An Al alloy melt having the said chemical compns. is cast into a mold cavity under pressurized condition and solidified under controlling its cooling rate (CR) to CR $\geq 5^\circ/\text{s}$ from the start of the solidification until its finishing. The cast alloys are suitable for bodies, parts, etc., for automobiles.

IT Casting of metals

(aluminum alloy; pressurized casting of

Al alloys followed by solidification under controlled

cooling rate for preparation of tough Al alloys)

IT Cast alloys

RL: PEP (Physical, engineering or chemical process); PYP (Physical process); TEM (Technical or engineered material use); PROC (Process); USES (Uses)

(aluminum; pressurized casting of Al

alloys followed by solidification under controlled cooling rate

for preparation of tough Al alloys)

IT Automobiles

Cooling

Solidification

(pressurized casting of Al alloys

followed by solidification under controlled cooling rate for preparation of

tough Al alloys)
 IT 856220-20-7 856220-21-8 856220-23-0 856220-26-3 856220-27-4
 856220-28-5 856220-29-6 856220-30-9 856220-31-0 856220-32-1
 RL: PEP (Physical, engineering or chemical process); PYP (Physical
 process); TEM (Technical or engineered material use); PROC (Process); USES
 (Uses)
 (pressurized casting of Al alloys
 followed by solidification under controlled cooling rate for preparation of
 tough Al alloys)

L30 ANSWER 25 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2005:449573 HCAPLUS
 DOCUMENT NUMBER: 142:467390
 TITLE: Powder-type mold releasing agents for cast metals
 INVENTOR(S): Sasaki, Hajime; Yoshida, Makoto; Hakiri, Katsutoshi;
 Gohonjo, Takashi; Fukuchi, Fumiaki; Ando,
 Katsutoshi; Shibata, Katsuhiro
 PATENT ASSIGNEE(S): Hanano Shoji Co., Ltd., Japan; Honda Motor Co., Ltd.
 SOURCE: Jpn. Kokai Tokkyo Koho, 9 pp.
 CODEN: JKXXAF
 DOCUMENT TYPE: Patent
 LANGUAGE: Japanese
 FAMILY ACC. NUM. COUNT: 1
 PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
JP 2005131673	A	20050526	JP 2003-370557	20031030
PRIORITY APPLN. INFO.:			JP 2003-370557	20031030

AB The agents comprise 30-70 weight% organic compound and/or graphite and contain
 aluminum hydroxide and/or zinc oxide of particle size 1-30 μm ,
 preferably 1-10 μm , as amphoteric compds. Easily releasable cast
 metals showing excellent over-coatability are obtained by use of the
 agents.

IT Parting materials
 (mold releasing powder; organic- and/or graphite-based releasing powder
 containing ZnO and/or Al hydroxide for preparation of cast metals
 showing excellent over-coatability)

IT Amphoteric materials
 Casting of metals
 (organic- and/or graphite-based releasing powder containing ZnO and/or
 Al hydroxide for preparation of cast metals showing excellent
 over-coatability)

IT Aluminum alloy, base
 RL: PEP (Physical, engineering or chemical process); PYP (Physical
 process); TEM (Technical or engineered material use); PROC (Process); USES
 (Uses)
 (organic- and/or graphite-based releasing powder containing ZnO and/or
 Al hydroxide for preparation of cast metals showing excellent
 over-coatability)

IT 37321-78-1, ADC 12
 RL: PEP (Physical, engineering or chemical process); PYP (Physical
 process); TEM (Technical or engineered material use); PROC (Process); USES
 (Uses)
 (organic- and/or graphite-based releasing powder containing ZnO and/or
 Al hydroxide for preparation of cast metals showing excellent
 over-coatability)

IT 1314-13-2, Zinc oxide, uses 7782-42-5, Graphite, uses 21645-51-2,
 Aluminum hydroxide, uses
 RL: TEM (Technical or engineered material use); USES (Uses)
 (organic- and/or graphite-based releasing powder containing ZnO and/or
 Al hydroxide for preparation of cast metals showing excellent

over-coatability)

L30 ANSWER 26 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2005:120012 HCAPLUS

DOCUMENT NUMBER: 142:181467

TITLE: Filler metal for welding of aluminum alloy material

INVENTOR(S): Kuriyama, Ryohei; Yamazaki, Kei; Nakano, Toshihiko

PATENT ASSIGNEE(S): Kobe Steel, Ltd., Japan

SOURCE: Jpn. Kokai Tokkyo Koho, 13 pp.

CODEN: JKXXAF

DOCUMENT TYPE: Patent

LANGUAGE: Japanese

FAMILY ACC. NUM. COUNT: 1

PATENT INFORMATION:

	PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
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	JP 2005034896	A	20050210	JP 2003-276355	20030717
PRIORITY APPLN. INFO.:				JP 2003-276355	20030717
AB	The claimed filler metal is an Al alloy containing Si ≤ 0.25 , Mn 0.50-1.00, Mg 3.00-3.50, Ti 0.02-0.50, B 0.001-0.010, Zr 0.10-0.40, Fe ≤ 0.25 , and Cu ≤ 0.10 weight%. The resulting welded Al alloy material provides low stress corrosion cracking and high-temperature cracking sensitivity.				
IT	Welding of metals (filler metal for welding of aluminum alloy material)				
IT	833465-81-9	833465-83-1	833465-85-3	833465-87-5	833465-89-7
	833465-91-1	833465-93-3	833465-95-5	833465-97-7	833465-99-9
	833466-01-6	833466-03-8	833466-05-0		
	RL: PEP (Physical, engineering or chemical process); PYP (Physical process); TEM (Technical or engineered material use); PROC (Process); USES (Uses) (filler; filler metal for welding of aluminum alloy material)				
IT	7440-42-8, Boron, uses RL: MOA (Modifier or additive use); TEM (Technical or engineered material use); USES (Uses) (microalloying element; filler metal for welding of aluminum alloy material)				

L30 ANSWER 27 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2005:159886 HCAPLUS

DOCUMENT NUMBER: 142:223719

TITLE: Cored wire electrode for the joint welding of high-strength aluminum alloys

INVENTOR(S): Bouaifi, Belkacem

PATENT ASSIGNEE(S): Germany

SOURCE: Ger. Offen., 7 pp.

CODEN: GWXXBX

DOCUMENT TYPE: Patent

LANGUAGE: German

FAMILY ACC. NUM. COUNT: 1

PATENT INFORMATION:

	PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
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	DE 10334959	A1	20050224	DE 2003-10334959	20030731
PRIORITY APPLN. INFO.:				DE 2003-10334959	20030731
AB	The invention concerns a cored wire electrode for the joint welding of building components from high-strength Al alloys with a tubular jacket, into which ≥ 1 powdered components are supplied, which are melt to the alloy by external heating of the filled jacket, whereby the jacket				

consists of preferably an Al-Mg-Mn-alloy and contains a filling of water-, or gas-atomized melts with a particle size of <0.2 mm. The Al-Mg-Mn-alloy contains Si 0.1-0.5, Fe 0.1-0.5, Cu 0.1-0.4, Mn 0.4-1.2, Mg 0.6-4.0, Cr 0.05-0.3, Zn 0.1-1.5, Ti 0.01-0.2, Zr 0.05-0.25, Sc 0-1.0 weight% and Al as balance. The cored wire electrode is especially used for the welding of high-strength Al alloys applied in the light metal construction of the automobile and aerospace industry.

IT Aerospace industry
(aviation and aeronautics; cored wire electrode for the joint welding of high-strength aluminum alloys applied for)

IT Automobiles
(cored wire electrode for the joint welding of high-strength aluminum alloys applied for)

IT Welding of metals
(flux-cored arc, electrodes; for the joint welding of high-strength aluminum alloys)

IT 841260-31-9 841260-32-0 841260-33-1
RL: CPS (Chemical process); DEV (Device component use); PEP (Physical, engineering or chemical process); PROC (Process); USES (Uses)
(cored wire electrode for the joint welding of high-strength aluminum alloys)

IT 11145-78-1, AlMg3Mn 12616-86-3, AlMg4.5Mn0.7 12720-80-8, AlMg4 37202-63-4, AlMg4.5Mn0.4
RL: CPS (Chemical process); DEV (Device component use); PEP (Physical, engineering or chemical process); PROC (Process); USES (Uses)
(filler material for cored wire electrode for the joint welding of high-strength aluminum alloys)

L30 ANSWER 28 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2005:1147772 HCAPLUS

DOCUMENT NUMBER: 144:492600

TITLE: Development of aluminum subframe using hot bulging and vacuum die casting

AUTHOR(S): Fukuchi, Fumiaki; Yahaba, Takanori; Ogawa, Tsutomu; Hori, Izuru; Akiyama, Hiroshi

CORPORATE SOURCE: Honda R+D Co., Ltd., Shimotakanezawa 4630, Haga-machi, Haga-gun, Tochigi, Japan

SOURCE: Review of Automotive Engineering (2005), 26(3), 313-318
CODEN: RAEEAH; ISSN: 1349-4724

PUBLISHER: Society of Automotive Engineers of Japan, Inc.

DOCUMENT TYPE: Journal

LANGUAGE: English

AB The world's first aluminum subframe has been developed which is composed of bulge-formed members and die cast members welded by MIG. The bulging employed a newly developed hot process with superior formability together with an alloy developed for the hot process. The die casting employed a newly developed vacuum die casting process and alloy which has superior weldability. A method of MIG welding was established, after the weldability of these new alloys was researched. The developed subframe achieved a cost reduction and 10% weight reduction while keeping functional performance equal to or better than the company's conventional aluminum subframe.

IT Automobiles
(bodies; development of aluminum automobile subframe using hot bulging and vacuum die casting)

IT Casting of metals
(die; development of aluminum automobile subframe using hot bulging and vacuum die casting)

IT Metalworking
 (forming, bulging; development of aluminum automobile subframe using hot bulging and vacuum die casting)

IT Casting process
 (vacuum; development of aluminum automobile subframe using hot bulging and vacuum die casting)

IT Aluminum alloy, base
 RL: PEP (Physical, engineering or chemical process); PYP (Physical process); TEM (Technical or engineered material use); PROC (Process); USES (Uses)
 (development of aluminum automobile subframe using hot bulging and vacuum die casting)

REFERENCE COUNT: 6 THERE ARE 6 CITED REFERENCES AVAILABLE FOR THIS RECORD. ALL CITATIONS AVAILABLE IN THE RE FORMAT

L30 ANSWER 29 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN
 ACCESSION NUMBER: 2005:440956 HCAPLUS
 DOCUMENT NUMBER: 144:354874
 TITLE: Superplastic magnalium for increased rates of superplastic forming
 AUTHOR(S): Portnoi, V. K.; Rylov, D. S.; Levchenko, V. S.; Alalykin, A. A.
 CORPORATE SOURCE: MISiS, Russia
 SOURCE: Tsvetnye Metally (Moscow, Russian Federation) (2005), (1), 84-87
 CODEN: TVMTAX; ISSN: 0372-2929
 PUBLISHER: Izdatel'skii Dom "Ruda i Metally"
 DOCUMENT TYPE: Journal
 LANGUAGE: Russian

AB Superplastic magnalium has been developed on the base of Russian Al-Mg alloy AMg6 and is directed at widening of the production range of automotive components from aluminum alloys via superplastic forming. The proposed alloy will increase the superplastic forming rate by 5-10 times, compared with that at present for alloy 5083 alloy (Russian analog AMg4). AMg6 has finer particle size than AMg4 and their dilatation curves, which have similar shape, show a relative lengthening of AMg6 which is 2-3 times greater than that of AMg4.

IT Microstructure
 (of superplastic magnalium alloy AMg6 compared to AMg4)

IT Plasticity
 (superplasticity, of Russian alloy AMg6; superplastic magnalium for increased rates of superplastic forming in manufacture of automotive components)

IT 81159-87-7, AMg4
 RL: PRP (Properties)
 (comparison of microstructure and superplasticity parameters of magnalium alloys AMg6 and AMg4)

IT 12732-16-0, AMg6
 RL: PRP (Properties)
 (superplastic magnalium for increased rates of superplastic forming in manufacture of automotive components)

L30 ANSWER 30 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN
 ACCESSION NUMBER: 2006:393724 HCAPLUS
 DOCUMENT NUMBER: 146:126286
 TITLE: XRFs determination of 10 alloying elements in superhard aluminum alloys
 AUTHOR(S): Wu, Yanqing; Xu, Hai
 CORPORATE SOURCE: Xi'an Huashan Machine Plant, Xi'an, 710043, Peop. Rep. China
 SOURCE: Lihua Jianyan, Huaxue Fence (2005), 41(1), 28-29
 CODEN: LJHFE2; ISSN: 1001-4020

PUBLISHER: Lihua Jianyan Zazhishe
DOCUMENT TYPE: Journal
LANGUAGE: Chinese

AB A rapid, accurate and precise XRF method for the determination of 10 alloying elements (Cu, Mg, Zn, Fe, Si, Mn, Cr, Ni, Ti, and Zr) in superhard Al alloys by SRS 300 XRF spectrometer was reported. The samples were lathed to have plain and smooth surfaces. A set of SRM's (prepared by the Southwest Aluminum Plant in 1989) was used for drawing of working standard curves. The simulation, regression anal., and correction for matrix effect were carried out by software and computer. The results of precision test for all these 10 elements showed that RSD's (n = 10) were <2.2%. The determination results of the 10 elements in 3 samples obtained by the method were in agreement with those obtained by conventional chemical method.

IT 7439-89-6, Iron, analysis 7439-95-4, Magnesium, analysis
7439-96-5, Manganese, analysis 7440-02-0, Nickel, analysis
7440-21-3, Silicon, analysis 7440-32-6, Titanium, analysis 7440-47-3,
Chromium, analysis 7440-66-6, Zinc, analysis 7440-67-7, Zirconium,
analysis

RL: ANT (Analyte); ANST (Analytical study)
(XRF determination of 10 alloying elements in superhard aluminum alloys)

IT 918789-28-3, Aluminum 81-98, chromium 0.1-0.4, copper
0.1-3, iron 0.1-0.7, magnesium 0.3-4, manganese
0.1-0.9, nickel 0-0.2, silicon 0.1-0.8, titanium 0-0.2, zinc 1.4-8.4,
zirconium 0-0.3

RL: NUU (Other use, unclassified); USES (Uses)
(sample; XRF determination of 10 alloying elements in superhard aluminum alloys)

L30 ANSWER 31 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2004:1125675 HCAPLUS

DOCUMENT NUMBER: 142:60742

TITLE: Aluminum-silicon-base alloy cast
products with high toughness and stress
corrosion cracking resistance and their manufacture

INVENTOR(S): Nakamura, Takeyoshi; Shibata, Katsuhiko

PATENT ASSIGNEE(S): Honda Motor Co., Ltd., Japan

SOURCE: Jpn. Kokai Tokkyo Koho, 9 pp.

CODEN: JKXXAF

DOCUMENT TYPE: Patent

LANGUAGE: Japanese

FAMILY ACC. NUM. COUNT: 1

PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
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JP 2004359988	A	20041224	JP 2003-157903	20030603
JP 4092255	B2	20080528		

PRIORITY APPLN. INFO.: JP 2003-157903 20030603

AB The cast products are obtained from semi-solid slurry of the alloy comprising Si 6.5-7.5, Cu 0.5-1.5, Mg 0.4-0.5, Ti <0.2 weight%, and balance Al and have volume ratio of solid-solidified region (SS) Vf 40-60%. The cast products are manufactured by pouring the semi-solid slurry with the above composition and solid ratio S 40-60% into a mold and cooling. The process may be thixocasting or rheocasting process. Primary crystallization of Si in liquid-solidified region and segregation are prevented. The cast products have high toughness and stress corrosion cracking resistance.

IT Cast alloys

RL: PEP (Physical, engineering or chemical process); PYP (Physical

process); TEM (Technical or engineered material use); PROC (Process); USES (Uses)

(aluminum-silicon; manufacture of Al-Si-base alloy cast products with high toughness and stress corrosion cracking resistance from semi-solid slurry)

IT Casting of metals

(rheocasting; manufacture of Al-Si-base alloy cast products with high toughness and stress corrosion cracking resistance from semi-solid slurry)

IT Casting of metals

(thixocasting; manufacture of Al-Si-base alloy cast products with high toughness and stress corrosion cracking resistance from semi-solid slurry)

IT 809275-84-1 809275-85-2 809275-87-4 809275-88-5

RL: PEP (Physical, engineering or chemical process); PYP (Physical process); TEM (Technical or engineered material use); PROC (Process); USES (Uses)

(manufacture of Al-Si-base alloy cast products with high toughness and stress corrosion cracking resistance from semi-solid slurry)

L30 ANSWER 32 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2004:1058568 HCAPLUS

DOCUMENT NUMBER: 142:42199

TITLE: Manufacture of Al-Si alloy cast having high toughness and stress corrosion cracking resistance

INVENTOR(S): Nakamura, Takeyoshi; Shibata, Katsuhiko; Minakami, Takahiro

PATENT ASSIGNEE(S): Honda Motor Co., Ltd., Japan

SOURCE: Jpn. Kokai Tokkyo Koho, 11 pp.
CODEN: JKXXAF

DOCUMENT TYPE: Patent

LANGUAGE: Japanese

FAMILY ACC. NUM. COUNT: 1

PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
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JP 2004346408	A	20041209	JP 2003-147850	20030526
PRIORITY APPLN. INFO.:			JP 2003-147850	20030526

AB The alloy comprises Si 5.5-8, Fe \leq 0.1, Mg 0.4-0.5, Ti \leq 0.2%, and Al bal. In the solid-liquid coexisting casting process of the alloy, the volume ratio of the solid phase solidification area (SS) is $30\% \leq V_f \leq 60\%$; and the Fe content of the liquid phase solidification area (SL) is Fe \leq 0.2 weight%. The alloy cast is manufactured by preparing the solid-liquid coexisting material with solid phase ratio being $30\% \leq S \leq 60\%$, pouring into the mold, and cooling.

IT Casting of metals

Impact strength
(manufacture of Al-Si alloy cast having high toughness and stress corrosion cracking resistance)

IT Stress corrosion cracking

(resistance; manufacture of Al-Si alloy cast having high toughness and stress corrosion cracking resistance)

IT 804566-20-9P 804566-22-1P 804566-25-4P

RL: IMF (Industrial manufacture); PEP (Physical, engineering or chemical process); PYP (Physical process); PREP (Preparation); PROC (Process)

(manufacture of Al-Si alloy cast having high toughness and stress corrosion cracking resistance)

L30 ANSWER 33 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2004:992914 HCAPLUS
DOCUMENT NUMBER: 141:414242
TITLE: Light-weight cast Al alloy machine
structural part
INVENTOR(S): Matsumoto, Yoichi; Shibata, Katsuhiko;
Nakamura, Takeyoshi
PATENT ASSIGNEE(S): Honda Motor Co., Ltd., Japan
SOURCE: Jpn. Kokai Tokkyo Koho, 11 pp.
CODEN: JKXXAF
DOCUMENT TYPE: Patent
LANGUAGE: Japanese
FAMILY ACC. NUM. COUNT: 1
PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
JP 2004322103	A	20041118	JP 2003-115779	20030421
JP 4116921	B2	20080709		

PRIORITY APPLN. INFO.: JP 2003-115779 20030421

AB The part is formed by casting a material with hypoeutectic Al-Si alloy composition while containing liquid phase and solid phase and has a main area integrated with a lightwt. area with decreased volume, wherein the Si content in the lightwt. area is higher than that in the main area. The lightwt. area has improved strength owing to increased Si content.

IT Cast alloys
RL: TEM (Technical or engineered material use); USES (Uses)
(aluminum; lightwt. cast Al-Si alloy
machine structural part having lightwt. area with high Si content)

IT Machinery parts
(lightwt. cast Al-Si alloy machine structural part
having lightwt. area with high Si content)

IT 11099-22-2 791616-38-1, Aluminum 92, copper 0.1, iron 0.1, magnesium 0.4, silicon 7.4, titanium 0.1
RL: PEP (Physical, engineering or chemical process); PYP (Physical process); TEM (Technical or engineered material use); PROC (Process); USES (Uses)
(hypoeutectic; lightwt. cast Al-Si alloy machine structural part having lightwt. area with high Si content)

L30 ANSWER 34 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2004:739087 HCAPLUS
DOCUMENT NUMBER: 141:246965
TITLE: Aluminum-magnesium alloy sheets
having high strength and deep drawability
INVENTOR(S): Kajiwara, Katsura; Matsumoto, Kazuhide
PATENT ASSIGNEE(S): Kobe Steel, Ltd., Japan
SOURCE: Jpn. Kokai Tokkyo Koho, 17 pp.
CODEN: JKXXAF
DOCUMENT TYPE: Patent
LANGUAGE: Japanese
FAMILY ACC. NUM. COUNT: 1
PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
JP 2004250738	A	20040909	JP 2003-41141	20030219
			JP 2003-41141	20030219

PRIORITY APPLN. INFO.:
AB Al-Mg alloys containing 2-6 weight% Mg and having textures characterized by having Cube orientation 10-30%, S orientation 30-50%, both Cu orientation and Brass orientation 5-20%, Goss orientation

≤10%, and (Cube + Goss)/(S + Cu + Brass) = 0.1-0.5. Preferably, the alloys also contain (A) Mn ≤1.0, (B) Cu ≤0.6, (C) Fe ≤0.7, Si ≤0.5, Cr ≤0.4, Zn ≤0.5, and/or Zr ≤0.3, and/or (D) 0.005-0.20 weight% Ti and optionally 0.0001-0.05 weight% B. The sheets have decreased ear ratio.

IT Texture (metallographic)
 (Al-Mg alloy sheets with certain texture orientation with high strength and deep drawability)

IT Drawing (forming)
 (deep; Al-Mg alloy sheets with certain texture orientation with high strength and deep drawability)

IT 12686-54-3 126744-93-2 749250-11-1 749250-12-2 749250-13-3
 749250-14-4 749250-15-5
 RL: TEM (Technical or engineered material use); USES (Uses)
 (Al-Mg alloy sheets with certain texture orientation with high strength and deep drawability)

IT 7440-42-8, Boron, uses
 RL: MOA (Modifier or additive use); TEM (Technical or engineered material use); USES (Uses)
 (microalloying element; Al-Mg alloy sheets with certain texture orientation with high strength and deep drawability)

L30 ANSWER 35 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2004:632102 HCAPLUS

DOCUMENT NUMBER: 141:160824

TITLE: Manufacture of closed section structure filled with foamed body

INVENTOR(S): Ishikawa, Ryoichi; Shibata, Katsuhiko; Hayakawa, Kimito

PATENT ASSIGNEE(S): Honda Motor Co., Ltd., Japan

SOURCE: Jpn. Kokai Tokkyo Koho, 9 pp.
 CODEN: JKXXAF

DOCUMENT TYPE: Patent

LANGUAGE: Japanese

FAMILY ACC. NUM. COUNT: 1

PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
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JP 2004218035	A	20040805	JP 2003-9242	20030117
JP 4233018	B2	20090304		
US 20040191107	A1	20040930	US 2004-758283	20040116
US 7141206	B2	20061128		

PRIORITY APPLN. INFO.: JP 2003-9242 A 20030117

AB The process comprises preparing metal powder, foaming agent, and metal plate, mixing the metal powder with the foaming agent, forming the mixture into planar shape, stretching and attaching the formed foaming agent mixture onto one side of the metal plate, wrapping the foaming agent mixture with the metal plate, plastic-deforming to obtain the closed section structure, and heating at the foaming temperature The obtained structure is suitable for automobile body.

IT Automobiles
 (bodies; manufacture of closed section structure filled with foamed body for)

IT Foaming agents
 (for manufacture of closed section structure filled with foamed body)

IT Cellular materials
 (manufacture of closed section structure filled with foamed body)

IT Copper alloy, base
 Magnesium alloy, base
 Zinc alloy, base

RL: PEP (Physical, engineering or chemical process); PYP (Physical

process); TEM (Technical or engineered material use); PROC (Process); USES (Uses)

(powder and plate, raw material; for manufacture of closed section structure filled with foamed body)

IT 124-38-9, Carbon dioxide, processes 13776-99-3, Titanium hydride (TiH)
RL: NUU (Other use, unclassified); PEP (Physical, engineering or chemical process); PYP (Physical process); PROC (Process); USES (Uses)

(foaming agent; for manufacture of closed section structure filled with foamed body)

IT 7429-90-5, Aluminum, processes 11099-22-2

RL: PEP (Physical, engineering or chemical process); PYP (Physical process); TEM (Technical or engineered material use); PROC (Process); USES (Uses)

(powder and plate, raw material; for manufacture of closed section structure filled with foamed body)

L30 ANSWER 36 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2004:427660 HCAPLUS

DOCUMENT NUMBER: 140:427270

TITLE: Stabilized carbonate powder as melt-foaming agent for manufacture of porous metal

INVENTOR(S): Ishikawa, Ryoichi; Shibata, Katsuhiko; Nakamura, Takashi

PATENT ASSIGNEE(S): Honda Motor Co., Ltd., Japan

SOURCE: Eur. Pat. Appl., 12 pp.

CODEN: EPXXDW

DOCUMENT TYPE: Patent

LANGUAGE: English

FAMILY ACC. NUM. COUNT: 1

PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
EP 1422302	A1	20040526	EP 2003-26197	20031117
EP 1422302	B1	20090318		
R: AT, BE, CH, DE, DK, ES, FR, GB, GR, IT, LI, LU, NL, SE, MC, PT, IE, SI, LT, LV, FI, RO, MK, CY, AL, TR, BG, CZ, EE, HU, SK				
JP 2004183095	A	20040702	JP 2003-358447	20031017
JP 3986489	B2	20071003		
US 20040126583	A1	20040701	US 2003-698015	20031031
US 20060173082	A1	20060803	US 2006-393865	20060331
US 7410523	B2	20080812		

PRIORITY APPLN. INFO.: JP 2002-335622 A 20021119
US 2003-698015 A3 20031031

AB The foaming agent for melt treatment in manufacture of a foamed or porous metal is CaCO₃ or MgCO₃ powder precoated with SiO₂ or silicate film for stability. The powder suitable for foaming of molten Al-7% Si alloy is prepared from CaCO₃ powder by copptn. coating in aqueous slurry with 2% Na₂SiO₃ at the pH of 6.87, followed by drying the coated powder at 100°.

IT Metals, uses

RL: TEM (Technical or engineered material use); USES (Uses)
(foamed, casting of; carbonate powder as melt-foaming agent for manufacture of porous metal or alloy)

IT Casting of metals

(melt foaming in; carbonate powder as melt-foaming agent for manufacture of porous metal or alloy)

IT Aluminum alloy, base

RL: EPR (Engineering process); PEP (Physical, engineering or chemical process); PROC (Process)

(molten, foaming of; carbonate powder as melt-foaming agent for manufacture of porous metal or alloy)

IT 1344-09-8, Water glass 7631-86-9, Silica, uses
 RL: TEM (Technical or engineered material use); USES (Uses)
 (film, carbonate powder for melt foaming with; carbonate powder as
 melt-foaming agent for manufacture of porous metal or alloy)

IT 12635-40-4
 RL: EPR (Engineering process); PEP (Physical, engineering or chemical
 process); PROC (Process)
 (molten, foaming of; carbonate powder as melt-foaming agent for manufacture
 of porous metal or alloy)

IT 471-34-1, Calcium carbonate, uses 546-93-0, Magnesium
 carbonate
 RL: TEM (Technical or engineered material use); USES (Uses)
 (powder, melt foaming with; carbonate powder as melt-foaming agent for
 manufacture of porous metal or alloy)

REFERENCE COUNT: 2 THERE ARE 2 CITED REFERENCES AVAILABLE FOR THIS
 RECORD. ALL CITATIONS AVAILABLE IN THE RE FORMAT

L30 ANSWER 37 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN
 ACCESSION NUMBER: 2004:249244 HCAPLUS
 DOCUMENT NUMBER: 140:274470
 TITLE: Heating/pressurized medium cooling process for cast
 aluminum alloy parts with improved
 surface quality, porosity and impact strength
 INVENTOR(S): Nakamura, Takeyoshi; Shibata, Katsuhiko
 PATENT ASSIGNEE(S): Honda Giken Kogyo K. K., Japan
 SOURCE: Ger. Offen., 7 pp.
 CODEN: GWXXBX
 DOCUMENT TYPE: Patent
 LANGUAGE: German
 FAMILY ACC. NUM. COUNT: 1
 PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
DE 10341575	A1	20040325	DE 2003-10341575	20030909
JP 2004099962	A	20040402	JP 2002-262890	20020909
US 20040103964	A1	20040603	US 2003-648831	20030827
PRIORITY APPLN. INFO.:			JP 2002-262890	A 20020909

AB Heat treatment of cast light metal alloys, especially cast Al
 alloys, involves (1) heating up to a range of a solid solution
 (preferably above the solidus temperature), (2) holding at such temperature,
 and (3)
 cooling by using a cooling medium (e.g., water) under pressure of
 200-2,000 bar. The procedure suppresses a porosity increase and prevents
 formation of blisters on the surface of the cast alloy parts.
 Strength of the castings is increased.

IT Cast alloys
 RL: PEP (Physical, engineering or chemical process); PRP (Properties); PYP
 (Physical process); PROC (Process)
 (aluminum; heat treatment of cast aluminum
 alloy parts under pressure for decreased porosity and surface
 blisters and increased strength)

IT Heat treatment
 Impact strength
 Porosity
 (heating/pressurized medium cooling process for cast aluminum
 alloy parts with improved surface quality, porosity and impact
 strength)

IT Cooling
 (under pressurized medium; heating/pressurized medium cooling process
 for cast aluminum alloy parts with improved surface
 quality, porosity and impact strength)

IT 12773-40-9, A356
RL: PEP (Physical, engineering or chemical process); PRP (Properties); PYP (Physical process); PROC (Process)
(heat treatment of cast aluminum alloy parts under pressure for decreased porosity and surface blisters and increased strength)

L30 ANSWER 38 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2004:377014 HCAPLUS
DOCUMENT NUMBER: 141:177674
TITLE: Characteristics of structural changes during superplastic deformation of alloy AMg4
AUTHOR(S): Pozdnyakova, A. V.; Portnoi, V. K.
CORPORATE SOURCE: Kafedra Metalloved. Tsvetn. Metallov, Mosk. Gos. Inst. Stali i Splavov (Tekhnol. Univ., Moscow, Russia
SOURCE: Izvestiya Vysshikh Uchebnykh Zavedenii, Tsvetnaya Metallurgiya (2004), (1), 53-56
CODEN: IVUTAK; ISSN: 0021-3438
PUBLISHER: Moskovskii Gosudarstvennyi Institut Stali i Splavov
DOCUMENT TYPE: Journal
LANGUAGE: Russian

AB Quant. metallog. was used to study changes of the structure (grain shape and size in longitudinal and transverse thickness sections) in the process of superplastic deformation of AMg4 alloy. The intermittent changes both in the sheet plane and thickness sections were observed. The intermittent elongation of the grains and recovery of their equiaxial shape confirmed the dynamic recrystn. phenomena during the superplastic deformation, i.e. this was a process occurring in the specimen bulk rather than a characteristic of the surface layer. In both thickness sections, division of the grains along the stretching axis was observed, which was a direct indication of the dynamic recrystn.

IT Grain size
(characteristics of structural changes during superplastic deformation of alloy AMg4)

IT Recrystallization
(dynamic; structural changes during superplastic deformation of alloy AMg4 in relation to)

IT Stress, mechanical
(flow; characteristics of structural changes during superplastic deformation of alloy AMg4 in relation to)

IT Microstructure
(grain size and shape; characteristics of structural changes during superplastic deformation of alloy AMg4)

IT Plastic deformation
(superplastic; characteristics of structural changes during superplastic deformation of alloy AMg4)

IT 81159-87-7, AMg4
RL: PEP (Physical, engineering or chemical process); PRP (Properties); PYP (Physical process); PROC (Process)
(characteristics of structural changes during superplastic deformation of alloy AMg4)

L30 ANSWER 39 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2004:504380 HCAPLUS
DOCUMENT NUMBER: 142:301548
TITLE: Methods for decreasing the content of harmful impurities in recycling of aluminum wastes
AUTHOR(S): Kalenik, O. N.; Nemenok, B. M.; Tribushevskii, V. L.; Dovnar, G. V.; Sitnichenko, M. M.
CORPORATE SOURCE: BNTU, Minsk, Belarus
SOURCE: Metallurgiya Mashinostroeniya (2004), (2), 11-13
CODEN: MMEAC2

PUBLISHER: OOO "Liteinoe Proizvodstvo"
DOCUMENT TYPE: Journal
LANGUAGE: Russian

AB The main impurities that impair the quality of Al alloy scrap are Fe, Mg, and Zn, and the removal of these impurities is considered in industrial recycling. The manufacture of secondary Al alloys from low-grade Al scrap and wastes is improved by decreasing these impurities to acceptable concns. The AMg4 scrap is suitable for replacing Mg ingots in the manufacture of Al-Mg alloys with decreased vaporization loss of Mg.

IT Recycling
(of aluminum alloy, scrap processing for; methods for decreasing metal impurities in recycling of aluminum alloy scrap and waste)

IT 7439-89-6, Iron, processes 7439-95-4, Magnesium, processes
RL: REM (Removal or disposal); PROC (Process)
(impurity in aluminum alloy scrap; methods for decreasing metal impurities in recycling of aluminum alloy scrap and waste)

IT 7440-66-6, Zinc, processes
RL: REM (Removal or disposal); PROC (Process)
(impurity in aluminum alloy scrap; methods for decreasing metal impurities in recycling of aluminum alloy scrap and waste)

IT 7429-90-5, Aluminum, uses 81159-87-7, AMg4
RL: TEM (Technical or engineered material use); USES (Uses)
(scrap, recycling of; methods for decreasing metal impurities in recycling of aluminum alloy scrap and waste)

L30 ANSWER 40 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2003:972274 HCAPLUS

DOCUMENT NUMBER: 140:7577

TITLE: Die casting having high toughness

INVENTOR(S): Toyoda, Yusuke; Mizukami, Takahiro
; Fukuchi, Fumiaki; Hata, Tsunehisa
; Shibata, Katsuhiko

PATENT ASSIGNEE(S): Honda Giken Kogyo Kabushiki Kaisha, Japan

SOURCE: PCT Int. Appl., 19 pp.

CODEN: PIXXD2

DOCUMENT TYPE: Patent

LANGUAGE: Japanese

FAMILY ACC. NUM. COUNT: 2

PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
-----	---	-----	-----	-----
WO 2003102257	A1	20031211	WO 2003-JP5993	20030514
W: AE, AG, AL, AM, AT, AU, AZ, BA, BB, BG, BR, BY, BZ, CA, CH, CN, CO, CR, CU, CZ, DE, DK, DM, DZ, EC, EE, ES, FI, GB, GD, GE, GH, GM, HR, HU, ID, IL, IN, IS, KE, KG, KP, KR, KZ, LC, LK, LR, LS, LT, LU, LV, MA, MD, MG, MK, MN, MW, MX, MZ, NI, NO, NZ, OM, PH, PL, PT, RO, RU, SC, SD, SE, SG, SK, SL, TJ, TM, TN, TR, TT, TZ, UA, UG, US, UZ, VC, VN, YU, ZA, ZM, ZW				
RW: GH, GM, KE, LS, MW, MZ, SD, SL, SZ, TZ, UG, ZM, ZW, AM, AZ, BY, KG, KZ, MD, RU, TJ, TM, AT, BE, BG, CH, CY, CZ, DE, DK, EE, ES, FI, FR, GB, GR, HU, IE, IT, LU, MC, NL, PT, RO, SE, SI, SK, TR, BF, BJ, CF, CG, CI, CM, GA, GN, GQ, GW, ML, MR, NE, SN, TD, TG				
JP 2003342664	A	20031203	JP 2002-157329	20020530
JP 4092138	B2	20080528		
JP 2004001010	A	20040108	JP 2002-157328	20020530
JP 4210473	B2	20090121		

AU 2003235302 A1 20031219 AU 2003-235302 20030514
 EP 1508627 A1 20050223 EP 2003-723374 20030514
 R: AT, BE, CH, DE, DK, ES, FR, GB, GR, IT, LI, LU, NL, SE, MC, PT,
 IE, SI, LT, LV, FI, RO, MK, CY, AL, TR, BG, CZ, EE, HU, SK
 US 20060137848 A1 20060629 US 2005-518151 20050927
 PRIORITY APPLN. INFO.: JP 2002-157328 A 20020530
 JP 2002-157329 A 20020530
 WO 2003-JP5993 W 20030514

AB A high-toughness die casting comprises an
 Al-Mg based alloy having a chemical composition, in
 weight %: $3.5 \leq \text{Mg} \leq 4.5$, $0.8 \leq \text{Mn}$
 ≤ 1.5 , $\text{Si} < 0.5$, $\text{Fe} < 0.5$, $\text{Ti} + \text{Zr} \geq 0.3$, $0.3 \leq \text{Ti/Zr}$
 ≤ 2 and balance of Al. The die casting
 exhibits high toughness and can be suitably used as a thin and
 large die casting.

IT Casting of metals

Toughness

(die casting having high toughness of
 aluminum-magnesium alloy)

IT 116658-27-6 627892-55-1 627892-56-2 627892-57-3
 627892-58-4 627892-59-5 627892-60-8
 627892-61-9 627892-62-0

RL: PEP (Physical, engineering or chemical process); PRP (Properties); PYP
 (Physical process); PROC (Process)

(die casting having high toughness of
 aluminum-magnesium alloy)

REFERENCE COUNT: 5 THERE ARE 5 CITED REFERENCES AVAILABLE FOR THIS
 RECORD. ALL CITATIONS AVAILABLE IN THE RE FORMAT

L30 ANSWER 41 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2003:133508 HCAPLUS

DOCUMENT NUMBER: 138:174180

TITLE: Aluminum-magnesium alloys for
 weldable high-strength strip resistant to corrosion

INVENTOR(S): Van Der Hoeven, Job Anthonius; Zuang, Linzhong;
 Schepers, Bruno

PATENT ASSIGNEE(S): Corus Aluminium N.V., Belg.; Corus Aluminium
 Walzprodukte Gmbh

SOURCE: PCT Int. Appl., 18 pp.
 CODEN: PIXXD2

DOCUMENT TYPE: Patent

LANGUAGE: English

FAMILY ACC. NUM. COUNT: 1

PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
WO 2003014405	A1	20030220	WO 2002-EP8627	20020731
W:			AE, AG, AL, AM, AT, AU, AZ, BA, BB, BG, BR, BY, BZ, CA, CH, CN, CO, CR, CU, CZ, DE, DK, DM, DZ, EC, EE, ES, FI, GB, GD, GE, GH, GM, HR, HU, ID, IL, IN, IS, JP, KE, KG, KP, KR, KZ, LC, LK, LR, LS, LT, LU, LV, MA, MD, MG, MK, MN, MW, MX, MZ, NO, NZ, OM, PH, PL, PT, RO, RU, SD, SE, SG, SI, SK, SL, TJ, TM, TN, TR, TT, TZ, UA, UG, US, UZ, VN, YU, ZA, ZM, ZW	
RW:			GH, GM, KE, LS, MW, MZ, SD, SL, SZ, TZ, UG, ZM, ZW, AM, AZ, BY, KG, KZ, MD, RU, TJ, TM, AT, BE, BG, CH, CY, CZ, DE, DK, EE, ES, FI, FR, GB, GR, IE, IT, LU, MC, NL, PT, SE, SK, TR, BF, BJ, CF, CG, CI, CM, GA, GN, GQ, GW, ML, MR, NE, SN, TD, TG	
DE 10231437	A1	20030227	DE 2002-10231437	20020711
GR 2002100348	A	20030328	GR 2002-100348	20020726
GR 1004282	B2	20030626		
AU 2002331383	A1	20030224	AU 2002-331383	20020731

AU 2002331383	B2	20071213		
EP 1461465	A1	20040929	EP 2002-767307	20020731
R: AT, BE, CH, DK, ES, GB, LI, LU, NL, SE, MC, PT, IE, SI, LT, LV, FI, RO, MK, CY, AL, TR, BG, CZ, EE, SK				
FR 2828498	A1	20030214	FR 2002-10077	20020808
FR 2828498	B1	20050902		
US 20040261922	A1	20041230	US 2004-486112	20040827
US 20070187009	A1	20070816	US 2007-740230	20070425
PRIORITY APPLN. INFO.:			EP 2001-203034	A 20010810
			EP 2002-75049	A 20020103
			EP 2002-77548	A 20020627
			WO 2002-EP8627	W 20020731
			US 2004-486112	A3 20040827

AB The corrosion-resistant Al alloy for weldable sheet or strip contains Mg 3.1-4.5, Mn 0.4-0.85, Zn 0.4-0.8, Cu 0.06-0.35, Cr <0.25, Fe <0.35, Si <0.2, Zr <0.25, Ti <0.3, and impurities at ≤0.05 each with total of ≤0.15%. The Al-alloy strip is preferably manufactured by cold rolling to the final thickness, followed by annealing with rapid heating at 2-200°/s, holding for ≤100 s at 480-570°, and cooling at 10-500°/s to below 150°. The Al-alloy strip is preferably 1.6-2.4 mm thick, and is suitable for welded pressure vessels resistant to intergranular corrosion. The Al alloy retains tensile yield strength ≥120 MPa after 1000-h holding at 100°. The typical Al alloy contains Mg 4.29, Mn 0.50, Zn 0.54, Cu 0.085, Cr 0.14, Fe 0.14, Si 0.04, Zr 0.001, and Ti 0.02%.

IT Welds
(Al-alloy; aluminum-magnesium alloys for weldable high-strength strip resistant to corrosion)

IT Pressure vessels
(Al-alloy; aluminum-magnesium alloys for welded pressure vessels resistant to corrosion)

IT 497821-82-6
RL: TEM (Technical or engineered material use); USES (Uses)
(alloying of; aluminum-magnesium alloys for weldable high-strength strip resistant to corrosion)

IT 497821-83-7 497821-84-8 497821-85-9 497821-86-0
RL: TEM (Technical or engineered material use); USES (Uses)
(microalloyed; aluminum-magnesium alloys for weldable high-strength strip resistant to corrosion)

REFERENCE COUNT: 6 THERE ARE 6 CITED REFERENCES AVAILABLE FOR THIS RECORD. ALL CITATIONS AVAILABLE IN THE RE FORMAT

L30 ANSWER 42 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2003:97562 HCAPLUS

DOCUMENT NUMBER: 138:157325

TITLE: Aluminum alloy excellent in machinability, and aluminum alloy material and method for production thereof

INVENTOR(S): Matsuoka, Hideaki; Yamanaka, Masaki; Yoshioka, Hiroki; Okamoto, Yasuo; Kitamura, Masakatsu

PATENT ASSIGNEE(S): Showa Denko K. K., Japan

SOURCE: PCT Int. Appl., '79 pp.

CODEN: PIXXD2

DOCUMENT TYPE: Patent

LANGUAGE: Japanese

FAMILY ACC. NUM. COUNT: 1

PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
-----	----	-----	-----	-----
WO 2003010349	A1	20030206	WO 2002-JP7517	20020725

W: AE, AG, AL, AM, AT, AU, AZ, BA, BB, BG, BR, BY, BZ, CA, CH, CN, CO, CR, CU, CZ, DE, DK, DM, DZ, EC, EE, ES, FI, GB, GD, GE, GH, GM, HR, HU, ID, IL, IN, IS, JP, KE, KG, KP, KR, KZ, LC, LK, LR, LS, LT, LU, LV, MA, MD, MG, MK, MN, MW, MX, MZ, NO, NZ, OM, PH, PL, PT, RO, RU, SD, SE, SG, SI, SK, SL, TJ, TM, TN, TR, TT, TZ, UA, UG, UZ, VN, YU, ZA, ZM, ZW

RW: GH, GM, KE, LS, MW, MZ, SD, SL, SZ, TZ, UG, ZM, ZW, AT, BE, BG, CH, CY, CZ, DE, DK, EE, ES, FI, FR, GB, GR, IE, IT, LU, MC, NL, PT, SE, SK, TR, BF, BJ, CF, CG, CI, CM, GA, GN, GQ, GW, ML, MR, NE, SN, TD, TG

CA 2454509	A1	20030206	CA 2002-2454509	20020725
AU 2002323939	A1	20030217	AU 2002-323939	20020725
US 20030143102	A1	20030731	US 2002-202669	20020725
EP 1413636	A1	20040428	EP 2002-755647	20020725
EP 1413636	B1	20090128		

R: AT, BE, CH, DE, DK, ES, FR, GB, GR, IT, LI, LU, NL, SE, MC, PT, IE, SI, LT, LV, FI, RO, MK, CY, AL, TR, BG, CZ, EE, SK

CN 1555423	A	20041215	CN 2002-818303	20020725
AT 422000	T	20090215	AT 2002-755647	20020725
JP 4227014	B2	20090218	JP 2003-515695	20020725
US 20060027291	A1	20060209	US 2005-236523	20050928
JP 2009024265	A	20090205	JP 2008-259961	20081006

PRIORITY APPLN. INFO.:

JP 2001-224661	A	20010725
US 2001-311363P	P	20010813
JP 2002-148340	A	20020522
JP 2003-515695	A3	20020725
US 2002-202669	A3	20020725
WO 2002-JP7517	W	20020725

AB A 1st Al alloy containing Mg 0.3-6, Si 0.3-10, Zn 0.05-1, and Sr 0.001-0.3%. A 2nd Al alloy contains the elements contained in the 1st alloy in amts. described above and further ≥ 1 from Cu, Fe, Mn, Cr, Zr, Ti, Na and Ca. A 3rd Al alloy containing Mg 0.1-6, Si 0.3-12.5, Cu ≥ 0.01 and < 1 , Zn 0.01-3, and Sr 0.001-0.5%. A 4th Al alloy contains the elements contained in the 1st alloy in amts. described above and further ≥ 1 from Ti, B, C, Fe, Cr, Mn, Zr, V, Sc, Ni, Na, Sb, Ca, Sn, Bi, and In. The alloys are manufactured by casting a billet at 10-180 mm/min, homogenizing by holding for ≥ 6 h at 400-570°, extruding at a billet temperature of 300-550°, an extrusion rate of 0.5-100 m/min, and extrusion ratio of 10-200, solution treating for ≥ 1 h at 400-570°, and then aging for 1-30 h at 90-300°. The alloys are excellent in machinability.

IT Extrusion of metals

(aluminum alloy excellent in machinability, and aluminum alloy material and method for production thereof)

IT 494836-73-6	494836-74-7	494836-75-8	494836-76-9	494836-77-0
494836-78-1	494836-79-2	494836-80-5	494836-81-6	494836-82-7
494836-83-8	494836-84-9	494836-85-0	494836-86-1	494836-87-2
494836-88-3	494836-89-4	494836-90-7	494836-91-8	494836-92-9
494836-93-0	494836-94-1	494836-95-2	494836-96-3	494836-97-4
494836-98-5	494836-99-6	494837-00-2	494837-01-3	494837-02-4
494837-03-5	494837-04-6	494837-05-7	494837-06-8	494837-07-9
494837-08-0	494837-09-1	494837-10-4	494837-11-5	494837-12-6
494837-13-7	494837-14-8	494837-15-9	494837-16-0	494837-17-1
494837-18-2	494837-19-3	494837-20-6	494837-21-7	494837-22-8
494837-23-9	494837-24-0	494837-25-1	494837-26-2	494837-27-3
494837-28-4	494837-29-5	494837-30-8	494837-31-9	494837-32-0
494837-33-1	494837-34-2	494837-35-3	494837-36-4	494837-37-5
494837-38-6	494837-39-7	494837-40-0	494837-41-1	494837-42-2
494837-43-3	494837-44-4	494837-45-5	494837-46-6	494837-47-7
494837-48-8	494837-49-9	494837-50-2	494837-51-3	494837-52-4
494837-53-5	494837-54-6	494837-55-7	494837-56-8	494837-57-9

RL: PEP (Physical, engineering or chemical process); PRP (Properties); PYP (Physical process); TEM (Technical or engineered material use); PROC (Process); USES (Uses)

REFERENCE COUNT: 8 THERE ARE 8 CITED REFERENCES AVAILABLE FOR THIS
RECORD. ALL CITATIONS AVAILABLE IN THE RE FORMAT

PATENT INFORMATION:

IT 1309-48-4, Magnesia, uses 1344-28-1, Alumina, uses

RL: TEM (Technical or engineered material use); USES (Uses)
 (lining, pores with, in metal matrix; foamed Al or Mg
 matrix manufactured with pores lined with oxide and filled with CO2 gas)
 IT 471-34-1, Calcium carbonate, processes 546-93-0, Magnesium
 carbonate
 RL: PEP (Physical, engineering or chemical process); PYP (Physical
 process); PROC (Process)
 (pore formation by, in molten metal matrix; foamed Al or
 Mg matrix manufactured with pores lined with oxide and filled with
 CO2 gas)
 IT 124-38-9, Carbon dioxide, uses
 RL: TEM (Technical or engineered material use); USES (Uses)
 (pores with, in metal matrix; foamed Al or Mg
 matrix manufactured with pores lined with oxide and filled with CO2 gas)
 REFERENCE COUNT: 13 THERE ARE 13 CITED REFERENCES AVAILABLE FOR THIS
 RECORD. ALL CITATIONS AVAILABLE IN THE RE FORMAT

L30 ANSWER 44 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN
 ACCESSION NUMBER: 2003:118376 HCAPLUS
 DOCUMENT NUMBER: 138:174176
 TITLE: Welded rolled or extruded construction products made
 of Al alloy with improved mechanical strength
 INVENTOR(S): Raynaud, Guy-Michel; Hoffmann, Jean-Luc; Cottignies,
 Laurent; Pillet, Georges
 PATENT ASSIGNEE(S): Fr.
 SOURCE: U.S. Pat. Appl. Publ., 4 pp., Cont.-in-part of U.S.
 6,444,059.
 CODEN: USXXCO
 DOCUMENT TYPE: Patent
 LANGUAGE: English
 FAMILY ACC. NUM. COUNT: 2
 PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
US 20030031580	A1	20030213	US 2002-189176	20020705
FR 2731018	A1	19960830	FR 1995-2387	19950224
FR 2731018	B1	19970404		
FR 2731019	A1	19960830	FR 1995-12065	19951009
FR 2731019	B1	19970822		
WO 9626299	A1	19960829	WO 1996-FR279	19960221
W: AU, CA, CN, FI, JP, KR, NO, NZ, PL, RU, SG, TR, UA, US				
RW: AT, BE, CH, DE, DK, ES, FR, GB, GR, IE, IT, LU, MC, NL, PT, SE				
EP 909828	A2	19990421	EP 1998-123582	19960221
EP 909828	A3	19990616		
R: CH, DE, DK, ES, FR, GB, IT, LI, NL, SE				
US 20010050118	A1	20011213	US 1997-875113	19970725
US 6444059	B2	20020903		
AU 2003201308	A1	20030612	AU 2003-201308	20030317
US 20050183795	A1	20050825	US 2004-856793	20040601
US 6969432	B2	20051129		

PRIORITY APPLN. INFO.:
 FR 1995-2387 A 19950224
 FR 1995-12065 A 19951009
 WO 1996-FR279 W 19960221
 US 1997-875113 A2 19970725
 EP 1996-904891 A3 19960221
 AU 2001-15034 A3 20010117
 US 2002-189176 B1 20020705

AB Rolled or extruded products for welded constructions made of Al-Mg
 -Mn Al-based alloy. These products contain Mg
 3.0-5.0, Mn 0.75-1.0, Fe 0-0.25, Si 0-0.25, Zn 0.02-0.40,
 optionally one or more of the elements Cr, Cu, Ti, Zr such that Cr <0.25,

Cu <0.20, Ti <0.20, Zr <0.20, other elements <0.05 each and <0.15 in total, where Mn + 2Zn >0.75. In the welded state, these products have improved mech. strength and resistance to fatigue without unfavorable consequences with regard to toughness and corrosion resistance, and are particularly suitable for naval construction, for industrial vehicles and for bicycle frames made of welded tubes.

IT Corrosion
 (resistance; welded rolled or extruded construction products made of Al alloy with improved mech. strength and fatigue resistance)

IT Alloying
 Bicycles
 Fatigue, mechanical
 Pipes and Tubes
 Ships
 Toughness
 Welding of metals
 (welded rolled or extruded construction products made of Al alloy with improved mech. strength and fatigue resistance)

IT 182216-69-9 182216-70-2 182216-71-3 182216-73-5 182216-75-7
 182216-76-8 496921-48-3 496921-49-4 496921-50-7
 496921-51-8
 RL: PRP (Properties); TEM (Technical or engineered material use); USES (Uses)
 (welded rolled or extruded construction products made of Al alloy with improved mech. strength and fatigue resistance)

L30 ANSWER 45 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2003:945616 HCAPLUS

DOCUMENT NUMBER: 140:7579

TITLE: Aluminum-magnesium-based alloys for casting

INVENTOR(S): Toyota, Yusuke; Minakami, Takahiro; Fukuchi, Fumiaki; Hata, Tsunehisa; Shibata, Katsuhiko

PATENT ASSIGNEE(S): Honda Motor Co., Ltd., Japan

SOURCE: Jpn. Kokai Tokkyo Koho, 5 pp.

CODEN: JKXXAF

DOCUMENT TYPE: Patent

LANGUAGE: Japanese

FAMILY ACC. NUM. COUNT: 2

PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
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JP 2003342664	A	20031203	JP 2002-157329	20020530
JP 4092138	B2	20080528		
WO 2003102257	A1	20031211	WO 2003-JP5993	20030514
W:	AE, AG, AL, AM, AT, AU, AZ, BA, BB, BG, BR, BY, BZ, CA, CH, CN, CO, CR, CU, CZ, DE, DK, DM, DZ, EC, EE, ES, FI, GB, GD, GE, GH, GM, HR, HU, ID, IL, IN, IS, KE, KG, KP, KR, KZ, LC, LK, LR, LS, LT, LU, LV, MA, MD, MG, MK, MN, MW, MX, MZ, NI, NO, NZ, OM, PH, PL, PT, RO, RU, SC, SD, SE, SG, SK, SL, TJ, TM, TN, TR, TT, TZ, UA, UG, US, UZ, VC, VN, YU, ZA, ZM, ZW			
RW:	GH, GM, KE, LS, MW, MZ, SD, SL, SZ, TZ, UG, ZM, ZW, AM, AZ, BY, KG, KZ, MD, RU, TJ, TM, AT, BE, BG, CH, CY, CZ, DE, DK, EE, ES, FI, FR, GB, GR, HU, IE, IT, LU, MC, NL, PT, RO, SE, SI, SK, TR, BF, BJ, CF, CG, CI, CM, GA, GN, GQ, GW, ML, MR, NE, SN, TD, TG			
AU 2003235302	A1	20031219	AU 2003-235302	20030514
EP 1508627	A1	20050223	EP 2003-723374	20030514
R:	AT, BE, CH, DE, DK, ES, FR, GB, GR, IT, LI, LU, NL, SE, MC, PT, IE, SI, LT, LV, FI, RO, MK, CY, AL, TR, BG, CZ, EE, HU, SK			
US 20060137848	A1	20060629	US 2005-518151	20050927

PRIORITY APPLN. INFO.: JP 2002-157328 A 20020530
 JP 2002-157329 A 20020530
 WO 2003-JP5993 W 20030514

AB The alloys comprise Mg 3.5-4.5, Mn 0.8-1.5, Si <0.5, Fe <0.5 weight%, Ti, Zr [with their total content (Ti + Zr) ≥0.3 weight% and their content ratio Ti/Zr 0.3-2 weight%], and balance Al. The alloys have improved toughness. The alloys, having a preferable temperature for melt pouring 720-730°, are suitable for large-size thin castings with min. thickness 1.2-3 mm and maximum melt flow distance in a mold cavity ≥200 mm.

IT Cast alloys
 RL: TEM (Technical or engineered material use); USES (Uses)
 (aluminum; Al-Mg-based cast alloys with improved toughness for large-size thin castings)

IT 389625-99-4 628716-44-9 628716-45-0 628716-46-1
 RL: TEM (Technical or engineered material use); USES (Uses)
 (Al-Mg-based cast alloys with improved toughness for large-size thin castings)

L30 ANSWER 46 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2003:782867 HCAPLUS

DOCUMENT NUMBER: 139:279740

TITLE: Die-cast aluminum-magnesium alloy products having ribs

INVENTOR(S): Toyota, Yusuke; Shibata, Katsuhiro;
 Hata, Tsunehisa; Fukuchi, Fumiaki;
 Minakami, Takahiro

PATENT ASSIGNEE(S): Honda Motor Co., Ltd., Japan

SOURCE: Jpn. Kokai Tokkyo Koho, 11 pp.
 CODEN: JKXXAF

DOCUMENT TYPE: Patent

LANGUAGE: Japanese

FAMILY ACC. NUM. COUNT: 1

PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
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JP 2003285150	A	20031007	JP 2002-87514	20020327
US 20030219618	A1	20031127	US 2003-392391	20030320
PRIORITY APPLN. INFO.:			JP 2002-87514	A 20020327

AB The Al-Mg alloy product, obtained by die-casting, has a sheet main body and ≥1 rib, where the length direction of the rib intersects the alloy melt flowing direction. Preferably, the Al-Mg alloy contains Mg 3.5-4.5, Si ≤0.25, Mn 0.8-1.5, Fe ≤0.5, and Ti 0.1-0.3 weight%. Since the alloy melt at edge parts of the rib rapidly solidifies, generation of casting defects is suppressed, and the product has high strength and toughness.

IT Casting of metals
 (die; die-cast Al-Mg alloy products having ribs with high strength and toughness)

IT 607356-70-7 607356-71-8 607356-72-9 607356-73-0 607356-74-1
 607356-75-2 607356-76-3 607356-77-4 607356-78-5 607356-79-6
 RL: PEP (Physical, engineering or chemical process); PYP (Physical process); TEM (Technical or engineered material use); PROC (Process); USES (Uses)
 (die-cast Al-Mg alloy products having ribs with high strength and toughness)

L30 ANSWER 47 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2003:390171 HCAPLUS
DOCUMENT NUMBER: 138:389216
TITLE: Manufacture of aluminum alloy billets
processed by mushy-state forming for transportation
equipment
INVENTOR(S): Mikubo, Shigeru; Mizouchi, Masafumi; Murayama,
Yasuyuki; Iwashita, Tsunaki
PATENT ASSIGNEE(S): Kyushu Mitsui Aluminium Co., Ltd., Japan
SOURCE: Jpn. Kokai Tokkyo Koho, 5 pp.
CODEN: JKXXAF
DOCUMENT TYPE: Patent
LANGUAGE: Japanese
FAMILY ACC. NUM. COUNT: 1
PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
JP 2003147498	A	20030521	JP 2001-338928	20011105
JP 3852915	B2	20061206		

PRIORITY APPLN. INFO.: JP 2001-338928 20011105

AB An Al alloy billet containing Zn 3.5-7.5, Mg 0.50-4.0, Si ≤ 0.50 , Fe ≤ 0.55 , Ti 0.001-0.50 and/or B 0.0001-0.5, and Cu 0.30-3.0, Mn 0.03-0.80, Zr 0.03-0.35, Cr 0.03-0.35, and/or V 0.03-0.2 weight% and having dendrite arm spacing $\leq 200 \mu\text{m}$ is produced, and then the billet is cold die-forged for introduction of distortion at distortion ratio 5-50%, working rate $\leq 50 \text{ mm/s}$, and a temperature lower than recrystn. temperature, heated at a temperature equal to or higher than the solidus temperature, and formed under mushy state at a temperature where liquid phase ratio of the billet becomes 20-80%. The obtained Al alloy billet have uniform spheroidal structure. The billet is useful for producing automobile parts, and so on.

IT Forging
(cold forging; manufacture of Al alloy billet with uniform spheroidal structure by mushy-state forming for transportation equipment)
IT Heat treatment
(manufacture of Al alloy billet with uniform spheroidal structure by mushy-state forming for transportation equipment)
IT Metalworking
(mushy state forming; manufacture of Al alloy billet with uniform spheroidal structure by mushy-state forming for transportation equipment)
IT 527685-40-1 527685-41-2 527685-42-3 527685-43-4 527685-44-5
RL: PEP (Physical, engineering or chemical process); PYP (Physical process); TEM (Technical or engineered material use); PROC (Process); USES (Uses)
(manufacture of Al alloy billet with uniform spheroidal structure by mushy-state forming for transportation equipment)

L30 ANSWER 48 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2003:391115 HCAPLUS
DOCUMENT NUMBER: 138:389217
TITLE: Manufacture of aluminum alloy billets by
semisolid forging for transportation equipments
INVENTOR(S): Mikubo, Shigeru; Mizouchi, Masafumi; Murayama,
Yasuyuki; Iwashita, Tsunaki
PATENT ASSIGNEE(S): Kyushu Mitsui Aluminium Co., Ltd., Japan
SOURCE: Jpn. Kokai Tokkyo Koho, 5 pp.
CODEN: JKXXAF
DOCUMENT TYPE: Patent
LANGUAGE: Japanese

FAMILY ACC. NUM. COUNT: 1
PATENT INFORMATION:

	PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
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	JP 2003147497	A	20030521	JP 2001-337404	20011102
	JP 3802796	B2	20060726		
PRIORITY APPLN. INFO.:				JP 2001-337404	20011102
AB	The alloy comprises Cu 0-0.2, Si 0-0.5, Mg 2-6, Zn 0-0.35, Fe 0-0.5; Ti 0.001-0.5, and/or B 0.0001-0.5; Mn 0.05-1.5, Zr 0.03-0.35, and/or Cr 0.03-0.4; and Al bal. with the space of the dendrites being $\leq 200 \mu\text{m}$. The billets are manufactured by cold mold-forging at distortion rate 5-50%, feeding rate $\leq 50 \text{ mm/s}$, and temperature \leq recrystg. temperature, heating to $>$ solidus line, and holding at liquid phase ratio of 20-80%. Preferably, before forging the Al alloy is treated by homogenizing at 450-550° for 1-10 h.				
IT	Forging Transportation (manufacture of aluminum alloy billets by semisolid forging for transportation equipments)				
IT	Cast alloys RL: PEP (Physical, engineering or chemical process); PYP (Physical process); TEM (Technical or engineered material use); PROC (Process); USES (Uses) (manufacture of aluminum alloy billets by semisolid forging for transportation equipments)				
IT	117304-61-7	528578-84-9	528578-85-0	528578-86-1	528578-87-2
	RL: PEP (Physical, engineering or chemical process); PYP (Physical process); TEM (Technical or engineered material use); PROC (Process); USES (Uses) (manufacture of aluminum alloy billets by semisolid forging for transportation equipments)				

L30 ANSWER 49 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2003:558148 HCAPLUS

DOCUMENT NUMBER: 139:200678

TITLE: Investigation of the microstructure and properties of a friction stir welded Al-Mg-Sc alloy

AUTHOR(S): Lapasset, G.; Girard, Y.; Campagnac, M. H.; Boivin, D.

CORPORATE SOURCE: ONERA, Chatillon, 92230, Fr.

SOURCE: Materials Science Forum (2003), 426-432(Pt. 4, THERMEC'2003), 2987-2992
CODEN: MSFOEP; ISSN: 0255-5476

PUBLISHER: Trans Tech Publications Ltd.

DOCUMENT TYPE: Journal

LANGUAGE: English

AB The objective of this study is to provide a better knowledge of the relationships between microstructural evolution and both mech. and corrosion properties of a friction stir welded Al-Mg-Sc alloy. Microstructures were studied by optical microscopy, transmission electron microscopy, electron backscattered diffraction and microhardness measurements. Tensile testing was carried out in order to determine the global behavior of the joint as well as the local behavior of the main zones of the joint. The weld nugget appears to be the softest zone of the weld. Its undermatching is analyzed with reference to the various hardening contributions of relevance in Al-Mg-Sc alloys. It is tentatively concluded that recrystn. which occurred in the nugget at the expense of the fine unrecrystd. structure of the base metal is primarily responsible for the loss of strength of the joint. Accelerated corrosion tests did not show any evidence of susceptibility to intergranular attack.

IT Welding of metals
Welds

(friction, stir; microstructure, corrosion, and mech. properties of friction stir welded Al-Mg-Sc alloy)

IT Corrosion
(intergranular; microstructure, corrosion, and mech. properties of friction stir welded Al-Mg-Sc alloy)

IT Crystal dislocations
Ductility
Microhardness
Microstructure
Precipitation hardening
Tensile strength
Yield strength
(microstructure, corrosion, and mech. properties of friction stir welded Al-Mg-Sc alloy)

IT Recrystallization
(texture; microstructure, corrosion, and mech. properties of friction stir welded Al-Mg-Sc alloy)

IT 475150-16-4, C557
RL: CPS (Chemical process); PEP (Physical, engineering or chemical process); PRP (Properties); PYP (Physical process); PROC (Process)
(microstructure, corrosion, and mech. properties of friction stir welded Al-Mg-Sc alloy)

REFERENCE COUNT: 14 THERE ARE 14 CITED REFERENCES AVAILABLE FOR THIS RECORD. ALL CITATIONS AVAILABLE IN THE RE FORMAT

L30 ANSWER 50 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2003:970775 HCAPLUS

DOCUMENT NUMBER: 140:221376

TITLE: CO2 laser welding of aluminium shipbuilding industry alloys. AA 5083, AA 5383, AA 5059, and AA 6082

AUTHOR(S): Ancona, Antonio; Daurelio, G.; De Filippis, L. A. C.; Ludovico, A. D.; Spera, A. M.

CORPORATE SOURCE: Unita di Ricerca - Dipartimento Interateneo di Fisica INFN, Univ. degli Studi di Bari, Bari, 70126, Italy

SOURCE: Proceedings of SPIE-The International Society for Optical Engineering (2003), 5120(XIV International Symposium on Gas Flow, Chemical Lasers, and High-Power Lasers, 2002), 577-587
CODEN: PSISDG; ISSN: 0277-786X

PUBLISHER: SPIE-The International Society for Optical Engineering

DOCUMENT TYPE: Journal

LANGUAGE: English

AB Al alloys are interesting in many and many industrial applications, from the classical aircraft industry to rail and road vehicles manufacturing (High Speed Train, Car Structure and Body). Recently much more attention for Al Alloys, 5000 and 6000 Series, was carried out by Shipbuilding Industry, especially for using in the H.S.L.C. (High Speed Light Craft). Therefore the aim of this exptl. work was to study, develop and test a reproducible CO2 laser welding procedure and technique of four specific alloys, that is AA 5083, AA 5383, AA 5059 (Al-Mg Alloys), and AA 6082 (Al-Mg-Si Alloy). Different techniques, methodologies, covering gases, nozzles, focusing lenses and mirrors, welding speed range, laser power range (1000 and 2500 W) have been carefully experimented. The melted zones properties have been evaluated by cross sections, and some visual inspections by a NIKON LUCIA Imaging System correlating each exptl. test, results and evaluations to the adopted process parameters and to the thermo-phys. properties of the tested alloys.

IT Microstructure
(CO2 laser welding of Al shipbuilding industry alloys)

IT Welding of metals

(laser; CO2 laser welding of Al shipbuilding industry alloys)
 IT 12616-86-3, AA 5083 12732-13-7, AA 6082 269058-32-4, AA 5059
 327622-69-5, AA 5383
 RL: CPS (Chemical process); PEP (Physical, engineering or chemical
 process); TEM (Technical or engineered material use); PROC (Process); USES
 (Uses)
 (CO2 laser welding of Al shipbuilding industry alloys)
 IT 124-38-9, Carbon dioxide, uses
 RL: NUU (Other use, unclassified); USES (Uses)
 (CO2 laser welding of Al shipbuilding industry alloys)
 REFERENCE COUNT: 7 THERE ARE 7 CITED REFERENCES AVAILABLE FOR THIS
 RECORD. ALL CITATIONS AVAILABLE IN THE RE FORMAT

L30 ANSWER 51 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN
 ACCESSION NUMBER: 2003:793073 HCAPLUS
 DOCUMENT NUMBER: 140:203216
 TITLE: Development of new high strength Al-Sc filler wires
 for fusion welding 7000 series aluminium
 aerospace alloys
 AUTHOR(S): Norman, A. F.; Birley, S. S.; Prangnell, P. B.
 CORPORATE SOURCE: Manchester Materials Science Centre, University of
 Manchester and UMIST, Manchester, M1 7HS, UK
 SOURCE: Science and Technology of Welding and Joining (2003),
 8(4), 235-245
 CODEN: STWJFX; ISSN: 1362-1718
 PUBLISHER: Maney Publishing
 DOCUMENT TYPE: Journal
 LANGUAGE: English

AB It has been reported that the transition metal Sc can improve the
 weldability and mech. properties of Al aerospace alloys that are normally
 considered to be 'unweldable'. However, little is currently known about
 the mechanisms by which Sc leads to such improvements. Here, the effect
 of the Sc concentration in the fusion zone of metal inert gas (MIG) welds in a
 typical 7000 series Al aerospace alloy, 7050, has been investigated in
 detail. It was found that at a critical Sc level (.apprx.0.4 weight%) a
 dramatic level of grain refinement occurs, leading to a highly uniform,
 ultrafine (.apprx.10 μ m) grain structure across the entire the fusion
 zone. Grain refinement was accompanied by an increase in the concentration of
 solute that was retained in solid solution after solidification, which led to
 a reduction in the volume fraction of eutectic per unit grain boundary area and
 a significant age hardening response in the fusion zone. The tensile
 properties of single pass MIG welds, produced with an Al-Sc filler wire,
 were compared to welds made using com. Al filler wires recommended for
 welding 7000 series alloys (5087, 5180, and 5039). The Sc-containing filler
 wire (Al-4Mg-2.8Zn-0.8Sc-0.1Ti-0.15 weight% Zn) greatly outperformed the com.
 filler wires, both in terms of weld strength and ductility. Further
 improvements in the tensile properties of the welds were achieved by
 exploiting the enhanced aging response of the Al-Sc filler wires, through
 welding in a solution heat treated condition and using a post-weld aging
 treatment. The underlying metallurgical processes by which Sc brings
 about these improvements are discussed.

IT Grain refinement
 (by scandium; development of new-high strength Al-Sc alloy filler wires
 for fusion welding of 7000 series aluminum aerospace alloys)

IT Elongation at break
 Microhardness
 Yield strength
 (development of new-high strength Al-Sc alloy filler wires for fusion
 welding of 7000 series aluminum aerospace alloys)

IT Precipitation hardening
 (during weld cooling; development of new-high strength Al-Sc alloy
 filler wires for fusion welding of 7000 series aluminum

aerospace alloys)

IT Welding of metals
(electrodes; development of new-high strength Al-Sc alloy filler wires for fusion welding of 7000 series aluminum aerospace alloys)

IT Welding of metals
(gas metal-arc; development of new-high strength Al-Sc alloy filler wires for fusion welding of 7000 series aluminum aerospace alloys)

IT Tensile strength
(ultimate; development of new-high strength Al-Sc alloy filler wires for fusion welding of 7000 series aluminum aerospace alloys)

IT 7440-20-2, Scandium, properties
RL: MOA (Modifier or additive use); PRP (Properties); USES (Uses)
(development of new-high strength Al-Sc alloy filler wires for fusion welding of 7000 series aluminum aerospace alloys)

IT 37301-61-4, AA 7050
RL: PEP (Physical, engineering or chemical process); PRP (Properties); PYP (Physical process); PROC (Process)
(development of new-high strength Al-Sc alloy filler wires for fusion welding of 7000 series aluminum aerospace alloys)

IT 51809-18-8, AA 5039 56036-95-4, AA 5180 660823-56-3, Aluminum 92, magnesium 4, scandium 0.8, titanium 0.1, zinc 2.8, zirconium 0.2 661475-83-8, AA 5087
RL: NUU (Other use, unclassified); PEP (Physical, engineering or chemical process); PRP (Properties); PYP (Physical process); PROC (Process); USES (Uses)
(weld filler; development of new-high strength Al-Sc alloy filler wires for fusion welding of 7000 series aluminum aerospace alloys)

REFERENCE COUNT: 28 THERE ARE 28 CITED REFERENCES AVAILABLE FOR THIS RECORD. ALL CITATIONS AVAILABLE IN THE RE FORMAT

L30 ANSWER 52 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2003:567915 HCAPLUS

DOCUMENT NUMBER: 140:131940

TITLE: Selection of filler wire for arc welding of aluminum alloys of the Al-Mg and Al-Cu systems

AUTHOR(S): Ryazantsev, V. I.; Filatov, Yu. A.; Ignat'ev, Yu. E.

CORPORATE SOURCE: NIAT, Russia

SOURCE: Svarochnoe Proizvodstvo (2003), (5), 32-35
CODEN: SVAPAI; ISSN: 0491-6441

PUBLISHER: Izdatel'stvo Mashinostroenie

DOCUMENT TYPE: Journal

LANGUAGE: Russian

AB Principal possibility of the new filler wires application for arc welding of aluminum alloys at the Al-Mg and Al--Cu systems is considered. Investigation results of mech. properties and alloys weldability according to different methods are shown. Recommendations for new filler wires application for different constructions from aluminum alloys are given.

IT Welding of metals
(electrodes; selection of filler wire for arc welding of aluminum alloys of Al-Mg and Al-Cu systems)

IT 12672-17-2, D20 12732-16-0, AMg6 37301-69-2, AA 1420 37301-70-5, AMg2 39412-99-2, AMg61 55321-16-9, AMg1 60999-06-6, Alloy 1205 64159-59-7, Alloy 1557 81159-87-7, AMg4 125352-52-5, AMg3 125726-63-8, AA 01570 130297-82-4, AA 1421 133554-29-7, AA 01460 135666-56-7, Alloy 01461 171757-20-3, Alloy 01217 197098-79-6, Alloy 01535 262854-04-6, Alloy 01515 262854-06-8, Alloy 01523 262854-08-0, Alloy 01545 460732-00-7, Alloy 01545K
RL: CPS (Chemical process); PEP (Physical, engineering or chemical process); PRP (Properties); PYP (Physical process); PROC (Process)

(selection of filler wire for arc welding of aluminum alloys
of Al-Mg and Al-Cu systems)

L30 ANSWER 53 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2004:254782 HCAPLUS

DOCUMENT NUMBER: 141:228899

TITLE: Evaluation of Sc-bearing aluminum alloy C557
for aerospace applications

AUTHOR(S): Domack, Marcia S.; Dicus, Dennis L.

CORPORATE SOURCE: Langley Research Center, Hampton, VA, USA

SOURCE: NASA/TM (2002), NASA/TM-2002-211633, i-ii, 1-9
CODEN: NATMA4; ISSN: 0499-9320

DOCUMENT TYPE: Report

LANGUAGE: English

AB The performance of the Al-Mg-Sc alloy C557 was evaluated to
assess its potential for a broad range of aerospace applications,
including airframe and launch vehicle structures. Of specific interest
were mech. properties at anticipated service temps. and thermal stability
of the alloy. Performance was compared with conventional airframe
aluminum alloys and with other emerging aluminum alloys
developed for specific service environments. Mech. properties and
metallurgical structure were evaluated for com. rolled sheet in the
as-received H116 condition and after thermal exposures at 107°.
Metallurgical analyses were performed to define grain morphol. and
texture, strengthening ppts., and to assess the effect of thermal
exposure.

IT Aerospace industry

Delamination

Fracture toughness

Microstructure

Texture (metallographic)

Thermal stability

(evaluation of mech. properties of Sc-bearing aluminum alloy
C557 for aerospace applications at cryogenic temps.)

IT Tensile strength

(ultimate; evaluation of mech. properties of Sc-bearing
aluminum alloy C557 for aerospace applications at cryogenic
temps.)

IT 7440-20-2, Scandium, uses

RL: MOA (Modifier or additive use); USES (Uses)

(evaluation of mech. properties of Sc-bearing aluminum alloy
C557 for aerospace applications at cryogenic temps.)

IT 475150-16-4, C557

RL: PRP (Properties)

(evaluation of mech. properties of Sc-bearing aluminum alloy
C557 for aerospace applications at cryogenic temps.)

REFERENCE COUNT: 9 THERE ARE 9 CITED REFERENCES AVAILABLE FOR THIS
RECORD. ALL CITATIONS AVAILABLE IN THE RE FORMAT

L30 ANSWER 54 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2002:265193 HCAPLUS

DOCUMENT NUMBER: 136:298237

TITLE: Production method of automobile parts by die
casting and heat treatment of aluminum
alloys

INVENTOR(S): Aoyama, Shunzo; Miura, Masaki; Mikasa, Tetsuo;
Fukuchi, Fumiaki; Ogawa, Tsutomu

PATENT ASSIGNEE(S): Ahresty Corp., Japan; Honda Motor Co., Ltd.

SOURCE: Jpn. Kokai Tokkyo Koho, 6 pp.
CODEN: JKXXAF

DOCUMENT TYPE: Patent

LANGUAGE: Japanese

FAMILY ACC. NUM. COUNT: 1

PATENT INFORMATION:

	PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
	-----	----	-----	-----	-----
	JP 2002105611	A	20020410	JP 2000-292481	20000926
PRIORITY APPLN. INFO.:				JP 2000-292481	20000926
AB	Automobile parts are manufactured from Al alloys containing Cu ≤ 0.1 , Si 7.5-10.0, Mg 0.25-0.6, Fe ≤ 0.25 Mn 0.5-1.2 and optionally Sr 0.01-0.02% by die casting and T5 heat treatment. The T5 heat treatment is set according to Mg content, and the heat treating temperature and heat treating time are set from a range of 403-473K and 1-6 h, resp.				
IT	Automobiles (parts; production method of automobile parts by die casting and heat treatment of aluminum alloys)				
IT	Casting of metals Heat treatment (production method of automobile parts by die casting and heat treatment of aluminum alloys)				
IT	406946-28-9	406946-29-0	406946-30-3	406946-31-4	406946-32-5
	406946-33-6				
	RL: PEP (Physical, engineering or chemical process); PYP (Physical process); TEM (Technical or engineered material use); PROC (Process); USES (Uses) (production method of automobile parts by die casting and heat treatment of aluminum alloys)				

L30 ANSWER 55 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2002:265149 HCAPLUS

DOCUMENT NUMBER: 136:282821

TITLE: Aluminum alloys for die-casting and automobile sub-frames therefrom by die-casting

INVENTOR(S): Komasaki, Toru; Sasaki, Hideto; Nishi, Naomi; Fukuchi, Fumiaki; Mikasa, Tetsuo; Kubo, Nobuo

PATENT ASSIGNEE(S): Ryobi, Ltd., Japan; Honda Motor Co., Ltd.

SOURCE: Jpn. Kokai Tokkyo Koho, 10 pp.

CODEN: JKXXAF

DOCUMENT TYPE: Patent

LANGUAGE: Japanese

FAMILY ACC. NUM. COUNT: 1

PATENT INFORMATION:

	PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
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	JP 2002105572	A	20020410	JP 2000-293086	20000926
	EP 1213366	A2	20020612	EP 2001-307824	20010914
	EP 1213366	A3	20020731		
	R: AT, BE, CH, DE, DK, ES, FR, GB, GR, IT, LI, LU, NL, SE, MC, PT, IE, SI, LT, LV, FI, RO, MK, CY, AL, TR				
PRIORITY APPLN. INFO.:				JP 2000-293086	A 20000926
AB	Al alloys for die-casting contain Si 8.0-9.0, Mg 0.35-0.45, Mn 0.3-0.4, Be 0.002-0.008, Fe < 0.2 , Cu ≤ 0.2 , Zn ≤ 0.1 , Ni ≤ 0.1 and Sn $\leq 0.1\%$. Sub-frames for automobiles are manufactured from the Al alloys by high speed, high-pressure die-casting wherein the die is evacuated to make the die cavity ≤ 10 Pa using a high vacuum exhaust means.				
IT	Automobiles Casting of metals				

(aluminum alloys for die-casting
and automobile sub-frames therefrom by die-casting)
IT 406720-09-0 406720-10-3 406720-11-4
RL: PEP (Physical, engineering or chemical process); PYP (Physical
process); TEM (Technical or engineered material use); PROC (Process); USES
(Uses)
(aluminum alloys for die-casting
and automobile sub-frames therefrom by die-casting)

L30 ANSWER 56 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2002:35978 HCAPLUS

DOCUMENT NUMBER: 136:105878

TITLE: Aluminum alloy sheets for sacrificial
corrosion prevention and composites therewith

INVENTOR(S): Tsuruno, Akihiro

PATENT ASSIGNEE(S): Kobe Steel, Ltd., Japan

SOURCE: Jpn. Kokai Tokkyo Koho, 7 pp.

CODEN: JKXXAF

DOCUMENT TYPE: Patent

LANGUAGE: Japanese

FAMILY ACC. NUM. COUNT: 1

PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
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JP 2002012935	A	20020115	JP 2000-197090	20000629
JP 3749089	B2	20060222		

PRIORITY APPLN. INFO.: JP 2000-197090 20000629

AB Al alloy sheets, which are clad to Al or Al alloy sheets to prevent
corrosion of the Al or Al alloy sheets by preferential self corrosion,
contain 0.05-0.5% Ti and 0.05-0.3% Zr. The Al alloy sheets are clad to
one side of core materials from other Al alloy sheets to obtain composite
materials. Optionally, a braze from Al-Si alloy, Al-Si-Zn alloy or Al-Si-
Mg alloy is clad to the other side of the core materials. The
composite materials are used for radiators, heaters, condensers,
evaporators, etc. of automobiles.

IT Composites

(aluminum alloy sheets for sacrificial corrosion prevention
and aluminum alloy sheet-clad composites)

IT Condensers

Evaporators

Heaters

(automobile; aluminum alloy sheets for sacrificial corrosion
prevention and aluminum alloy sheet-clad composites for)

IT Radiators

(automotive; aluminum alloy sheets for sacrificial corrosion
prevention and aluminum alloy sheet-clad composites for)

IT Corrosion prevention

(sacrificial; aluminum alloy sheets for sacrificial corrosion
prevention and aluminum alloy sheet-clad composites)

IT 12670-40-5 58229-40-6 389625-99-4 389626-00-0 389626-01-1
389626-02-2 389626-03-3 389626-04-4 389626-05-5 389626-06-6
389626-07-7 389626-08-8 389626-11-3 389626-12-4

RL: PRP (Properties); TEM (Technical or engineered material use); USES
(Uses)

(aluminum alloy sheets for sacrificial corrosion prevention
and aluminum alloy sheet-clad composites)

IT 11099-22-2 12617-23-1 91275-79-5

RL: TEM (Technical or engineered material use); USES (Uses)

(braze; aluminum alloy sheets for sacrificial corrosion
prevention and aluminum alloy sheet-clad composites)

IT 211815-54-2 389626-13-5 389626-14-6 389626-15-7 389626-16-8

RL: TEM (Technical or engineered material use); USES (Uses)
(core; aluminum alloy sheets for sacrificial corrosion
prevention and aluminum alloy sheet-clad composites)

L30 ANSWER 57 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2002:887384 HCAPLUS

DOCUMENT NUMBER: 138:93724

TITLE: High temperature, high strain rate embrittlement of
Al-Mg-Mn alloy: evidence of
cleavage of an fcc alloy

AUTHOR(S): Deschamps, A.; Peron, S.; Brechet, Y.; Ehrstrom,
J.-C.; Poizat, L.

CORPORATE SOURCE: LTPCM/ENSEEG, CNRS UMR 5614, St Martin d'Heres, 38
402, Fr.

SOURCE: Materials Science and Technology (2002), 18(10),
1085-1091

CODEN: MSCTEP; ISSN: 0267-0836

PUBLISHER: Maney Publishing

DOCUMENT TYPE: Journal

LANGUAGE: English

AB The fracture behavior in tension of an Al-Mg-Mn alloy
has been investigated. At high temps. and strain rates, intergranular
brittle fracture is observed along with cleavage fracture. Intergranular
fracture is related to local melting at the grain boundaries. Cleavage
occurs in equal proportions on the {100} and {110} crystallog. planes.
The area fraction of cleavage facets on the fracture surface has been
quantified. Their initiation is shown to be related both to the liquid
metal embrittlement of the grain boundaries and to the presence of brittle
Mn containing particles at the grain boundaries. Cleavage fracture in
an aluminum alloy also requires an inhibition of plastic flow
which prevents plastic blunting at the crack tip. It is proposed that
this modification of the plastic behavior is provided by the decrease in
stacking fault energy at high temps. in Al-Mg alloys.

IT Brittle fracture
Embrittlement
Fracture surface morphology
Stacking fault energy
Strain

(cleavage fracture and high-temperature, high-strain rate embrittlement of
Al-Mg-Mn alloy)

IT Fracture (materials)
(ductile; cleavage fracture and high-temperature, high-strain rate
embrittlement of Al-Mg-Mn alloy)

IT Fracture (materials)
(intergranular; cleavage fracture and high-temperature, high-strain rate
embrittlement of Al-Mg-Mn alloy)

IT 327622-69-5, AA5383

RL: PEP (Physical, engineering or chemical process); PRP (Properties); PYP
(Physical process); PROC (Process)

(cleavage fracture and high-temperature, high-strain rate embrittlement of
Al-Mg-Mn alloy)

REFERENCE COUNT: 27 THERE ARE 27 CITED REFERENCES AVAILABLE FOR THIS
RECORD. ALL CITATIONS AVAILABLE IN THE RE FORMAT

L30 ANSWER 58 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2002:704425 HCAPLUS

DOCUMENT NUMBER: 137:373439

TITLE: Evaluation of Sc-bearing aluminum alloy C557
for aerospace applications

AUTHOR(S): Domack, M. S.; Dicus, D. L.

CORPORATE SOURCE: NASA Langley Research Center, Hampton, VA, 23681-2199,
USA

SOURCE: Materials Science Forum (2002), 396-402(Pt. 2,
Aluminium Alloys 2002), 839-844
CODEN: MSFOEP; ISSN: 0255-5476

PUBLISHER: Trans Tech Publications Ltd.

DOCUMENT TYPE: Journal

LANGUAGE: English

AB The performance of the Al-Mg-Sc alloy C557 was evaluated to
assess its potential for a broad range of aerospace applications,
including airframe and launch vehicle structures. Of specific interest
were mech. properties at anticipated service temps. and thermal stability
of the alloy. Performance was compared with conventional airframe Al
alloys and with other emerging Al alloys developed for specific service
environments. Mech. properties and metallurgical structure were evaluated
for com. rolled sheet in the as-received H116 condition and after thermal
exposures at 107°. Metallurgical analyses were performed to define
grain morphol. and texture, strengthening ppts., and to assess the effect
of thermal exposure.

IT Aerospace industry
Fracture toughness
Microstructure
Strength
(evaluation of Sc-bearing aluminum alloy C557 for aerospace
applications)

IT 7440-20-2, Scandium, processes
RL: MOA (Modifier or additive use); PEP (Physical, engineering or chemical
process); PRP (Properties); PYP (Physical process); PROC (Process); USES
(Uses)
(evaluation of Sc-bearing aluminum alloy C557 for aerospace
applications)

IT 475150-16-4, C557
RL: PRP (Properties)
(evaluation of Sc-bearing aluminum alloy C557 for aerospace
applications)

REFERENCE COUNT: 9 THERE ARE 9 CITED REFERENCES AVAILABLE FOR THIS
RECORD. ALL CITATIONS AVAILABLE IN THE RE FORMAT

L30 ANSWER 59 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2001:808251 HCAPLUS

DOCUMENT NUMBER: 135:347609

TITLE: Manufacture of nanosize aluminum alloy
powders by attrition milling with a surfactant

INVENTOR(S): Upadhy, Kamleshwar; Hoffman, Wesley P.

PATENT ASSIGNEE(S): United States Dept. of the Air Force, USA

SOURCE: U.S., 6 pp.
CODEN: USXXAM

DOCUMENT TYPE: Patent

LANGUAGE: English

FAMILY ACC. NUM. COUNT: 1

PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
-----	----	-----	-----	-----
US 6312643	B1	20011106	US 1997-957013	19971024
PRIORITY APPLN. INFO.:			US 1997-957013	19971024

AB The nanosize Al-alloy powder is prepared by attrition milling under inert
gas with mech. alloying in the presence of a surfactant to prevent surface
oxidation, nitridation, or contamination. The attrition mill is preferably
operated with tool-steel balls optionally precoated with WC, using
purified Ar atmospheric The prepared Al-alloy powders have the particle or
grain
size of 1-250 nm, and can be consolidated into a dense billet without
high-temperature sintering stage. The pressed billets can be finished for

desired microstructure and properties by hot isostatic pressing, extrusion, and/or forging, especially at 250-550° and 10-50 kpsi. The typical Al alloys for nanosize powder manufacture contain Zn 2.5-10, Mg 1.0-6.4, Cu 1.0-1.72, Zr 0.2-6.5, Ni 0.20-1.0, Fe 0.1-1.20, Si 0.05-1.5, Mn 0.1-2.5, Cr 0.1-2.5, Ti 0.02-0.5, and B 0.1-1.0%. The typical alloy with the grain size of 40-60 nm was hot-isostatically pressed to 98.5% of theor. d. and then extruded, and showed tensile strength of 90-102 kpsi, elongation of 4-6%, and Vickers microhardness of 1.39 GPa.

- IT Powder metallurgy
(Al-alloy; nanosize aluminum alloy powders manufactured by attrition milling with surfactant)
- IT Surfactants
(attrition milling with; manufacture of nanosize aluminum alloy powders by attrition milling with surfactant)
- IT Milling (size reduction)
(attrition, Al-alloy; manufacture of nanosize aluminum alloy powders by attrition milling with surfactant)
- IT Sintering
(low-temperature; nanosize aluminum alloy powders manufactured by attrition milling for low-temperature sintering)
- IT Aluminum alloy, base
RL: PEP (Physical, engineering or chemical process); PROC (Process)
(powder, attrition milling of; manufacture of nanosize aluminum alloy powders by attrition milling with surfactant)
- IT 7440-37-1, Argon, processes
RL: PEP (Physical, engineering or chemical process); PROC (Process)
(atmospheric, milling in; manufacture of nanosize aluminum alloy powders by attrition milling with surfactant in Ar)
- IT 12597-69-2, Steel, uses
RL: TEM (Technical or engineered material use); USES (Uses)
(balls, milling with; manufacture of nanosize aluminum alloy powders by attrition milling with surfactant)
- IT 12070-12-1, Tungsten carbide (WC)
RL: TEM (Technical or engineered material use); USES (Uses)
(coating, milling balls with; manufacture of nanosize aluminum alloy powders by attrition milling with surfactant)
- IT 371165-10-5
RL: PEP (Physical, engineering or chemical process); PROC (Process)
(powder, attrition milling of; manufacture of nanosize aluminum alloy powders by attrition milling with surfactant)

REFERENCE COUNT: 8 THERE ARE 8 CITED REFERENCES AVAILABLE FOR THIS RECORD. ALL CITATIONS AVAILABLE IN THE RE FORMAT

L30 ANSWER 60 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2001:18779 HCAPLUS
DOCUMENT NUMBER: 134:89611
TITLE: Manufacture of composite article and insert for it
INVENTOR(S): Toyota, Yusuke; Hata, Tsunehisa; Ito, Takeo;
Nagase, Katsuya; Shimizu, Hideo
PATENT ASSIGNEE(S): Honda Motor Co., Ltd., Japan
SOURCE: Jpn. Kokai Tokkyo Koho, 11 pp.
CODEN: JKXXAF
DOCUMENT TYPE: Patent
LANGUAGE: Japanese
FAMILY ACC. NUM. COUNT: 1
PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
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JP 2001001129	A	20010109	JP 1999-178732	19990624
PRIORITY APPLN. INFO.:			JP 1999-178732	19990624
AB The composite article consists of a light alloy cast and an				

Fe-base alloy insert. The insert is coated with Cu-Ni alloy containing 10-90 weight% Ni. The insert is obtained by coating of an Fe alloy body with a Ni layer and a Cu layer (and a Ag layer) successively and heat treatment of them in a reducing atmospheric Adhesion of the insert to the cast is improved.

- IT Cast alloys
RL: IMF (Industrial manufacture); TEM (Technical or engineered material use); PREP (Preparation); USES (Uses)
(aluminum; manufacture of composite article of light alloy cast and iron alloy insert)
- IT Alloys, processes
RL: PEP (Physical, engineering or chemical process); TEM (Technical or engineered material use); PROC (Process); USES (Uses)
(light; manufacture of composite article of light alloy cast and iron alloy insert)
- IT Casting of metals
Composites
(manufacture of composite article of light alloy cast and iron alloy insert)
- IT Cast alloys
RL: IMF (Industrial manufacture); TEM (Technical or engineered material use); PREP (Preparation); USES (Uses)
(manufacture of composite article of light alloy cast and iron alloy insert)
- IT Diffusion
(thermal, preparation of copper-nickel alloy for insert coating; manufacture of composite article of light alloy cast and iron alloy insert)
- IT Iron alloy, base
RL: TEM (Technical or engineered material use); USES (Uses)
(manufacture of composite article of light alloy cast and iron alloy insert)
- IT 39463-63-3P
RL: PNU (Preparation, unclassified); TEM (Technical or engineered material use); PREP (Preparation); USES (Uses)
(coating on insert; manufacture of composite article of light alloy cast and iron alloy insert)
- IT 37321-78-1, ADC12
RL: PEP (Physical, engineering or chemical process); TEM (Technical or engineered material use); PROC (Process); USES (Uses)
(manufacture of composite article of light alloy cast and iron alloy insert)
- IT 12743-56-5, S25C, uses 138342-39-9, Cast iron, (JIS FC250), uses
RL: TEM (Technical or engineered material use); USES (Uses)
(manufacture of composite article of light alloy cast and iron alloy insert)

L30 ANSWER 61 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2001:729831 HCAPLUS

DOCUMENT NUMBER: 135:260814

TITLE: Aluminum-magnesium die-casting alloy for automotive frames

INVENTOR(S): Spanjers, Martinus Godefridus Johannes; Haszler, Alfred Johann Peter; Sampath, Desikan

PATENT ASSIGNEE(S): Corus Aluminium Voerde G.m.b.H., Germany; Corus Aluminium Walzprodukte G.m.b.H.

SOURCE: Eur. Pat. Appl., 12 pp.

CODEN: EPXXDW

DOCUMENT TYPE: Patent

LANGUAGE: English

FAMILY ACC. NUM. COUNT: 1

PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
EP 1138794	A1	20011004	EP 2001-200977	20010315
EP 1138794	B1	20070214		
R: AT, BE, CH, DE, DK, ES, FR, GB, GR, IT, LI, LU, NL, SE, MC, PT, IE, SI, LT, LV, FI, RO, CY, TR				
AT 353983	T	20070315	AT 2001-200977	20010315
US 20020006352	A1	20020117	US 2001-816686	20010326
US 6773664	B2	20040810		
US 20040161359	A1	20040819	US 2004-776605	20040212
US 6929706	B2	20050816		
PRIORITY APPLN. INFO.:			EP 2000-201156	A 20000331
			EP 2000-203660	A 20001020
			US 2001-816686	A3 20010326
AB	The alloy for casting operations comprises Mg 2.7-6.0, Mn 0.4-1.4, Zn 0.10-1.5, Zr ≤0.3, V ≤0.3, Sc ≤0.3, Ti ≤0.2, Fe ≤1.0, Si ≤1.4%, balance - aluminum. In one embodiment, the alloy containing Mg 5.8, Mn 0.54, Zn 0.51, Si 0.34, Fe 0.23, Zr 0.11%, Al - balance, had in as-cast condition the yield strength of 170 MPa, ultimate tensile strength of 305 MPa, and elongation of 14.2% and was applicable for high-pressure die-casting.			
IT	Elongation, mechanical Tensile strength Yield strength (aluminum-magnesium die-casting alloy for automotive frames)			
IT	Automobiles (bodies; aluminum-magnesium die-casting alloy for automotive frames)			
IT	Casting of metals (die; aluminum-magnesium die-casting alloy for automotive frames)			
IT	361484-70-0	361484-71-1	361484-72-2	361484-73-3
	361484-74-4	361484-75-5	361484-76-6	
	RL: PEP (Physical, engineering or chemical process); PRP (Properties); PROC (Process) (aluminum-magnesium die-casting alloy for automotive frames)			
REFERENCE COUNT: 12 THERE ARE 12 CITED REFERENCES AVAILABLE FOR THIS RECORD. ALL CITATIONS AVAILABLE IN THE RE FORMAT				

L30 ANSWER 62 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2001:812752 HCAPLUS

DOCUMENT NUMBER: 135:374935

TITLE: Deformation and fracture behaviour of aluminium sections joined by means of metal-inert gas welding

AUTHOR(S): Oeser, Sabine; Memhard, Dieter; Blauel, Johann Georg
CORPORATE SOURCE: Fraunhofer Institut fuer Werkstoffmechanik, Freiburg, Germany

SOURCE: Schweissen & Schneiden (2001), 53(9), E202-E205, 566, 570-573
CODEN: SCSCA4; ISSN: 0036-7184

PUBLISHER: Verlag fuer Schweissen und Verwandte Verfahren
DVS-Verlag

DOCUMENT TYPE: Journal

LANGUAGE: English/German

AB Large AlMgSi0.7 sections joined by means of metal-inert gas welding were examined with regard to their deformation, damage and fracture behavior. In addition to fracture-mech. investigations, tensile tests were performed on

small specimens at different strain rates. With the aid of a model and by means of numerical simulation, the deformation behavior of the whole welded joint was determined from the information about the individual material regions (weld metal, heat-affected zone and parent metal). This method can be used in order to math. predict the behavior of a joint in relation to its deformation, load-carrying capacity and defect tolerance.

- IT Deformation (mechanical)
Fracture (materials)
Simulation and Modeling, physicochemical
(deformation and fracture of aluminum sections joined by metal-inert gas welding)
- IT Welding of metals
(gas metal-arc; deformation and fracture of aluminum sections joined by metal-inert gas welding)
- IT Welds
(metal-inert gas; deformation and fracture of aluminum sections joined by metal-inert gas welding)
- IT 81988-24-1, AlMgSi0.7
RL: DEV (Device component use); PEP (Physical, engineering or chemical process); PRP (Properties); PROC (Process); USES (Uses)
(deformation and fracture of aluminum sections joined by metal-inert gas welding)
- IT 75686-78-1, AlMg4.5MnZr
RL: MOA (Modifier or additive use); USES (Uses)
(filler; deformation and fracture of aluminum sections joined by metal-inert gas welding)
- REFERENCE COUNT: 6 THERE ARE 6 CITED REFERENCES AVAILABLE FOR THIS RECORD. ALL CITATIONS AVAILABLE IN THE RE FORMAT

L30 ANSWER 63 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2001:672423 HCAPLUS

DOCUMENT NUMBER: 135:307091

TITLE: Features of superplastic deformation of some aluminum alloys

AUTHOR(S): Novikov, I. I.; Portnoi, V. K.

CORPORATE SOURCE: Kafedra Metallovedeniya Tsvetnykh Metallov, Mosk. Gos. Inst. Stali Splavov, Moscow, Russia

SOURCE: Izvestiya Vysshikh Uchebnykh Zavedenii, Tsvetnaya Metallurgiya (2001), (4), 4-11
CODEN: IVUTAK; ISSN: 0021-3438

PUBLISHER: Moskovskii Gosudarstvennyi Institut Stali i Splavov

DOCUMENT TYPE: Journal

LANGUAGE: Russian

AB A group of superplastic D20, D19, AMg4 and Neopral aluminum alloys, which are characterized by high superplastic properties at temps. near solidus, was studied. They differ from other superplastic aluminum alloys (for example, 7475 alloy of the system Al-Zn-Mg-Cu) in low contribution of grain boundary sliding (7-20%) that is not characteristic for the majority of similar alloys. Working mechanisms in this group of superplastic aluminum alloys are intragranular dislocation sliding and diffusion creep, which are to result in extension of grains in the direction of deformation. This picture was observed in our expts. to the same degree as in other alloys. Qual. anal. of the grain shape and sizes under superplastic deformation of the studied alloys showed that insufficient grain boundary sliding is compensated by dynamic recrystn. to lead in dividing the extended grains in parts and generating new ones.

IT Plastic deformation
(superplastic; features of superplastic deformation of aluminum alloys)

IT 12672-17-2, D20 39331-96-9, D19 39461-63-7, AA7475 81159-87-7
, AMg4 110414-16-9, Neopral

RL: PEP (Physical, engineering or chemical process); PRP (Properties);
 PROC (Process)
 (features of superplastic deformation of aluminum alloys)

L30 ANSWER 64 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2000:806831 HCAPLUS

DOCUMENT NUMBER: 133:338606

TITLE: Aluminum-magnesium alloys
 resistant to corrosive exfoliation and suitable for
 welded construction

INVENTOR(S): Haszler, Alfred Johann Peter; Sampath, Desikan

PATENT ASSIGNEE(S): Corus Aluminium Walzprodukte G.m.b.H., Germany

SOURCE: PCT Int. Appl., 23 pp.

CODEN: PIXXD2

DOCUMENT TYPE: Patent

LANGUAGE: English

FAMILY ACC. NUM. COUNT: 1

PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
WO 2000066800	A1	20001109	WO 2000-EP4410	20000504
W:	AE, AL, AM, AT, AU, AZ, BA, BB, BG, BR, BY, CA, CH, CN, CR, CU, CZ, DE, DK, DM, EE, ES, FI, GB, GD, GE, GH, GM, HR, HU, ID, IL, IN, IS, JP, KE, KG, KP, KR, KZ, LC, LK, LR, LS, LT, LU, LV, MA, MD, MG, MK, MN, MW, MX, NO, NZ, PL, PT, RO, RU, SD, SE, SG, SI, SK, SL, TJ, TM, TR, TT, TZ, UA, UG, US, UZ, VN, YU, ZA, ZW			
RW:	GH, GM, KE, LS, MW, SD, SL, SZ, TZ, UG, ZW, AT, BE, CH, CY, DE, DK, ES, FI, FR, GB, GR, IE, IT, LU, MC, NL, PT, SE, BF, BJ, CF, CG, CI, CM, GA, GN, GW, ML, MR, NE, SN, TD, TG			
EP 1050595	A1	20001108	EP 1999-201391	19990504
R:	AT, BE, CH, DE, DK, ES, FR, GB, GR, IT, LI, LU, NL, SE, MC, PT, IE, SI, LT, LV, FI, RO			
CA 2370160	A1	20001109	CA 2000-2370160	20000504
CA 2370160	C	20041207		
EP 1177323	A1	20020206	EP 2000-931231	20000504
EP 1177323	B1	20030409		
EP 1177323	B2	20080716		
R:	AT, BE, CH, DE, DK, ES, FR, GB, GR, IT, LI, LU, NL, SE, MC, PT, IE, SI, LT, LV, FI, RO, CY			
AU 750846	B2	20020801	AU 2000-49229	20000504
JP 2002543289	T	20021217	JP 2000-615421	20000504
AT 237002	T	20030415	AT 2000-931231	20000504
PT 1177323	T	20030829	PT 2000-931231	20000504
ES 2194728	T3	20031201	ES 2000-931231	20000504
ZA 2001008805	A	20021025	ZA 2001-8805	20011025
IN 2001CN01489	A	20070427	IN 2001-CN1489	20011030
US 6695935	B1	20040224	US 2002-959602	20020215
US 20040109787	A1	20040610	US 2003-725501	20031203
PRIORITY APPLN. INFO.:			EP 1999-201391	A 19990504
			WO 2000-EP4410	W 20000504
			US 2002-959602	A1 20020215
AB	The Al-Mg alloys for welded structures resistant to corrosion contain Mg 3.5-6.0, Mn 0.4-1.2, Zn 0.4-1.5, Zr ≤0.25, Cr ≤0.3, Ti ≤0.2, Fe ≤0.5, Si ≤0.5, Cu ≤0.4%, and ≥1 of Bi 0.005-0.1, Pb 0.005-0.1, Sn 0.01-0.1, Ag 0.01-0.5, Sc 0.01-0.5, Li 0.01-0.5, V 0.01-0.3, Ce 0.01-0.3, Y 0.01-0.3, Ni 0.01-0.3, and impurities ≤0.15% total. The microalloying with Bi decreases the precipitation of Mg-containing phases on grain boundaries, resulting in corrosion resistance higher than that of the AA 5083 Al-Mg alloy. The alloy is optionally used as a drawn wire for welding filler. The alloy weld yield strength is			

≥140 MPa in rolled plates or extruded shapes, especially for shipbuilding or transportation vehicles. The typical Al-Mg alloy having tensile strength of 325 MPa, yield point 150 MPa, and elongation 20.5% contains Mg 4.85, Mn 0.65, Zn 0.59, Zr 0.10, Cr 0.04, Ti 0.10, Fe 0.15, Si 0.09, Cu 0.03, and Bi 0.07%.

IT Welds
(Al-Mg alloys for welded construction resistant to corrosive exfoliation)

IT Ships
(Al-Mg alloys for welded ship construction resistant to corrosive exfoliation)

IT Vehicles
(transportation; Al-Mg alloys for welded construction resistant to corrosive exfoliation)

IT 303953-81-3
RL: TEM (Technical or engineered material use); USES (Uses)
(alloying of; Al-Mg alloys for welded construction resistant to corrosive exfoliation)

IT 303953-83-5 303953-84-6 303953-85-7 303953-86-8 303953-87-9
303953-88-0 303953-89-1 303953-90-4 303953-91-5
RL: TEM (Technical or engineered material use); USES (Uses)
(microalloyed; Al-Mg alloys for welded construction resistant to corrosive exfoliation)

IT 303953-82-4
RL: TEM (Technical or engineered material use); USES (Uses)
(microalloying of; Al-Mg alloys for welded construction resistant to corrosive exfoliation)

IT 7440-69-9, Bismuth, uses
RL: MOA (Modifier or additive use); USES (Uses)
(microalloying with; Al-Mg alloys with Bi for welded structures resistant to corrosive exfoliation)

REFERENCE COUNT: 11 THERE ARE 11 CITED REFERENCES AVAILABLE FOR THIS RECORD. ALL CITATIONS AVAILABLE IN THE RE FORMAT

L30 ANSWER 65 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN
ACCESSION NUMBER: 2000:666670 HCAPLUS
DOCUMENT NUMBER: 133:241383
TITLE: Weldable aluminium alloy structural component
INVENTOR(S): Haszler, Alfred Johann Peter; Mechsner, Klaus Alfons
PATENT ASSIGNEE(S): Corus Aluminium Walzprodukte G.m.b.H., Germany
SOURCE: PCT Int. Appl., 29 pp.
CODEN: PIXXD2
DOCUMENT TYPE: Patent
LANGUAGE: English
FAMILY ACC. NUM. COUNT: 1
PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
WO 2000054967	A1	20000921	WO 2000-EP2549	20000317
W: AE, AL, AM, AT, AU, AZ, BA, BB, BG, BR, BY, CA, CH, CN, CR, CU, CZ, DE, DK, DM, EE, ES, FI, GB, GD, GE, GH, GM, HR, HU, ID, IL, IN, IS, JP, KE, KG, KP, KR, KZ, LC, LK, LR, LS, LT, LU, LV, MA, MD, MG, MK, MN, MW, MX, NO, NZ, PL, PT, RO, RU, SD, SE, SG, SI, SK, SL, TJ, TM, TR, TT, TZ, UA, UG, US, UZ, VN, YU, ZA, ZW				
RW: GH, GM, KE, LS, MW, SD, SL, SZ, TZ, UG, ZW, AT, BE, CH, CY, DE, DK, ES, FI, FR, GB, GR, IE, IT, LU, MC, NL, PT, SE, BF, BJ, CF, CG, CI, CM, GA, GN, GW, ML, MR, NE, SN, TD, TG				
CA 2367752	A1	20000921	CA 2000-2367752	20000317
CA 2367752	C	20040831		
US 6337147	B1	20020108	US 2000-527832	20000317

EP 1169177 A1 20020109 EP 2000-922538 20000317
R: AT, BE, CH, DE, DK, ES, FR, GB, GR, IT, LI, LU, NL, SE, MC, PT,
IE, SI, LT, LV, FI, RO
NZ 514456 A 20020927 NZ 2000-514456 20000317
JP 2002539328 T 20021119 JP 2000-605023 20000317
JP 4053243 B2 20080227
AU 760996 B2 20030529 AU 2000-42899 20000317
PRIORITY APPLN. INFO.: EP 1999-200831 A 19990318
WO 2000-EP2549 W 20000317

AB Improved shape and strength of the weld in a welded structure are obtained
by use of a weldable aluminum product comprising a structural
component which is a sheet, a plate or an extruded body and is made of an
aluminum alloy containing not more than 1.5 wt % Zn. This component
has, adhered on at least one side, a cladding layer made of an
AA7xxx-series alloy having a corrosion potential lower than that of said
alloy of said structural component. The alloy of the structural component
is preferably an AA5xxx-series alloy containing Mg in the range 2 to
6 wt %.

IT Transportation
(marine; weldable aluminum alloy structural component)

IT Aerospace industry
Automobiles
Construction materials
Corrosion-resistant materials
Welding
Welding of metals
(weldable aluminum alloy structural component)

IT 7439-95-4, Magnesium, uses 7440-66-6, Zinc, uses
RL: MOA (Modifier or additive use); USES (Uses)
(weldable aluminum alloy structural component)

IT 12616-73-8, AA 5183 12616-86-3, AA 5083 12675-84-2, AA 7072
269058-32-4, AA 5059 292606-84-9 292606-85-0 292606-86-1
RL: PEP (Physical, engineering or chemical process); PRP (Properties); TEM
(Technical or engineered material use); PROC (Process); USES (Uses)
(weldable aluminum alloy structural component)

REFERENCE COUNT: 5 THERE ARE 5 CITED REFERENCES AVAILABLE FOR THIS
RECORD. ALL CITATIONS AVAILABLE IN THE RE FORMAT

L30 ANSWER 66 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2000:314627 HCAPLUS
DOCUMENT NUMBER: 132:324736
TITLE: Aluminum-alloy plates for large weldable
parallel-plate structures with internal stiffener ribs
INVENTOR(S): Haszler, Alfred Johann Peter; Sampath, Desikan;
Mechsner, Klaus Alfons
PATENT ASSIGNEE(S): Hoogovens Aluminium Walzprodukte G.m.b.H., Germany
SOURCE: PCT Int. Appl., 28 pp.
CODEN: PIXXD2
DOCUMENT TYPE: Patent
LANGUAGE: English
FAMILY ACC. NUM. COUNT: 1
PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
WO 2000026020	A1	20000511	WO 1999-EP8316	19991029
W:	AE, AL, AM, AT, AU, AZ, BA, BB, BG, BR, BY, CA, CH, CN, CR, CU, CZ, DE, DK, DM, EE, ES, FI, GB, GD, GE, GH, GM, HR, HU, ID, IL, IN, IS, JP, KE, KG, KP, KR, KZ, LC, LK, LR, LS, LT, LU, LV, MA, MD, MG, MK, MN, MW, MX, NO, NZ, PL, PT, RO, RU, SD, SE, SG, SI, SK, SL, TJ, TM, TR, TT, TZ, UA, UG, US, UZ, VN, YU, ZA, ZW			
RW:	GH, GM, KE, LS, MW, SD, SL, SZ, TZ, UG, ZW, AT, BE, CH, CY, DE,			

	DK, ES, FI, FR, GB, GR, IE, IT, LU, MC, NL, PT, SE, BF, BJ, CF,			
	CG, CI, CM, GA, GN, GW, ML, MR, NE, SN, TD, TG			
CA	2349004	A1	20000511	CA 1999-2349004 19991029
CA	2349004	C	20060613	
BR	9914953	A	20010724	BR 1999-14953 19991029
EP	1133390	A1	20010919	EP 1999-955933 19991029
EP	1133390	B1	20040310	
	R: AT, BE, CH, DE, DK, ES, FR, GB, GR, IT, LI, LU, NL, SE, MC, PT,			
	IE, FI			
TR	200101171	T2	20010921	TR 2001-1171 19991029
AU	747689	B2	20020516	AU 2000-12681 19991029
JP	2003502167	T	20030121	JP 2000-579436 19991029
CN	1103280	C	20030319	CN 1999-812945 19991029
AT	261354	T	20040315	AT 1999-955933 19991029
PT	1133390	T	20040730	PT 1999-955933 19991029
ES	2214898	T3	20040916	ES 1999-955933 19991029
RU	2237578	C2	20041010	RU 2001-114506 19991029
ZA	2001002910	A	20011012	ZA 2001-2910 20010409
NO	2001002107	A	20010427	NO 2001-2107 20010427
US	6848233	B1	20050201	US 2001-830448 20010730
PRIORITY APPLN. INFO.:				EP 1998-203665 A 19981030
				EP 1999-201767 A 19990604
				WO 1999-EP8316 W 19991029

AB The large Al-alloy panel is manufactured from 2 parallel plates with the intermediate corrugated stiffener sheet made of the Al alloy containing Mg 1.5-6.0, Mn 0.3-1.4, Zn 0.4-5.0, Fe ≤0.5, Si ≤0.5, Zr ≤0.30, and optionally Cr 0.05-0.30, Ti 0.01-0.20, V 0.05-0.25, Ag 0.05-0.40, and/or Cu ≤0.40% with impurities at ≤0.05 each and 0.15% total. The corrugated stiffener sheet is preferably 0.2-1.0 mm thick, and is heat treated for the H-temper or O-anneal with the yield/tensile strength ratio of 0.4-0.9, and having good rolling and bending formability. The Al-alloy stiffener sheet is optionally clad with higher-purity Al alloy, and after corrugation is welded to the parallel plates (especially by laser-beam welding) to manufacture the structural laminate suitable for marine-ship decks and panel applications. The typical Al alloy for the stiffener sheet .apprx.1 mm thick contains Mg 5.20, Mn 0.84, Zn 0.50, Fe 0.19, Si 0.11, Zr 0.13, Cr 0.049, Ti 0.015, and Cu 0.013%, and shows tensile strength of .apprx.315 MPa in the O-temper, vs. only .apprx.157 MPa for the AA 3004 alloy.

IT Construction materials
(boards, parallel-plate; aluminum-magnesium alloy for corrugated stiffener core in parallel-plate panels)

IT Welding of metals
(laser, of Al-alloy sheets; aluminum-magnesium alloy for clad-sheet core welded in parallel-plate panels)

IT Cladding
(of Al-alloy sheets; aluminum-magnesium alloy for clad-sheet stiffener core in parallel-plate panels)

IT Ships
(structural panels for; aluminum-magnesium alloy for clad-sheet core welded in parallel-plate panels)

IT 267005-59-4
RL: TEM (Technical or engineered material use); USES (Uses)
(alloying of; aluminum-magnesium alloy for corrugated stiffener core in parallel-plate panels)

IT 267005-60-7 267005-61-8
RL: TEM (Technical or engineered material use); USES (Uses)
(for panel laminates; aluminum-magnesium alloy for corrugated stiffener core in parallel-plate panels)

REFERENCE COUNT: 8 THERE ARE 8 CITED REFERENCES AVAILABLE FOR THIS RECORD. ALL CITATIONS AVAILABLE IN THE RE FORMAT

L30 ANSWER 67 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 2001:116174 HCAPLUS
DOCUMENT NUMBER: 134:196601
TITLE: High temperature cleavage in Al-Mg alloys
AUTHOR(S): Peron, S.; Brechet, Y.; Deschamps, A.; Ehrstrom, J. C.; Poizat, L.
CORPORATE SOURCE: Pechiney Centre de Recherches de Voreppe, Voreppe, 38341, Fr.
SOURCE: Advances in Mechanical Behaviour, Plasticity and Damage, Proceedings, Tours, France, Nov. 7-9, 2000 (2000), Volume 2, 1297-1302. Editor(s): Miannay, Dominique. Elsevier Science Ltd.: Oxford, UK. CODEN: 69AYFC
DOCUMENT TYPE: Conference
LANGUAGE: English

AB The fracture behavior of a high strength 5000 series aluminum alloy has been investigated in conditions relevant to industrial reversible hot rolling. To do so, a ductility test implemented on a servo-hydraulic compression machine was used. The material was deformed in the range 490°C to 560°C at a high strain rate (up to 25 s⁻¹). Under these conditions, the material exhibits a ductile to brittle transition. The brittle behavior is due to both intergranular and transgranular cleavage like fracture. Particular attention was paid on the latter mechanism. Examns. of the fracture surface with the SEM directly after deformation or after etch-pitting the surface were performed. Cross sections were also examined in the optical microscope after anodic oxidation, to confirm the transgranular aspect of fracture. The corresponding mechanism is found to be cleavage, as transgranular planar fracture surfaces are crystallog. well defined. Cleavage takes place both on (100) and (110) planes with the same frequency, independently of the deformation conditions in the cleavage range. A possible scenario for the appearance of high temperature cleavage has been proposed. It involves liquid metal embrittlement (LME) at some grain boundaries due to the fusion of deep eutectics. This LME leads to sharp cracks able to initiate cleavage under fast deformation. This scenario is consistent with metallog. observations of local melting as well as with the dependence of the overall features of fracture with temperature and strain rates.

IT Structural phase transition
(ductile-to-brittle; high-temperature cleavage in Al-Mg alloys)

IT Ductility
Embrittlement
Fracture (materials)
Fracture surface morphology
Plastic deformation
Strain
(high-temperature cleavage in Al-Mg alloys)

IT 327622-69-5, AA 5383
RL: PEP (Physical, engineering or chemical process); PRP (Properties); PROC (Process)
(high-temperature cleavage in Al-Mg alloys)

REFERENCE COUNT: 11 THERE ARE 11 CITED REFERENCES AVAILABLE FOR THIS RECORD. ALL CITATIONS AVAILABLE IN THE RE FORMAT

L30 ANSWER 68 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 1999:549420 HCAPLUS
DOCUMENT NUMBER: 131:160372
TITLE: High-strength aluminum-magnesium alloys for application in welded construction
INVENTOR(S): Haszler, Alfred Johann Peter; Sampath, Desikan
PATENT ASSIGNEE(S): Hoogovens Aluminium Walzprodukte G.m.b.H., Germany
SOURCE: PCT Int. Appl., 20 pp.

DOCUMENT TYPE: Patent
 LANGUAGE: English
 FAMILY ACC. NUM. COUNT: 1
 PATENT INFORMATION:

CODEN: PIXXD2

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
WO 9942627	A1	19990826	WO 1999-EP1011	19990218
W: AL, AM, AT, AU, AZ, BA, BB, BG, BR, BY, CA, CH, CN, CU, CZ, DE, DK, EE, ES, FI, GB, GE, GH, GM, HR, HU, ID, IL, IN, IS, JP, KE, KG, KP, KR, KZ, LC, LK, LR, LS, LT, LU, LV, MD, MG, MK, MN, MW, MX, NO, NZ, PL, PT, RO, RU, SD, SE, SG, SI, SK, SL, TJ, TM, TR, TT, UA, UG, US, UZ, VN, YU, ZW				
RW: GH, GM, KE, LS, MW, SD, SZ, UG, ZW, AT, BE, CH, CY, DE, DK, ES, FI, FR, GB, GR, IE, IT, LU, MC, NL, PT, SE, BF, BJ, CF, CG, CI, CM, GA, GN, GW, ML, MR, NE, SN, TD, TG				
AU 9927257	A	19990906	AU 1999-27257	19990218
EP 1078109	A1	20010228	EP 1999-907554	19990218
EP 1078109	B1	20030122		
EP 1078109	B2	20060913		
R: AT, BE, CH, ES, FR, GB, IT, LI, NL, SE, PT				
TR 200003222	T2	20010921	TR 2000-3222	19990218
AT 231562	T	20030215	AT 1999-907554	19990218
PT 1078109	T	20030630	PT 1999-907554	19990218
ES 2191418	T3	20030901	ES 1999-907554	19990218
BR 9909219	A	20050412	BR 1999-9219	19990218
ZA 9901360	A	19990820	ZA 1999-1360	19990219
NO 2000004154	A	20000818	NO 2000-4154	20000818
US 20030145912	A1	20030807	US 2002-299814	20021120
PRIORITY APPLN. INFO.:			EP 1998-200560	A 19980220
			WO 1999-EP1011	W 19990218
			US 2000-622157	B1 20001016

AB The Al-Mg alloys suitable for extrusions or rolled strip contain Mg >3.0 to 4.5 (especially 3.5-4.5), Mn 0.4-1.2, Zn 0.4-1.7, Zr 0.05-0.25, Cr ≤0.3, Ti ≤0.2, V ≤0.2, Li ≤0.5, Sc ≤0.5, Fe ≤0.5, Si ≤0.5, Cu ≤0.15, and Ag ≤0.4% with impurities ≤0.05 each and ≤0.15% total. The Al- Mg alloy is suitable for manufacture of high-strength containers or welded structural parts, especially for operation near 80-100°. The alloy ingot is typically preheated at 300-530° to decrease segregation, hot rolled, and optionally finished by cold rolling, and the resulting strip is heat treated for high-strength applications and corrosion resistance. The typical alloy for manufacture of the strip 1.2 mm thick with longitudinal tensile strength of 292 MPa contains Mg 3.9, Mn 0.74, Zn 0.53, Zr 0.13, Cr 0.05, Ti 0.02, Fe 0.31, Si 0.14, and Cu 0.05%. The alloy strength and ductility are comparable to those of the low-Zn AA 5083 Al-alloy strip susceptible to sensitized and intergranular corrosion.

IT Welding of metals
 (aluminum alloys; aluminum-magnesium alloy for high-strength strip and welded construction)

IT 237423-56-2
 RL: TEM (Technical or engineered material use); USES (Uses)
 (alloying of; aluminum-magnesium alloy for high-strength strip and welded construction)

IT 237423-57-3 237423-58-4 237423-59-5 237423-60-8
 RL: TEM (Technical or engineered material use); USES (Uses)
 (aluminum-magnesium alloy for high-strength strip and welded construction)

REFERENCE COUNT: 9 THERE ARE 9 CITED REFERENCES AVAILABLE FOR THIS RECORD. ALL CITATIONS AVAILABLE IN THE RE FORMAT

L30 ANSWER 69 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 1999:344803 HCAPLUS
DOCUMENT NUMBER: 130:355579
TITLE: Pressure-cast aluminum alloy structural parts
INVENTOR(S): Winkler, Reinhard; Wust, Jurgen
PATENT ASSIGNEE(S): Alusuisse Technology & Management AG, Switz.; Alcan Technology & Management AG
SOURCE: Eur. Pat. Appl., 6 pp.
CODEN: EPXXDW
DOCUMENT TYPE: Patent
LANGUAGE: German
FAMILY ACC. NUM. COUNT: 1
PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
EP 918095	A1	19990526	EP 1997-810884	19971120
EP 918095	B1	20030326		
R: AT, BE, CH, DE, DK, ES, FR, GB, GR, IT, LI, LU, NL, SE, MC, PT, IE, SI, LT, LV, FI, RO				
PT 918095	T	20030630	PT 1997-810884	19971120
ES 2192257	T3	20031001	ES 1997-810884	19971120
HU 9802626	A1	19990928	HU 1998-2626	19981112
HU 220128	B	20011128		
PL 186936	B1	20040430	PL 1998-329760	19981118
BR 9804709	A	19991109	BR 1998-4709	19981119
PRIORITY APPLN. INFO.:			EP 1997-810884	A 19971120
AB	The A1 alloy contains Sc 0.05-0.4 and Zr 0.1-0.4 in addition to Si ≤0.5, Fe ≤0.1, Mn 0.1-1.6, Mg ≤5.0, Ti ≤0.3, and Zn ≤0.1%. The alloy is suitable for production of structural parts (especially crash elements and space frame knots for motor vehicles and automobiles) by pressure casting. The parts can be used at ≤180°. Requirements on strength and ductility are fulfilled already in the as-cast condition and optionally after heat treatment at 200-400° but without high-temperature annealing.			
IT	Cast alloys RL: DEV (Device component use); TEM (Technical or engineered material use); USES (Uses) (aluminum; pressure-cast aluminum alloy crash elements for automobiles)			
IT	Automobiles (parts; pressure-cast aluminum alloy for)			
IT	Safety (pressure-cast aluminum alloy crash elements for automobiles)			
IT	7440-20-2, Scandium, uses 7440-67-7, Zirconium, uses RL: TEM (Technical or engineered material use); USES (Uses) (in pressure-cast aluminum alloy)			
IT	224648-10-6 224648-13-9 224648-15-1 RL: DEV (Device component use); TEM (Technical or engineered material use); USES (Uses) (pressure-cast aluminum alloy crash elements for automobiles)			
REFERENCE COUNT:	4	THERE ARE 4 CITED REFERENCES AVAILABLE FOR THIS RECORD. ALL CITATIONS AVAILABLE IN THE RE FORMAT		

L30 ANSWER 70 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 1999:660710 HCAPLUS
DOCUMENT NUMBER: 132:38694
TITLE: Application of metal matrix composites to automotive manufacturing

AUTHOR(S): Hayashi, Tadayoshi; Shibata, Katsuhiro;
 Shibata, Kazuo
 CORPORATE SOURCE: Honda R&D Co., Ltd., Tochigi, 321-33, Japan
 SOURCE: Advances in Science and Technology (Faenza, Italy)
 (1999), 22(Advanced Structural Fiber Composites),
 381-392
 CODEN: ASETE5
 PUBLISHER: Techna
 DOCUMENT TYPE: Journal; General Review
 LANGUAGE: English

AB A review with 6 refs. is given of the application of MMCs for autoparts
 and of the reasons why their application has not been expanded. After
 considering the issues, two examples of studies on MMCs for future
 applications are described. For engine block applications, the authors
 explain results of a process improvement which enables the usual
 high-pressure die-casting process to be used. For
 brake disk applications the authors describe results of an MMC brake disk
 performance in an automotive application.

IT Metal matrix composites
 (application of Al alloy metal matrix composites to
 automotive manufacture)

IT Engines
 (cylinder blocks, automobile; application of Al alloy
 metal matrix composites to automotive manufacture)

IT Casting of metals
 (die, of composites; application of Al alloy metal
 matrix composites to automotive manufacture)

IT Brakes (mechanical)
 (disk, automobile; application of Al alloy metal
 matrix composites to automotive manufacture)

IT 1344-28-1, Alumina, uses 12616-75-0, Aa6061 37263-88-0
 RL: DEV (Device component use); USES (Uses)
 (composites; application of Al alloy metal matrix
 composites to automotive manufacture)

REFERENCE COUNT: 6 THERE ARE 6 CITED REFERENCES AVAILABLE FOR THIS
 RECORD. ALL CITATIONS AVAILABLE IN THE RE FORMAT

L30 ANSWER 71 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN
 ACCESSION NUMBER: 1998:493351 HCAPLUS
 DOCUMENT NUMBER: 129:139368
 ORIGINAL REFERENCE NO.: 129:28416h,28417a
 TITLE: Beam support made of a light alloy for construction
 industry
 INVENTOR(S): Torimizu, Yoshimei
 PATENT ASSIGNEE(S): Furukawa Electric Co., Ltd., Japan
 SOURCE: Jpn. Kokai Tokkyo Koho, 7 pp.
 CODEN: JKXXAF
 DOCUMENT TYPE: Patent
 LANGUAGE: Japanese
 FAMILY ACC. NUM. COUNT: 1
 PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
-----	----	-----	-----	-----
JP 10195876	A	19980728	JP 1997-170794	19970627
PRIORITY APPLN. INFO.:			JP 1996-304696	A 19961115

AB A beam support installed in trenches to support walls comprises a shaft
 that can move reciprocally in a cylindrical body and side plates for
 driving and locking the shaft. The plates and the shaft are made of an Al
 alloy, preferably, containing Mg 2.1-2.9, Si 0.3-0.4, Fe 0.3-0.5, Cu
 1.2-2.0, Mn 0.25-0.35, Zn 5.1-7.0, Cr 0.18-0.28, and Zr + Ti
 ≤0.25%, or Mg 0.1-3.7, Si 0.06-0.7, Fe 0.08-1.4, Cu

0.05-2.6, Mn \leq 0.8, Zn 0.8-9.7, Cr 0-0.35, Zr 0-0.5, and Ti 0-0.2%. The difference in hardness between the shaft and the locking plate is -60~30.

IT Shafts
(beam support made of light alloy for construction industry)
IT Hardness (mechanical)
(of shaft and lock plate; beam support made of light alloy for construction industry)
IT 12616-75-0, Aa6061 12627-49-5, Aa7075 89701-09-7, Aa6066 210692-09-4
210692-10-7 210692-11-8 210692-12-9 210692-13-0 210692-15-2
210692-17-4 210692-20-9
RL: TEM (Technical or engineered material use); USES (Uses)
(beam support made of aluminum alloy for construction industry)

L30 ANSWER 72 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 1998:794168 HCAPLUS

DOCUMENT NUMBER: 130:84627

TITLE: Dependence of the thermal conductivity of alloys of the Al-Mg system on the composition and temperature

AUTHOR(S): Vertogradskii, V. A.; Bel'skayal, I. N.

CORPORATE SOURCE: All-Russia Institute of Aircraft Materials (VIAM), Moscow, Russia

SOURCE: Metal Science and Heat Treatment (Translation of Metallovedenie i Termicheskaya Obrabotka Metallov) (1998), 40(5-6), 231-233
CODEN: MHTRAN; ISSN: 0026-0673

PUBLISHER: Consultants Bureau

DOCUMENT TYPE: Journal

LANGUAGE: English

AB It is a common practice to study the dependencies of the phys. properties of alloys on the temperature and the composition without generalizing the results.

It is more logical to study these dependences complexly, i.e., as fragments of so-called composition-temperature-property diagrams (whole diagrams in the ideal case). Today's math. and computer possibilities provide processing of the dependences of any property on the temperature and composition even

for multicomponent systems. Math. analogs replace the composition-property graphical diagrams. The present work generalizes data on the thermal conductivity of 11 alloys of the Al-Mg system that contain 1 to 14% Mg in the temperature range of 20-350°C. The results are obtained in the form of a single regression equation that describes the data on the thermal conductivity within the range 86-190 W/(m · K) with a standard deviation of 0.7%. The choice of the regression equation is based on the existence of an analogy between heat transfer and elec. transfer in metallic systems and on dependences of the elec. resistivity on the temperature and the concentration of the alloying elements known from solid-state physics.

IT Thermal conductivity
(dependence of thermal conductivity of Al-Mg alloys on composition and temperature)

IT Electric resistance
Heat transfer
(in modeling of dependence of thermal conductivity of Al-Mg alloys on composition and temperature)

IT Simulation and Modeling, physicochemical
(regressive; dependence of thermal conductivity of Al-Mg alloys on composition and temperature)

IT 12732-16-0, AMG6 12773-43-2, AL8 37301-70-5, AMG2 55321-16-9, AMG1
61089-26-7, AL13 72267-09-5, AMG5 81159-87-7, AMG4

125352-52-5, AMG3 125726-63-8, Alloy 1570 135667-16-2, VAL16
218268-83-8, AL22 alloy

RL: PRP (Properties)

(dependence of thermal conductivity of Al-Mg alloys on composition and temperature)

REFERENCE COUNT: 4 THERE ARE 4 CITED REFERENCES AVAILABLE FOR THIS RECORD. ALL CITATIONS AVAILABLE IN THE RE FORMAT

L30 ANSWER 73 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 1997:678977 HCAPLUS

DOCUMENT NUMBER: 127:321810

ORIGINAL REFERENCE NO.: 127:63027a,63030a

TITLE: Aluminum-magnesium alloys for high-strength plates and large welded structures

INVENTOR(S): Haszler, Alfred Johann Peter; Sampath, Desikan

PATENT ASSIGNEE(S): Hoogovens Aluminium Walzprodukte G.m.b.H., Germany

SOURCE: Eur. Pat. Appl., 8 pp.

CODEN: EPXXDW

DOCUMENT TYPE: Patent

LANGUAGE: English

FAMILY ACC. NUM. COUNT: 3

PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
-----	----	-----	-----	-----
EP 799900	A1	19971008	EP 1996-200967	19960404
R: NL				
CA 2250977	A1	19971016	CA 1997-2250977	19970327
CA 2250977	C	20020326		
WO 9738146	A1	19971016	WO 1997-EP1623	19970327
W: AL, AM, AT, AU, AZ, BA, BB, BG, BR, BY, CA, CH, CN, CU, CZ, DE, DK, EE, ES, FI, GB, GE, HU, IL, IS, JP, KE, KG, KP, KR, KZ, LC, LK, LR, LS, LT, LU, LV, MD, MG, MK, MN, MW, MX, NO, NZ, PL, PT, RO, RU, SD, SE, SG, SI, SK, TJ, TM, TR, TT, UA, UG, US, UZ, VN				
RW: GH, KE, LS, MW, SD, SZ, UG, AT, BE, CH, DE, DK, ES, FI, FR, GB, GR, IE, IT, LU, MC, NL, PT, SE, BF, BJ, CF, CG, CI, CM, GA, GN, ML, MR, NE, SN, TD, TG				
AU 9722933	A	19971029	AU 1997-22933	19970327
AU 735772	B2	20010712		
EP 892858	A1	19990127	EP 1997-915470	19970327
EP 892858	B1	20001102		
EP 892858	B2	20070815		
R: AT, BE, CH, DE, DK, ES, FR, GB, GR, IT, LI, NL, SE, PT, FI				
CN 1217030	A	19990519	CN 1997-194225	19970327
CN 1061697	C	20010207		
JP 11507102	T	19990622	JP 1997-535649	19970327
JP 3262278	B2	20020304		
BR 9708513	A	20000104	BR 1997-8513	19970327
NZ 331972	A	20000428	NZ 1997-331972	19970327
TR 9801984	T2	20000721	TR 1998-1984	19970327
AT 197317	T	20001115	AT 1997-915470	19970327
ES 2153189	T3	20010216	ES 1997-915470	19970327
PT 892858	T	20010430	PT 1997-915470	19970327
RU 2194787	C2	20021220	RU 1998-119895	19970327
IN 1997MA00692	A	20060929	IN 1997-MA692	19970402
ZA 9702889	A	19971103	ZA 1997-2889	19970404
NO 9804634	A	19981002	NO 1998-4634	19981002
NO 326337	B1	20081110		
KR 2000005424	A	20000125	KR 1998-708178	19981002
US 6238495	B1	20010529	US 1999-155652	19990224
HK 1019235	A1	20010713	HK 1999-104293	19991004
GR 3035225	T3	20010430	GR 2001-400041	20010111

US 20010025675	A1	20011004	US 2001-785523	20010220
US 6342113	B2	20020129		
PRIORITY APPLN. INFO.:			EP 1996-200967	A 19960404
			EP 1997-915470	A 19970327
			WO 1997-EP1623	W 19970327
			US 1999-155652	A1 19990224

AB The high-strength Al-Mg alloys contain Mg 4.5-7, Mn 0.4-1.2, Zn 0.4-5, Zr ≤ 0.3 , Cr ≤ 0.3 , Ti ≤ 0.2 , Fe ≤ 0.5 , Si ≤ 0.5 , and Cu $\leq 0.4\%$ with residual impurities at ≤ 0.05 each and $\leq 0.15\%$ total. The alloy ingots are preheated for homogenizing and hot rolled at 400-530°, and the resulting plates are cold rolled with intermediate annealing and finally annealed at 200-550°. The low-d. alloy typically contains Mg 5.2-5.6, Mn 0.7-0.9, and Zn 0.4-1.5%, and shows increased strength compared with that of AA 5083 alloy, as well as similar resistance to corrosion and pitting. The typical alloy with tensile strength of 404 MPa after heat treatment to H321 temper contains Mg 4.7, Mn 0.8, Zn 0.6, Zr 0.13, Ti 0.12, Fe 0.23, Si 0.13, and Cu 0.1%, vs. only 305 MPa for the AA 5083 alloy.

IT Welding of metals

(structural; aluminum-magnesium alloys for cold-rolled plates and welded construction)

IT 197586-37-1 197586-38-2 197586-39-3 197586-40-6

RL: TEM (Technical or engineered material use); USES (Uses)
(high-strength; aluminum-magnesium alloys for cold-rolled plates and welded structures)

IT 12616-86-3, AA 5083

RL: TEM (Technical or engineered material use); USES (Uses)
(modification of, with zinc; aluminum-magnesium alloys for cold-rolled plates and welded structures)

L30 ANSWER 74 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 1996:263352 HCAPLUS

DOCUMENT NUMBER: 124:296246

ORIGINAL REFERENCE NO.: 124:54779a,54782a

TITLE: Mechanisms of superplastic deformation of aluminum alloy AMg4

AUTHOR(S): Novikov, I. I.; Nikiforov, A. O.; Polkin, V. I.; Levchenko, V. S.

CORPORATE SOURCE: Mosk. Gos. Inst. Stali Splavov, Russia

SOURCE: Izvestiya Vysshikh Uchebnykh Zavedenii, Tsvetnaya Metallurgiya (1996), (1), 43-8
CODEN: IVUTAK; ISSN: 0021-3438

PUBLISHER: Severo-Kavkazskii Gosudarstvennyi Tekhnologicheskii Universitet

DOCUMENT TYPE: Journal

LANGUAGE: Russian

AB An aluminum based alloy containing 4,3% Mg, 0,6% Mn and 0,2% Cr has been studied. The strain rate dependences of flow stress and index m at superplastic deformation (SPD) were determined within temperature range 500-570 °C. The maximum of the total elongation (1000%) and the index m (0,75) correspond with temperature 570 °C, which is 0,98 T_m . Shifts of marker scratches at grain boundaries and an increase in distance between transverse scratches within grains, as a result of SPD, were measured by SEM technique. The contribution of grain boundary sliding into the total strain determined under optimal conditions ($T = 570$ °C, $\dot{\epsilon} = 1 \cdot 10^{-3}$ s $^{-1}$) is very small (6%) and increases with a decrease in SPD temperature. A conclusion is made that intragranular dislocation slip and diffusion creep are the main SPD mechanisms at 570 °C, and grain equiaxiality is maintained by dynamic recrystn.

IT 81159-87-7, AMg4
RL: PEP (Physical, engineering or chemical process); PROC (Process)
(mechanisms of superplastic deformation of aluminum alloy)

L30 ANSWER 75 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 1995:934047 HCAPLUS
DOCUMENT NUMBER: 124:63743
ORIGINAL REFERENCE NO.: 124:11833a,11836a
TITLE: Manufacture of painted aluminum alloy sheets
for high-strength stay-on tabs of cans
INVENTOR(S): Kaneda, Yutaka; Okamoto, Fumito
PATENT ASSIGNEE(S): Kobe Steel Ltd, Japan
SOURCE: Jpn. Kokai Tokkyo Koho, 6 pp.
CODEN: JKXXAF
DOCUMENT TYPE: Patent
LANGUAGE: Japanese
FAMILY ACC. NUM. COUNT: 1
PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
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JP 07197217	A	19950801	JP 1993-352091	19931229
PRIORITY APPLN. INFO.:			JP 1993-352091	19931229
AB	A slab of Al alloy containing Mg 3.5-5.5, Mn 0.2-1.0, Cu 0.05-0.4, and optionally Si \leq 0.30, Fe \leq 0.4, Cr \leq 0.25, Zn \leq 0.35, Zr \leq 0.15, and Ti \leq 0.20 is homogenized at 450-550°, hot rolled, cold rolled, annealed, cold rolled at a draft of 65-85% to give a maximum size of crystal grain (as viewed from the surface of the rolled sheet) \leq 30 μ m, painted and baked with a maximum heating temperature affecting the sheet of 200-280°.			

IT Cans
(manufacture of painted aluminum alloy sheets for high-strength stay-on tabs of cans)

IT 170635-47-9 170635-48-0 172274-25-8
RL: PEP (Physical, engineering or chemical process); TEM (Technical or engineered material use); PROC (Process); USES (Uses)
(manufacture of painted aluminum alloy sheets for high-strength stay-on tabs of cans)

L30 ANSWER 76 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 1995:833447 HCAPLUS
DOCUMENT NUMBER: 124:183182
ORIGINAL REFERENCE NO.: 124:33786h,33787a
TITLE: Clad aluminum alloys with pitting corrosion resistance for working fluid pipes
INVENTOR(S): Itagaki, Takeshi; Toma, Ken
PATENT ASSIGNEE(S): Mitsubishi Aluminium, Japan
SOURCE: Jpn. Kokai Tokkyo Koho, 11 pp.
CODEN: JKXXAF
DOCUMENT TYPE: Patent
LANGUAGE: Japanese
FAMILY ACC. NUM. COUNT: 1
PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
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JP 07179970	A	19950718	JP 1993-346271	19931222
PRIORITY APPLN. INFO.:			JP 1993-346271	19931222
AB	The clad Al alloys consist of 0.7-2% Mn-containing Al-Mn alloy cores having 0.2-10.6% Mn-containing Al-Mn alloy inner layers and Al or \leq 0.1% Mn-containing Al alloy outer layers on at least one side of the Al-Mn alloy core. The cores			

may be from Al alloys containing Mn 0.7-2, and optionally Mg 0.1-5, Si 0.3-2, Cu 0.01-0.2, Zr 0.05-0.25, Ti 0.05-0.25, V 0.05-0.25, Cr 0.05-0.25, and/or Fe 0.5-1.5%. The inner layers may be from Al alloys containing Mn 0.2-0.6, and optionally Mg 0.1-5, Si 0.3-2, and/or Cu 0.01-0.2%. The outer layers may be from Al alloys containing Mn \leq 0.1, and optionally Zn 0.1-2, In 0.005-0.05, Sn 0.05-0.2, Mg 0.1-5, Si 0.3-2, and/or Cu 0.01-0.2%.

IT Cladding

Pipes and Tubes

(low-Mn Al alloy cores having 2-layer claddings containing high-Mn Al alloy inner and Al outer for pitting corrosion resistance for pipes)

IT 12670-22-3 12780-47-1 56847-73-5 126744-83-0 169256-35-3
169256-36-4 169256-37-5 169256-38-6 169256-39-7 169256-40-0
169361-47-1 169361-48-2

RL: PEP (Physical, engineering or chemical process); TEM (Technical or engineered material use); PROC (Process); USES (Uses)

(cladding inner; low-Mn Al alloy cores having 2-layer claddings containing high-Mn Al alloy inner and Al outer for pitting corrosion resistance for pipes)

IT 7429-90-5, Aluminum, processes 12616-96-5 12665-79-1
39285-45-5 50944-85-9 52361-37-2 59392-25-5 71040-31-8
96742-19-7 169256-41-1 169256-42-2 169256-43-3 169256-44-4
169256-45-5 169256-46-6 169256-47-7 169256-48-8 169361-49-3

RL: PEP (Physical, engineering or chemical process); TEM (Technical or engineered material use); PROC (Process); USES (Uses)

(cladding outer; low-Mn Al alloy cores having 2-layer claddings containing high-Mn Al alloy inner and Al outer for pitting corrosion resistance for pipes)

IT 12670-19-8 12673-16-4 86666-27-5 133014-13-8 138438-97-8
164016-19-7 165740-40-9 169256-22-8 169256-23-9 169256-24-0
169256-25-1 169256-26-2 169256-27-3 169256-28-4 169256-29-5
169256-30-8 169256-31-9 169256-32-0 169256-33-1 169256-34-2
169361-45-9 169361-46-0

RL: PEP (Physical, engineering or chemical process); TEM (Technical or engineered material use); PROC (Process); USES (Uses)

(core; low-Mn Al alloy cores having 2-layer claddings containing high-Mn Al alloy inner and Al outer for pitting corrosion resistance for pipes)

L30 ANSWER 77 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 1994:13100 HCAPLUS

DOCUMENT NUMBER: 120:13100

ORIGINAL REFERENCE NO.: 120:2601a,2604a

TITLE: High-strength and high corrosion-resistant aluminum alloy clad materials for low-temperature brazing

INVENTOR(S): Kishino, Kunihiro; Yamaguchi, Motoyoshi

PATENT ASSIGNEE(S): Furukawa Aluminium, Japan

SOURCE: Jpn. Kokai Tokkyo Koho, 5 pp.

CODEN: JKXXAF

DOCUMENT TYPE: Patent

LANGUAGE: Japanese

FAMILY ACC. NUM. COUNT: 1

PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
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JP 05214475	A	19930824	JP 1991-56208	19910123
PRIORITY APPLN. INFO.:			JP 1991-56208	19910123
AB Al alloys containing Mg 1.5-3.5 and Cr 0.01-0.35, Mn 0.01-1.80, Zr 0.01-0.35, Hf 0.03-0.5, V 0.03-0.35, Ni 0.03-3.5, Fe				

0.02-1.5, and/or Ti 0.005-0.35, and optionally Si 0.03-2.5%, where Cu amount is controlled to <0.5%, are coated with brazes (which melt at ≤500°) on ≥1 side to give the materials. The materials may comprise a sacrificial layer on 1 side. The materials are useful for automobile heat exchangers.

IT Heat-exchange apparatus
(automobile, aluminum alloy clads for, for low-temperature brazing)
IT Cladding
(of aluminum alloys, with brazes, for heat exchangers)
IT 42611-25-6 106902-02-7
RL: USES (Uses)
(brazes, aluminum alloy clad with, for heat exchangers)
IT 151789-43-4 151789-44-5 151789-45-6 151789-46-7
RL: USES (Uses)
(clad, with braze coatings, for heat exchangers)
IT 12675-84-2
RL: USES (Uses)
(sacrificial layer, aluminum alloy clad with, for heat exchangers)

L30 ANSWER 78 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 1993:217890 HCAPLUS
DOCUMENT NUMBER: 118:217890
ORIGINAL REFERENCE NO.: 118:37481a,37484a
TITLE: Manufacture of aluminum-magnesium
-silicon alloy nuts for caulking
INVENTOR(S): Fukuchi, Fumiaki; Yasunaga, Kunihiro; Sato,
Masakazu; Umemura, Hironori
PATENT ASSIGNEE(S): Honda Motor Co., Ltd., Japan; Pop Rivet Fastener Kk
SOURCE: Jpn. Kokai Tokkyo Koho, 4 pp.
CODEN: JKXXAF
DOCUMENT TYPE: Patent
LANGUAGE: Japanese
FAMILY ACC. NUM. COUNT: 1
PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
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JP 05033108	A	19930209	JP 1991-189714	19910730
JP 3069973	B2	20000724		

PRIORITY APPLN. INFO.: JP 1991-189714 19910730

AB The nuts consisting of a tubular body with a flange on 1 end and helical groove on the inner wall of the other end are manufactured from Al-Mg-Si alloy and are solution annealed in reducing atmospheric at 490-520° for 2-4 h or in air at 490-510° for 2-4 h. The manufactured caulking nuts maintain fastening torque under crucial service conditions.

IT Nuts (mechanical)
(caulking, aluminum-magnesium-silicon alloy
, manufacture of, solution annealing for fatigue strength in)
IT Annealing
(solution, of aluminum-magnesium-silicon alloy
caulking nuts, for fatigue strength)
IT 12615-50-8P 12616-75-0P, AA6061
RL: PEP (Physical, engineering or chemical process); PREP (Preparation);
PROC (Process)
(caulking nuts, solution annealing in manufacture of, for fatigue strength)

L30 ANSWER 79 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 1993:259431 HCAPLUS
DOCUMENT NUMBER: 118:259431
ORIGINAL REFERENCE NO.: 118:45023a,45026a

TITLE: Aluminum alloys with torsion-bending fatigue resistance and formability for sheets and pulleys
 INVENTOR(S): Ogura, Kenichi; Kanemitsu, Yukio
 PATENT ASSIGNEE(S): Furukawa Aluminium, Japan; Kanemitsu Kk
 SOURCE: Jpn. Kokai Tokkyo Koho, 4 pp.
 CODEN: JKXXAF
 DOCUMENT TYPE: Patent
 LANGUAGE: Japanese
 FAMILY ACC. NUM. COUNT: 1
 PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
JP 04371545	A	19921224	JP 1991-174383	19910619
PRIORITY APPLN. INFO.:			JP 1991-174383	19910619
AB The alloys contain Mg 2-6, Ti 0.005-0.2 and/or B 0.0005-0.02%, and optionally Mn 0.02-2.0, Cu 0.01-2.0, Cr 0.02-1.0, and/or Zr 0.01-0.3, and impurities such as Si ≤0.2, Fe ≤0.2, and Zn ≤0.5%.				
IT Pulleys (aluminum-magnesium alloy sheets for, torsion-bending fatigue resistance and formability of)				
IT 145077-00-5	145077-01-6	145077-02-7	145077-03-8	147928-73-2
147978-43-6 147978-44-7				
RL: USES (Uses) (for sheets and pulleys, torsion-bending fatigue resistance and formability of)				

L30 ANSWER 80 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 1993:43792 HCAPLUS
 DOCUMENT NUMBER: 118:43792
 ORIGINAL REFERENCE NO.: 118:7827a,7830a
 TITLE: Aluminum alloys for heat rollers in copying or printing machines
 INVENTOR(S): Aiura, Sunao; Kaita, Kazuhiro; Takezoe, Osamu
 PATENT ASSIGNEE(S): Kobe Steel, Ltd., Japan
 SOURCE: Jpn. Kokai Tokkyo Koho, 5 pp.
 CODEN: JKXXAF
 DOCUMENT TYPE: Patent
 LANGUAGE: Japanese
 FAMILY ACC. NUM. COUNT: 1
 PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
JP 04210444	A	19920731	JP 1990-410437	19901212
PRIORITY APPLN. INFO.:			JP 1990-410437	19901212
AB The alloys contain Fe 0.1-0.4, Cu 0.15-0.6, Mn 0.6-1.5, Mg 3.0-5.5, Si <0.5, and optionally Zr 0.05-0.2%. The alloys optionally contain 0.005-0.1% Ti.				
IT Copying process (apparatus, heat rollers in, aluminum alloys for)				
IT Printing apparatus (rollers, heat, aluminum alloys for)				
IT 7440-32-6, Titanium, uses 7440-67-7, Zirconium, uses				
RL: USES (Uses) (aluminum alloys containing, for heat rollers of copiers and printers)				
IT 145418-38-8	145418-39-9	145418-40-2	145418-41-3	145418-42-4
145418-43-5 145418-44-6 145418-45-7				
RL: USES (Uses) (for heat rollers of copiers and printers)				

L30 ANSWER 81 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 1993:454005 HCAPLUS

DOCUMENT NUMBER: 119:54005

ORIGINAL REFERENCE NO.: 119:9685a,9688a

TITLE: Effect of scandium on mechanical properties of welded joints of aluminum alloy 1420

AUTHOR(S): Labur, T. M.; Ishchenko, A. Ya.

CORPORATE SOURCE: Inst. Elektrosvariki im. Patona, Ukraine

SOURCE: Avtomaticheskaya Svarka (1992), (11-12), 53-4

CODEN: AVSVAU; ISSN: 0005-111X

DOCUMENT TYPE: Journal

LANGUAGE: Russian

AB Ar-arc welding of Al alloy 1420 with filler wires AMg4 and AMg63 containing $\leq 0.5\%$ Sc increased weld strength at satisfactory toughness. The Sc-alloyed filler wires gave weld strength factor 0.68-0.72, nominal fracture stress 70-75%, and 2.5-4.0 times higher fracture toughness.

IT Welds
(aluminum alloy, mech. properties of, scandium effect on)

IT Welding
(argon-shielded arc, of aluminum alloy, scandium effect on mech. properties in)

IT 7440-20-2, Scandium, uses

RL: USES (Uses)

(in filler wire, mech. properties of welded joints of aluminum alloy in relation to)

IT 71714-78-8, AMg63 81159-87-7, AMg4

RL: USES (Uses)

(welding of aluminum alloy with wire of, scandium effect on mech. properties in)

IT 37301-69-2, Alloy 01420

RL: USES (Uses)

(welds of, mech. properties of, scandium effect on)

L30 ANSWER 82 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 1992:44979 HCAPLUS

DOCUMENT NUMBER: 116:44979

ORIGINAL REFERENCE NO.: 116:7685a,7688a

TITLE: Arc welding of aluminum alloy 1420 products

AUTHOR(S): Ilyushenko, R. V.; Tretyak, N. G.; Lozovskaya, A. V.; Ishchenko, A. Ya.

CORPORATE SOURCE: Inst. Elektrosvariki im. Patona, Kiev, USSR

SOURCE: Avtomaticheskaya Svarka (1991), (4), 53-6, 60

CODEN: AVSVAU; ISSN: 0005-111X

DOCUMENT TYPE: Journal

LANGUAGE: Russian

AB Welds were prepared by multipass Ar-arc welding with nonconsumable electrode using wires SvAMg3, SvAMg63, AMg4, and VAL-16. The coarse-porous welds were formed in welding of the 2-6-mm thick sheets due to the H surface saturation. The seam porosity was eliminated by pulsed-arc welding and prior removing of ≥ 0.05 -mm deep layers. The surface layers have little effect on welding of thicker plates; microcavities near the fusion line are the main defects. The higher porosity occurred in welding with SvAMg3 wire than with AMg4, SvAMg63, and VAL-16 having lower m.p. The amount of microcavities in weld metal and in fusion and heat-affected zones significantly decreased and high mech. properties were attained in multipass welding with wires containing 4.5-6.0% Mg. The hot-cracking resistance of welds increased using the modified Al-Mg welding wires to obtain welds with fine-crystalline structure.

IT Welds
(aluminum alloy, gas tungsten-arc, porosity and hot-cracking

resistance of)

IT Welding
(gas tungsten-arc, of aluminum alloy, with magnesium
-containing wires, porosity and hot cracking in relation to)

IT 37301-69-2, Alloy 1420
RL: PEP (Physical, engineering or chemical process); PROC (Process)
(welding of, argon-arc nonconsumable-electrode, porosity and hot
cracking resistance in relation to)

IT 71714-78-8, SvAMg63 81159-87-7, AMg4 125352-52-5, SvAMg3
135667-16-2, VAL-16
RL: USES (Uses)
(welding wire, for aluminum alloy, porosity and hot cracking
in relation to)

L30 ANSWER 83 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN
ACCESSION NUMBER: 1991:476647 HCAPLUS
DOCUMENT NUMBER: 115:76647
ORIGINAL REFERENCE NO.: 115:13139a,13142a
TITLE: Physical microheterogeneity of aluminum
-alloy welds and initiation of corrosion-mechanical
defects
AUTHOR(S): Galkanov, V. A.; Sorokin, V. N.
CORPORATE SOURCE: TsNIIproekstalkonstr., USSR
SOURCE: Svarochnoe Proizvodstvo (1991), (3), 35-7
CODEN: SVAPAI; ISSN: 0491-6441
DOCUMENT TYPE: Journal
LANGUAGE: Russian

AB Stress corrosion of Ar arc welds of Al-Zn-Mg (heat-treated alloy
1915) and Al-Mg (strain-hardened AMg4) plates prepared with 1557
(AMg5) wire was studied. Interface microheterogeneity was responsible for
the initial stage of weld corrosion cracking. Elastoplastic deformation
did not affect strongly the corrosion crack nucleation.

IT Welds
(aluminum alloys, corrosion cracking of, microheterogeneity
effect on)

IT Welding
(argon-shielded arc, of aluminum alloys, interface
microheterogeneity effect on corrosion cracking in relation to)

IT 72267-09-5, AMg 5
RL: USES (Uses)
(welding wire, for aluminum alloys, interface
microheterogeneity effect on corrosion cracking in relation to)

IT 37360-00-2, Alloy 1915 81159-87-7, AMg4
RL: USES (Uses)
(welds, corrosion cracking of, effect of microheterogeneity on)

L30 ANSWER 84 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN
ACCESSION NUMBER: 1992:64724 HCAPLUS
DOCUMENT NUMBER: 116:64724
ORIGINAL REFERENCE NO.: 116:11071a,11074a
TITLE: Structure and properties of aluminum-
magnesium-lithium alloy welds
AUTHOR(S): Ilyushenko, R. V.; Lozovskaya, A. V.; Sklabinskaya, I.
E.; Tretyak, N. G.; Chaika, A. A.
CORPORATE SOURCE: Inst. Elektrosvariki im. Patona, Kiev, USSR
SOURCE: Avtomaticheskaya Svarka (1991), (7), 23-6
CODEN: AVSVAU; ISSN: 0005-111X
DOCUMENT TYPE: Journal
LANGUAGE: Russian

AB The effect of Sc addition on microstructure and mech. properties of the Al-
Mg-Li alloy welds was studied. Welding wires AMg4 and AMg63
containing Sc and Sc-free and SvAMg63 were used. The strength (350-360 MPa)

of the Al-Mg-Li-Sc welds was higher by 30-60 MPa than that of alloy 1420. Optimal combination of strength, ductility, and toughness of the Al-Mg-Li-Sc welds was attained using wire SvAMg63 containing 0.17% Sc. Welding with wire containing 0.5% Sc caused grain refinement. Heating to 670 K followed by artificial aging at 410 K after welding increased the weld strength to 380-400 MPa.

- IT Welds
 (aluminum-magnesium-lithium alloy, microstructure and mech. properties of, scandium addition and heat treatment effects on)
- IT Welding
 (gas tungsten-arc, of aluminum-magnesium-lithium alloys, scandium addition effect on)
- IT 7440-20-2, Scandium, uses
 RL: USES (Uses)
 (aluminum alloy welds containing, microstructure and mech. properties of)
- IT 71714-78-8, AMg63 81159-87-7, AMg4
 RL: USES (Uses)
 (welding with, of aluminum-magnesium-lithium alloys, scandium addition effect on)
- IT 37301-69-2, Alloy 1420
 RL: USES (Uses)
 (welds, microstructure and mech. properties of, scandium addition effect on)

L30 ANSWER 85 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 1990:557081 HCAPLUS

DOCUMENT NUMBER: 113:157081

ORIGINAL REFERENCE NO.: 113:26625a,26628a

TITLE: Processing of nonrecrystallized aluminum alloy sheets and plates.

INVENTOR(S): Cho, Alex

PATENT ASSIGNEE(S): Aluminum Co. of America, USA

SOURCE: Eur. Pat. Appl., 12 pp.

CODEN: EPXXDW

DOCUMENT TYPE: Patent

LANGUAGE: English

FAMILY ACC. NUM. COUNT: 1

PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
EP 368005	A1	19900516	EP 1989-118810	19891010
EP 368005	B1	19960911		
R: DE, FR, GB				
US 4927470	A	19900522	US 1988-256840	19881012
US 4946517	A	19900807	US 1988-256520	19881012
US 4988394	A	19910129	US 1988-256521	19881012
JP 02194153	A	19900731	JP 1989-266083	19891012
PRIORITY APPLN. INFO.:			US 1988-256520	A 19881012
			US 1988-256521	A 19881012
			US 1988-256840	A 19881012

AB Al alloy sheets and plates are prepared by ramp annealing, solution heat treating, quenching, and aging. The annealing is preferably started at ≤ 400 or ≤ 750 and finished at $680-850^{\circ}\text{F}$, or started at $350-450$ and in 2-8 h finished at $750-850^{\circ}\text{F}$. The process is suitable for retention of fine-grained structure in the AA2000, AA6000, AA7000, and AA8000 type alloys. The resulting products show improved strength as well as fracture toughness, especially for aircraft applications. Thus, an Al alloy ingot (containing Zn 10, Mg 1.8, Cu 1.5, and Zr 0.12%) was heated and hot rolled to 1.5-in.-thick slabs, which were annealed at $750-880^{\circ}\text{F}$ and then hot rolled to 0.3-in.-thick

plates. The plates were heated for 16 h at 400°F, heated further to 800°F in 4 h, held for 1 h, and water quenched. The resulting microstructure showed no recrystn.

IT Aluminum alloy, base

RL: USES (Uses)

(processing of nonrecrystd., for toughness)

IT 129703-71-5, Aluminum 87, copper 1.5, magnesium 1.8,
zinc 10, zirconium 0.1 129703-72-6 129703-73-7 129703-74-8
129703-75-9 129703-76-0

RL: USES (Uses)

(processing of nonrecrystd., for toughness)

L30 ANSWER 86 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 1990:635772 HCAPLUS

DOCUMENT NUMBER: 113:235772

ORIGINAL REFERENCE NO.: 113:39703a,39706a

TITLE: Mechanical properties of butt welds of wrought
aluminum alloys

AUTHOR(S): Ryazantseva, V. I.; Grinin, V. V.; Ovchinnikov, V. V.

CORPORATE SOURCE: USSR

SOURCE: Svarochnoe Proizvodstvo (1990), (8), 8-10

CODEN: SVAPAI; ISSN: 0491-6441

DOCUMENT TYPE: Journal

LANGUAGE: Russian

AB The static strength of butt welds of wrought Al alloys depends slightly on the method or regime of welding, or on the filler wire. The product form (sheet, hot-rolled plate, strip, forging, or stamping) affects most the static strength of weldment. The ductility of welds depends on the method of welding, filler metal, and product form. The best ductility is attained using pulsed arc welding. The fatigue strength of welds is determined by the weld shape and the product form, while the effect of other parameters is insignificant. The 01570-type alloys are recommended for weldments working at low-cycle loads (≤ 210 -240 MPa); such weldments show a 1.5-5 fold increase in durability, compared to that of conventional AMg3, AMg4, AMg6, or 1201-type alloys.

IT Welds

(butt, aluminum alloys, mech. properties of)

IT Welding

(butt, of aluminum alloys)

IT 37301-69-2, Alloy 1420 64159-59-7, Alloy 01557 71631-36-2, Alloy 1177
71714-78-8, AMg63 81159-87-7, AMg4 125352-52-5, AMg3
125726-66-1, Alloy 1571

RL: USES (Uses)

(welding with filler of, mech. properties in relation to butt)

IT 12672-17-2, Alloy 1201 12732-16-0, AMg6 54424-86-1 55926-30-2,
Aluminum base, lithium, magnesium 125726-63-8, Alloy
01570 125726-92-3, M40-1

RL: USES (Uses)

(welds, mech. properties of butt)

L30 ANSWER 87 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 1989:462193 HCAPLUS

DOCUMENT NUMBER: 111:62193

ORIGINAL REFERENCE NO.: 111:10467a,10470a

TITLE: Corrosion of aluminum alloys in closed
agricultural premises

AUTHOR(S): Rogozhina, E. P.; Koltunova, G. A.; Pashkova, O. A.;
Golubev, A. I.

CORPORATE SOURCE: TsNIIProektstal'konstruktsiya, USSR

SOURCE: Zashchita Metallov (1989), 25(1), 120-4

CODEN: ZAMEA9; ISSN: 0044-1856

DOCUMENT TYPE: Journal

LANGUAGE: Russian

AB The corrosion of Al-Mg, Al-Mn, Al-Zn-Mg, and Al-Mg-Si alloys and tech. grade Al in agricultural buildings (greenhouses, fruit-vegetable canning plant, fertilizer storage building, champignon growing chamber) was studied. The corrosion resistance was greatly enhanced by anodization in a 20% H₂SO₄ + 1% H₂C₂O₄ electrolyte at anodic c.d. 200 A/m² and 18-23°.

IT Agriculture and Agricultural chemistry
(corrosion of aluminum alloys in, anodization for resistance to)

IT Anodization
(of aluminum alloys, for corrosion resistance in agricultural buildings)

IT 11146-15-9, AMtsN2 37302-00-4, 1911T 37360-00-2, 1915T 63747-67-1, AVT

RL: PEP (Physical, engineering or chemical process); PROC (Process)
(corrosion of, in agricultural building, anodization for resistance to)

IT 7429-90-5, Aluminum, reactions 11121-92-9 37268-38-5, AD1M
37301-70-5 81159-87-7, AMg4M

RL: PEP (Physical, engineering or chemical process); PROC (Process)
(corrosion of, in agricultural buildings, anodization for resistance to)

L30 ANSWER 88 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 1990:123446 HCAPLUS

DOCUMENT NUMBER: 112:123446

ORIGINAL REFERENCE NO.: 112:20839a,20842a

TITLE: Estimation of the weldability of aluminum alloys

AUTHOR(S): Ryazantsev, V. I.; Grinin, V. V.; Ovchinnikov, V. V.

CORPORATE SOURCE: USSR

SOURCE: Svarochnoe Proizvodstvo (1989), (9), 7-9

CODEN: SVAPAI; ISSN: 0491-6441

DOCUMENT TYPE: Journal

LANGUAGE: Russian

AB Hot-crack susceptibility in elec.-arc welding of sheet, plate, and forged specimens 2-15 mm thick was evaluated for Al alloys AMg6, 1201, Al-Mg-Li, Al-Cu-Li, 1570-1, 1570-2, M40, and M40-1. Weldability tests were conducted with wire fillers. Hot cracks and delamination defects were typically associated with chemical inhomogeneous structure in the cross-section of industrial preforms. Welding automation and use of rotary electrodes were considered. Application of pulses to elec. current increased weld quality.

IT Welds
(in aluminum alloy strip, hot cracks in, structure effect on)

IT Welding
(of aluminum alloys, hot crack susceptibility in relation to)

IT 11100-85-9, M40 (Aluminum alloy) 12732-16-0, AMg6
37321-72-5, Alloy 1201 (aluminum alloy) 54424-86-1
55926-30-2, Aluminum base, lithium, magnesium
125726-64-9, Alloy 1570-1 (aluminum alloy) 125726-65-0, Alloy
1570-2 (aluminum alloy) 125726-92-3, M40-1 (Aluminum alloy)

RL: PEP (Physical, engineering or chemical process); PROC (Process)
(welding of, hot crack susceptibility in, structure effect on)

IT 37301-69-2, Alloy 1420 (aluminum alloy) 71631-36-2, Alloy 1177
(aluminum alloy) 71714-78-8, AMg63 81159-87-7
106747-93-7, AK5 (aluminum alloy) 125726-66-1, Alloy 1571 (aluminum alloy)

RL: USES (Uses)
(welding wire filler, hot crack susceptibility in relation to)

L30 ANSWER 89 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 1988:10081 HCAPLUS
DOCUMENT NUMBER: 108:10081
ORIGINAL REFERENCE NO.: 108:1729a,1732a
TITLE: Improvement of strength and rupture-toughness
of aluminum alloy containing lithium
PATENT ASSIGNEE(S): Boeing Co., USA
SOURCE: Jpn. Kokai Tokkyo Koho, 6 pp.
CODEN: JKXXAF
DOCUMENT TYPE: Patent
LANGUAGE: Japanese
FAMILY ACC. NUM. COUNT: 2
PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
JP 62164859	A	19870721	JP 1986-251564	19861021
US 4840682	A	19890620	US 1985-800503	19851121
CA 1280341	C	19910219	CA 1986-514223	19860721
US 4999061	A	19910312	US 1989-337956	19890414
PRIORITY APPLN. INFO.:			US 1985-800503	A 19851121
			US 1983-567227	B2 19831230

AB Al alloys containing Li 1.0-3.2, Mg 0-5.5, Cu 0-4.5, Zr 0.08-0.15, Mn 0-1.2, Fe \leq 0.3, Si \leq 0.5, Zn \leq 0.25, Ti \leq 0.15, and others \leq 0.3% are solution treated, quenched, and aged for 1-80 h at 200-300°.

IT 111892-44-5
RL: USES (Uses)
(solution treatment and quenching and aging of, for strengthening and toughening)

L30 ANSWER 90 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 1988:99472 HCAPLUS
DOCUMENT NUMBER: 108:99472
ORIGINAL REFERENCE NO.: 108:16271a,16274a
TITLE: Aluminum alloy for die-casting without cracking
INVENTOR(S): Hirasawa, Hiroaki; Takikita, Takanori
PATENT ASSIGNEE(S): Nippon Light Metal Co., Ltd., Japan
SOURCE: Jpn. Kokai Tokkyo Koho, 4 pp.
CODEN: JKXXAF
DOCUMENT TYPE: Patent
LANGUAGE: Japanese
FAMILY ACC. NUM. COUNT: 1
PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
JP 62142739	A	19870626	JP 1985-282990	19851218
PRIORITY APPLN. INFO.:			JP 1985-282990	19851218

AB The die-casting Al alloy contains Zn 2-4.0, Mg 4-7, Fe 0.4-1.0, Mn 0.2-1.0, Si 0.4-1.0, and Cu 0.01-0.5, optionally with Ti $<$ 0.2, B $<$ 0.1%, and/or Zr 0.05-0.2%. The castings show tensile strength 28-36 kg/mm² with high elongation after heat treatment. Thus, molten Al alloy (containing Zn 3.0, Mg 5.0, Fe 0.6, Mn 0.4, Si 0.6, Ti 0.005, B 0.001, and Cu 0.1%) was poured at 700° to manufacture a die-cast plate 6 mm thick. The plate was solution heat-treated at 500° and quenched in water. Tensile strength was 33, yield strength 21 kg/mm², and elongation 5%, vs. 35.8, 33 kg/mm², and 1.2% for a similar plate from AADC 12 alloy. A die-cast plate 6 mm thick with fins 0.5-2 mm thick did not form casting cracks, while a similar product from AADC 12 formed cracks at the fin roots.

IT Casting process
 (die-, of aluminum-magnesium-zinc alloy, structural
 parts without cracks by)

IT 7439-95-4
 RL: USES (Uses)
 (casting process, die-, of aluminum-
 magnesium-zinc alloy, structural parts without cracks by)

IT 112985-66-7 112985-67-8 112985-68-9
 RL: PEP (Physical, engineering or chemical process); PROC (Process)
 (die casting of, heat treatment after, strength and
 ductility by)

L30 ANSWER 91 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 1987:501308 HCAPLUS
 DOCUMENT NUMBER: 107:101308
 ORIGINAL REFERENCE NO.: 107:16460h,16461a
 TITLE: Metallic gasket
 INVENTOR(S): Sakai, Yakichi
 PATENT ASSIGNEE(S): Nippon Gakki Co., Ltd., Japan; Hamamatsu Gasket
 Seisakusho Ltd.
 SOURCE: Ger. Offen., 7 pp.
 CODEN: GWXXBX
 DOCUMENT TYPE: Patent
 LANGUAGE: German
 FAMILY ACC. NUM. COUNT: 1
 PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
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DE 3633988	A1	19870416	DE 1986-3633988	19861006
DE 3633988	C2	19900613		
US 4810591	A	19890307	US 1986-916293	19861007
PRIORITY APPLN. INFO.:			JP 1985-224777	A 19851011

AB A composite for manufacture of gaskets for internal-combustion engines consists of a substrate 0.10-0.35 mm thick made from a heat-resistant spring steel containing C 0.4-1.0 and Si 0.1-0.5%, or from a Ni alloy containing 10-25% Cr, clad on each side with 0.03-0.15 mm thick layers of Cu, Al, Cu alloy containing Fe ≤ 3.5 , Sn ≤ 3.0 , Zn ≤ 45 , Al ≤ 12 , Mn ≤ 2.0 , Ni ≤ 35 , and P $\leq 0.5\%$, or Al alloy containing Si ≤ 1.2 , Fe ≤ 1.0 , Cu ≤ 5.0 , Mn ≤ 1.5 , Mg ≤ 5.0 , Cr ≤ 0.5 , Zn ≤ 5.0 , Ti ≤ 0.5 , V ≤ 0.5 , and Zr $\leq 0.5\%$. Optionally, the steel contains Mn ≤ 1.0 , Cr ≤ 1.5 , and/or V $\leq 0.5\%$. The Ni alloy optionally contains Fe ≤ 30 , C ≤ 0.2 , Si ≤ 1.0 , Mn ≤ 1.0 , Cu ≤ 1.0 , Al ≤ 2.0 , Ti ≤ 3.0 , and/or (Nb + Ta) $\leq 1.5\%$. Preparation of gaskets involves heat treatment at 350-500° and at 450-500° for Cu and Al cladding, resp. Thus, the composite having SUS 301 H alloy substrate had a tensile strength of >170 kg/mm² after heat treatment at .apprx.400°.

IT Gaskets
 (steel-copper or steel-aluminum composite, for
 internal-combustion engines)

IT 12725-26-7 109982-56-1
 RL: USES (Uses)
 (composite of, with copper or aluminum cladding, for
 internal-combustion engine gaskets)

IT 58674-67-2 109982-57-2, uses and miscellaneous 109982-58-3
 109982-59-4 109982-60-7 109982-61-8 109982-62-9 109982-63-0
 110000-08-3
 RL: USES (Uses)
 (composite of, with copper, aluminum, copper alloy, or

aluminum alloy cladding, for internal-combustion engine
gaskets)

REFERENCE COUNT: 7 THERE ARE 7 CITED REFERENCES AVAILABLE FOR THIS
RECORD. ALL CITATIONS AVAILABLE IN THE RE FORMAT

L30 ANSWER 92 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 1986:54985 HCAPLUS
DOCUMENT NUMBER: 104:54985
ORIGINAL REFERENCE NO.: 104:8801a,8804a
TITLE: Low-density aluminum alloys
INVENTOR(S): Skinner, David John; Okazaki, Kenji; Adam, Colin
McClean
PATENT ASSIGNEE(S): Allied Corp., USA
SOURCE: Eur. Pat. Appl., 28 pp.
CODEN: EPXXDW
DOCUMENT TYPE: Patent
LANGUAGE: English
FAMILY ACC. NUM. COUNT: 1
PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
EP 158769	A1	19851023	EP 1985-100476	19850118
EP 158769	B1	19880504		
R: CH, DE, FR, GB, LI				
US 4661172	A	19870428	US 1984-584856	19840229
CA 1228491	A1	19871027	CA 1985-474001	19850211
JP 60208445	A	19851021	JP 1985-40244	19850228
JP 02036661	B	19900820		
JP 01272742	A	19891031	JP 1988-67998	19880322

PRIORITY APPLN. INFO.: US 1984-584856 A 19840229

AB Light-weight Al alloys having high strength and toughness are
suitable for structural components in aircraft, spacecraft, and
automobiles. The alloys contain Li 2.7-5, Mg 0.5-8, and Zr
0.25-2 with Cu, Si, Sc, Ti, V, Hf, Be, Cr, Mn, Fe, Co, and/or Ni
0.5-5%. The alloys have a fine-grained matrix of supersatd. Al-alloy
solid solution with uniformly dispersed intermetallic phases. Powdered alloys
are sintered in vacuum at elevated temperature, followed by a solution heat
treatment, quenching in a fluid bath, and optionally stretching and aging.
Thus, Al alloy containing Li 4, Cu 3, Mg 1.5, and Zr 1.25% after
heat treatment for 2 h at 350° had intermetallic phases 10-20% the
size of those in a similar alloy containing 0.2% Zr.

IT 100081-46-7 100081-47-8 100100-19-4

RL: USES (Uses)

(strength and toughness of low-d.)

L30 ANSWER 93 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 1984:55598 HCAPLUS
DOCUMENT NUMBER: 100:55598
ORIGINAL REFERENCE NO.: 100:8455a,8458a
TITLE: Corrosion of aluminum alloys in farm
buildings
AUTHOR(S): Rogozhina, E. P.; Pashkova, O. A.; Golubev, A. I.
CORPORATE SOURCE: Tsentr. Nauchno-Issled. Proektn. Inst. Stroit.
Metallokonstr., Moscow, USSR
SOURCE: Zashchita Metallov (1983), 19(6), 879-84
CODEN: ZAMEA9; ISSN: 0044-1856
DOCUMENT TYPE: Journal
LANGUAGE: Russian

AB The corrosion resistance of Al alloys AD1M [37268-38-5], AMg2AP
[88505-71-9], AMg4N [81159-87-7], AD31T5 [11121-92-9], AVAT
[73929-28-9], 1915T [61536-53-6], and 1911T [37302-00-4] in livestock

barns and waste treatment systems was studied, some of the samples being anodized in 20% H₂SO₄ solution with the addition of 1% oxalic acid at 2 A/cm² and 18-23°. In the low-aggressive atmospheric of livestock barns, corrosion was not observed in AMg₄N and its depth in Al-Mg and Al-Mg-Si alloys was <60 μ only. Al-Mg-Zn alloys required the protection provided by the 8-9 μ anodic-oxide coating, properties of which remained unchanged for <2 yr. In the waste treatment system atmospheric (aeration tanks), AMg₄N had the highest corrosion resistance, whereas 1915T, the lowest one. All the alloys required protection. The anodic-oxide coating was highly effective.

IT Anodization

(of aluminum alloys, for corrosion protection in livestock barns and aeration tanks)

IT Coating materials

(anodic, on aluminum alloys, for corrosion protection in livestock barns and aeration tanks)

IT 11121-92-9 37268-38-5 37301-70-5 37302-00-4 37360-00-2
73929-28-9 81159-87-7

RL: PEP (Physical, engineering or chemical process); PROC (Process)
(corrosion of, in livestock barns and aeration tanks, anodic oxide coating effect on)

L30 ANSWER 94 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 1981:51594 HCAPLUS

DOCUMENT NUMBER: 94:51594

ORIGINAL REFERENCE NO.: 94:8365a,8368a

TITLE: Welding crack behavior of aluminum alloys

AUTHOR(S): Schoer, H.

CORPORATE SOURCE: Leichtmet.-Forschungsinstit., Verein. Alum.-Werke A.-G.,
Bonn, Fed. Rep. Ger.

SOURCE: Metall (Isernhagen, Germany) (1980), 34(6), 546-51
CODEN: MTLLAF; ISSN: 0026-0746

DOCUMENT TYPE: Journal

LANGUAGE: German

AB Formation of hot cracks was studied in welds of 99-99.9% Al, non-hardenable Al-Mn, Al-Mg. and Al-Mg-Mn alloys, and precipitation-hardenable Al-Cu-Mg, Al-Cu-Si-Mn, Al-Mg-Si, Al-Zn-Mg, and Al-Zn-Mg-Cu alloys. The resistance to weld cracking of AlZn_{4.5}Mg₁ [12675-83-1] decreased in the presence of Cu and increased with increasing Zr content introduced with the filler wires S-AlMg₅Zr [75686-79-2] and S-AlMg_{4.5}MnZr [75686-78-1]. The formation of H-induced micropores in the welds of alloy AlZn_{4.5}Mg₁F35 was also decreased the presence of Zr.

IT Welds

(aluminum alloy, cracking of)

IT 7429-90-5, uses and miscellaneous 12608-67-2 12615-48-4 12616-75-0
12616-86-3 12627-49-5 12675-83-1 12675-83-1 12732-10-4
12732-13-7 75635-87-9 75686-78-1

RL: USES (Uses)

(cracking of welded)

IT 12675-83-1 75686-78-1

RL: USES (Uses)

(welding by, of aluminum alloys, cracking in relation to)

L30 ANSWER 95 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 1977:159815 HCAPLUS

DOCUMENT NUMBER: 86:159815

ORIGINAL REFERENCE NO.: 86:25067a,25070a

TITLE: Aluminum alloy

INVENTOR(S): Fridlyander, I. N.; Anan'in, S. N.; Gol'dbukht, G. E.;
Balakhontsev, G. A.; Moskvichev, G. G.; Byvalov, A.

PATENT ASSIGNEE(S): A.; Efremov, N. L.; Seredkin, A. V.; Nazarov, A. N.
 SOURCE: USSR
 U.S.S.R. From: Otkrytiya, Izobret., Prom. Obraztsy,
 Tovarnye Znaki 1976, 53(38), 90.
 CODEN: URXXAF
 DOCUMENT TYPE: Patent
 LANGUAGE: Russian
 FAMILY ACC. NUM. COUNT: 1
 PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
SU 531883	A1	19761015	SU 1974-2042052	19740708
PRIORITY APPLN. INFO.:			SU 1974-2042052	A 19740708
AB To increase strength and maintain corrosion resistance, Zr, Be, and Sb are added. A typical Al alloy [62388-53-8] contains Mg 1.8-4.2, Si 0.3-1.7, Cu 0.01-1.6, Mn 0.1-0.8, Cr 0.01-0.3, Fe 0.01-0.9, Zn 0.01-1.5, Ti 0.001-0.15, Ni 0.001-0.2, Pb 0.0001-0.05, Sn 0.0001-0.05, Zr 0.001-0.15, Be 0.0001-0.01, and Sb 0.001-0.15 weight%.				
IT 7440-36-0, properties 7440-41-7, properties 7440-67-7, properties RL: PRP (Properties) (corrosion resistance and strength of aluminum-magnesium alloys containing)				
IT 62388-53-8 RL: USES (Uses) (corrosion-resistant high-strength)				

L30 ANSWER 96 OF 96 HCAPLUS COPYRIGHT 2009 ACS on STN

ACCESSION NUMBER: 1977:110161 HCAPLUS
 DOCUMENT NUMBER: 86:110161
 ORIGINAL REFERENCE NO.: 86:17365a,17368a
 TITLE: Aluminum alloy
 INVENTOR(S): Fridlyander, I. N.; Isaev, V. I.; Chekanov, A. N.;
 Vinokurov, N. D.; Kopytov, G. A.; Savchuk, A. I.;
 Zhigalovskii, B. V.; Bazhenov, V. A.; Varlamov, S. B.;
 et al.
 PATENT ASSIGNEE(S): USSR
 SOURCE: U.S.S.R. From: Otkrytiya, Izobret., Prom. Obraztsy,
 Tovarnye Znaki 1976, 53(37), 68-9.
 CODEN: URXXAF
 DOCUMENT TYPE: Patent
 LANGUAGE: Russian
 FAMILY ACC. NUM. COUNT: 1
 PATENT INFORMATION:

PATENT NO.	KIND	DATE	APPLICATION NO.	DATE
SU 530919	A1	19761005	SU 1975-2165712	19750731
PRIORITY APPLN. INFO.:			SU 1975-2165712	A 19750731
AB The Al alloy [61992-87-8] with increased strength, ductility, and stress rupture strength contains Zn 7-11, Mg 2.5-4.0, Cu 1.0-2.6, Mn 0.2-1.2, Zr 0.1-0.3, Ti 0.02-0.3, Be 0.005-0.2, Co 0.05-0.3, and Nb 0.005-0.2 in addition to Cr, B, V, Cd, W, Fe, and/or Si 0.0001-0.3 weight% each.				
IT 61992-87-8 RL: USES (Uses) (with increased ductility and strength)				

=> d cost

COST IN U.S. DOLLARS

SINCE FILE
 ENTRY

TOTAL
 SESSION

CONNECT CHARGES	5.56	65.82
NETWORK CHARGES	0.14	2.38
SEARCH CHARGES	0.00	85.60
DISPLAY CHARGES	324.48	336.91

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FULL ESTIMATED COST	330.18	490.71

DISCOUNT AMOUNTS (FOR QUALIFYING ACCOUNTS)	SINCE FILE	TOTAL
	ENTRY	SESSION
CA SUBSCRIBER PRICE	-78.72	-78.72

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LOGOFF? (Y)/N/HOLD:y

COST IN U.S. DOLLARS	SINCE FILE	TOTAL
	ENTRY	SESSION
FULL ESTIMATED COST	333.03	493.56

DISCOUNT AMOUNTS (FOR QUALIFYING ACCOUNTS)	SINCE FILE	TOTAL
	ENTRY	SESSION
CA SUBSCRIBER PRICE	-78.72	-78.72

STN INTERNATIONAL LOGOFF AT 09:24:41 ON 30 MAR 2009